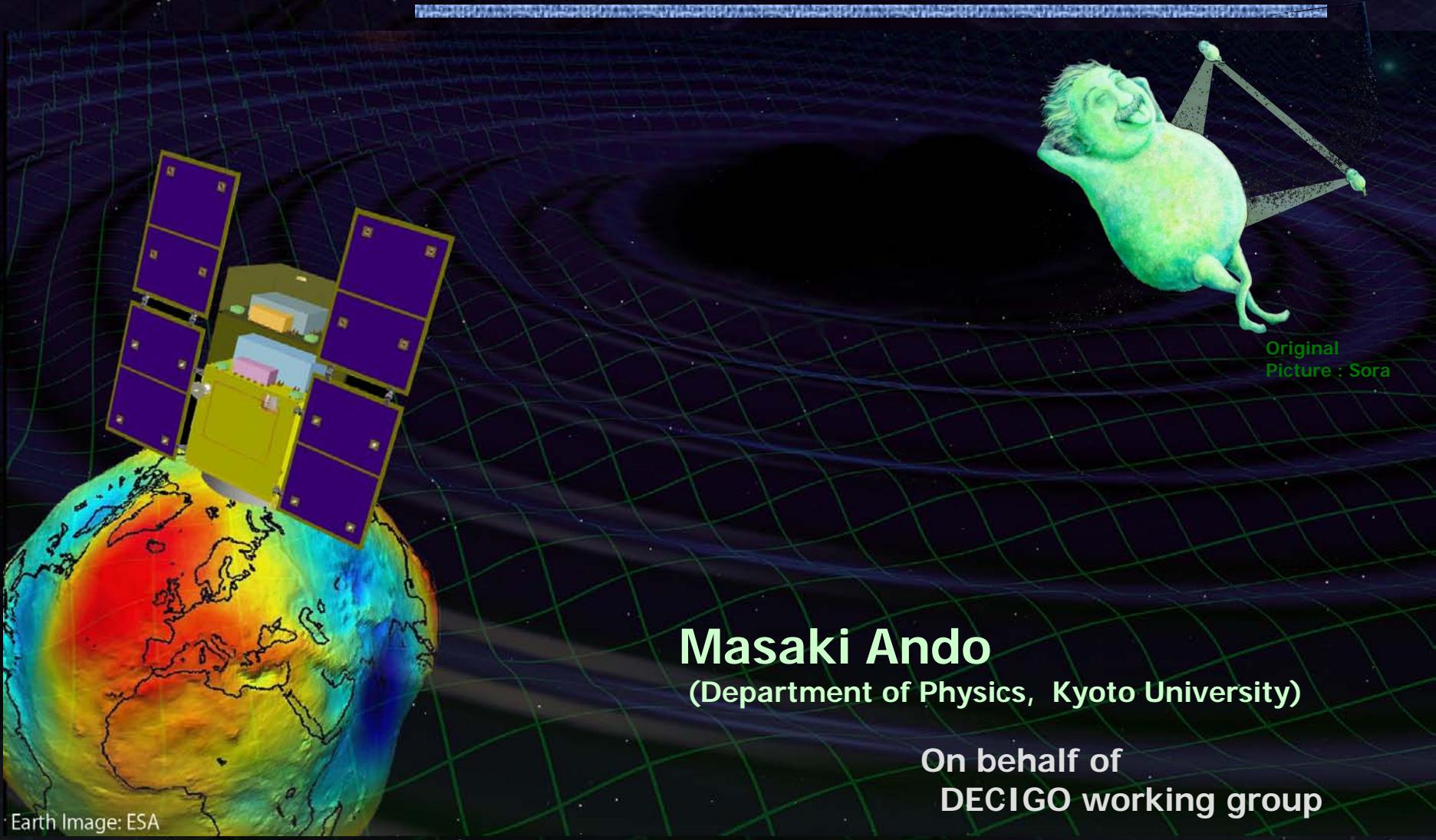


DECIGO and DECIGO Pathfinder



Masaki Ando

(Department of Physics, Kyoto University)

On behalf of
DECIGO working group

1. DECIGO

Overview and Science

Pre-conceptual Design

2. DECIGO Pathfinder

Overview and Science

Design and Status

Space Demonstration

3. Summary



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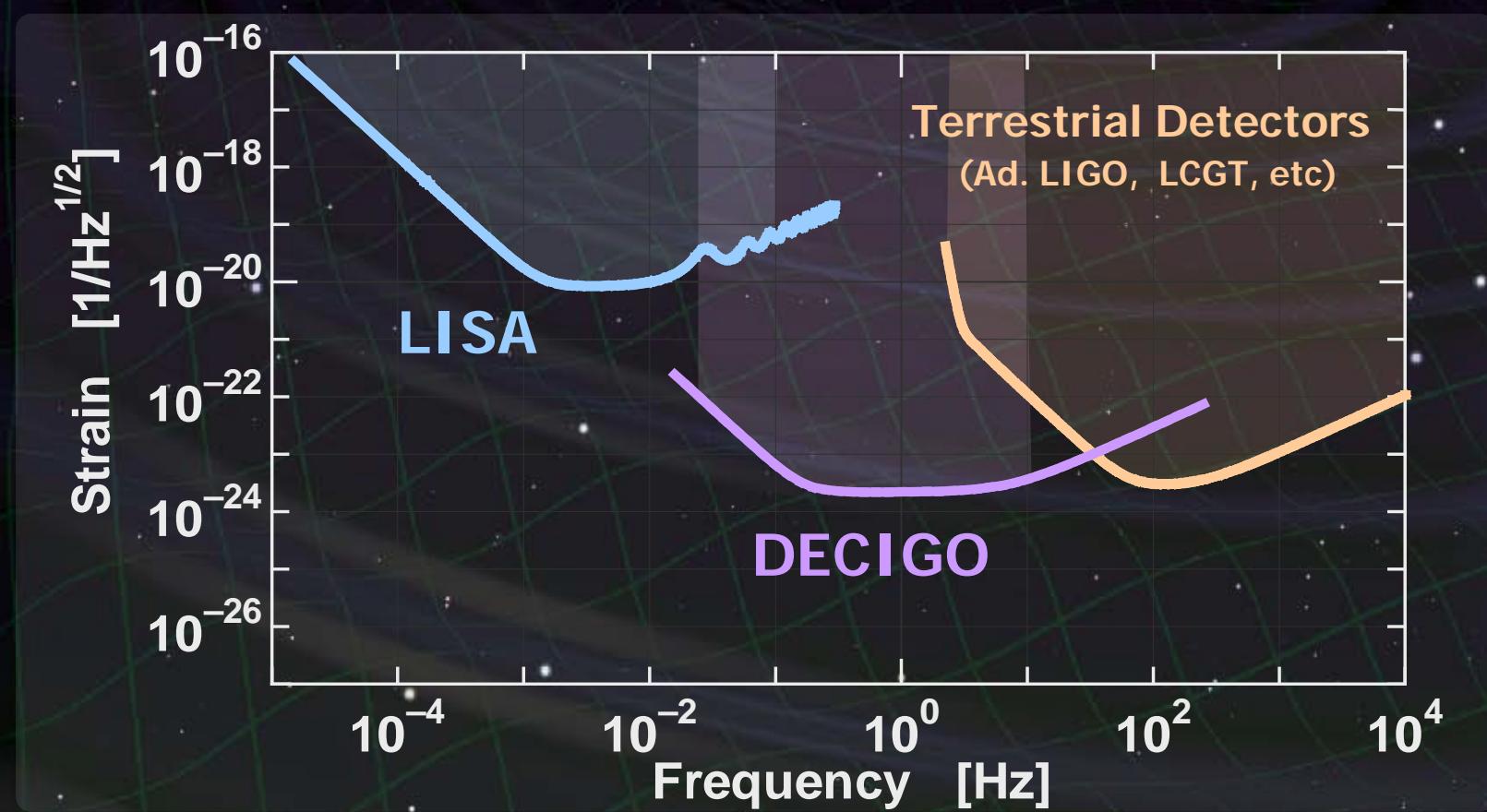
3. Summary

DECIGO (Deци-Hertz Interferometer Gravitational wave Observatory)

Space GW antenna (~2027)
Obs. band around 0.1 Hz



'Bridge' the obs.gap between
LISA and Terrestrial detectors

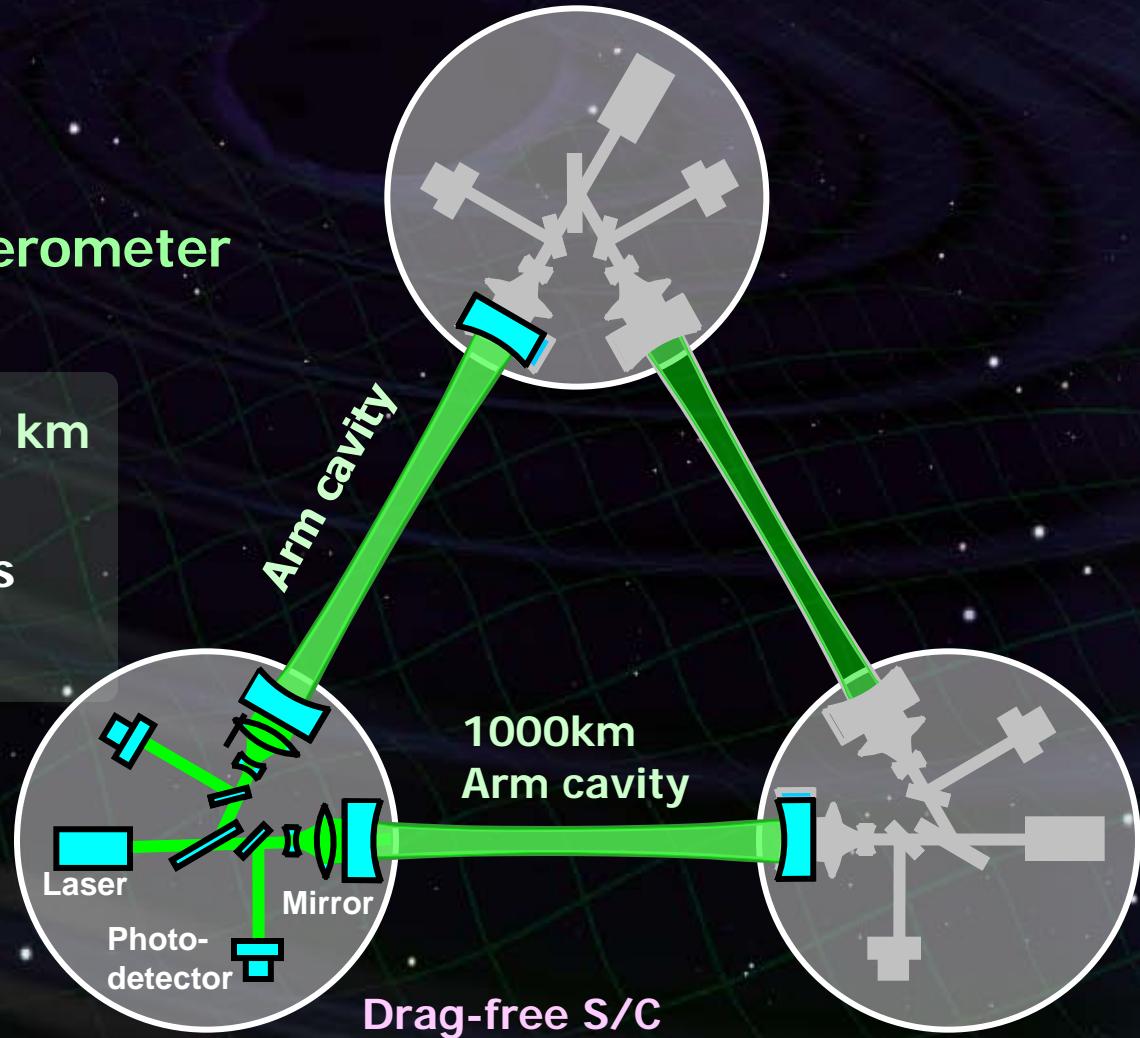


DECIGO Interferometer



Interferometer Unit: Differential FP interferometer

Baseline length: 1000 km
3 S/C formation flight
3 FP interferometers
Drag-free control



Targets and Science



IMBH binary inspiral

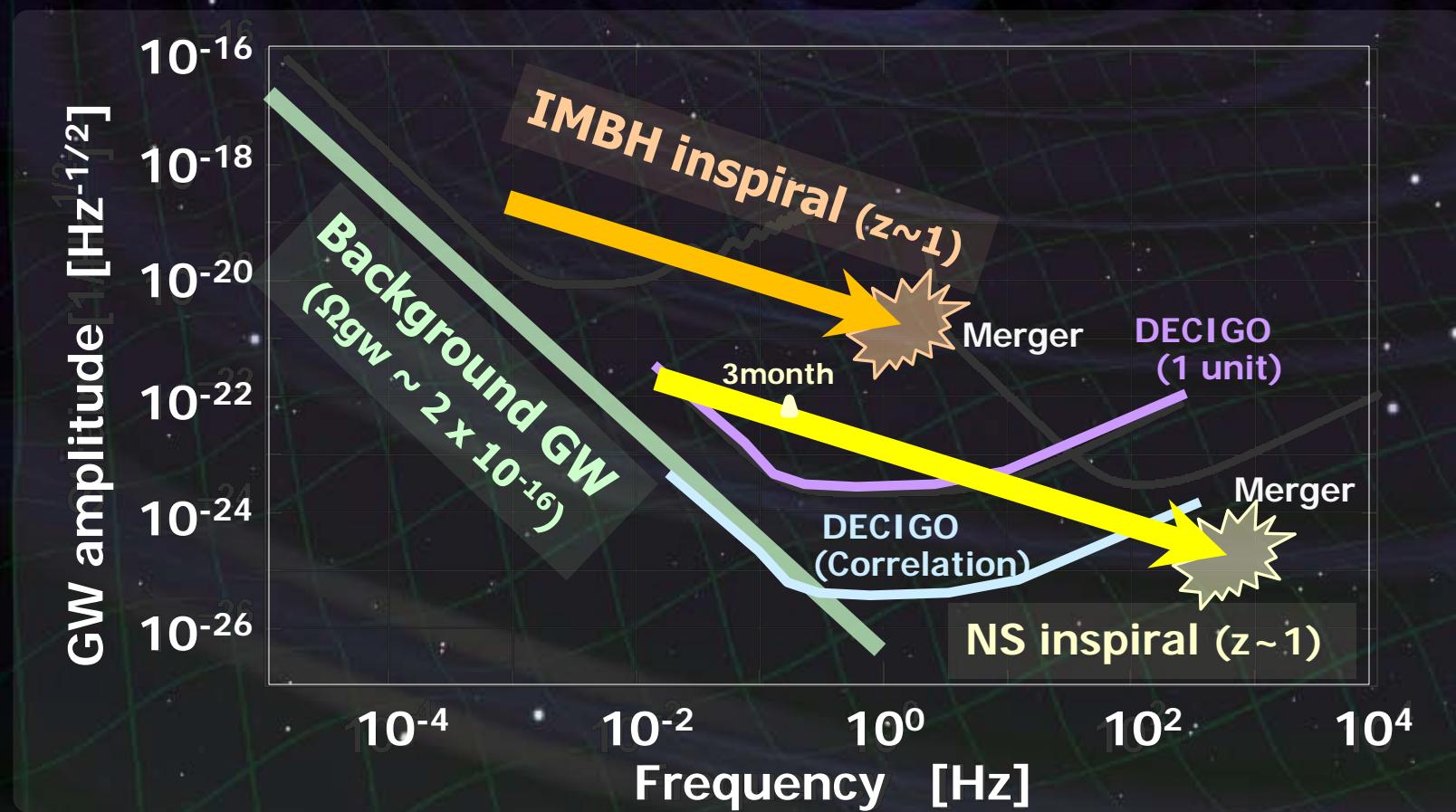
NS binary inspiral

Stochastic background

Galaxy formation (Massive BH)

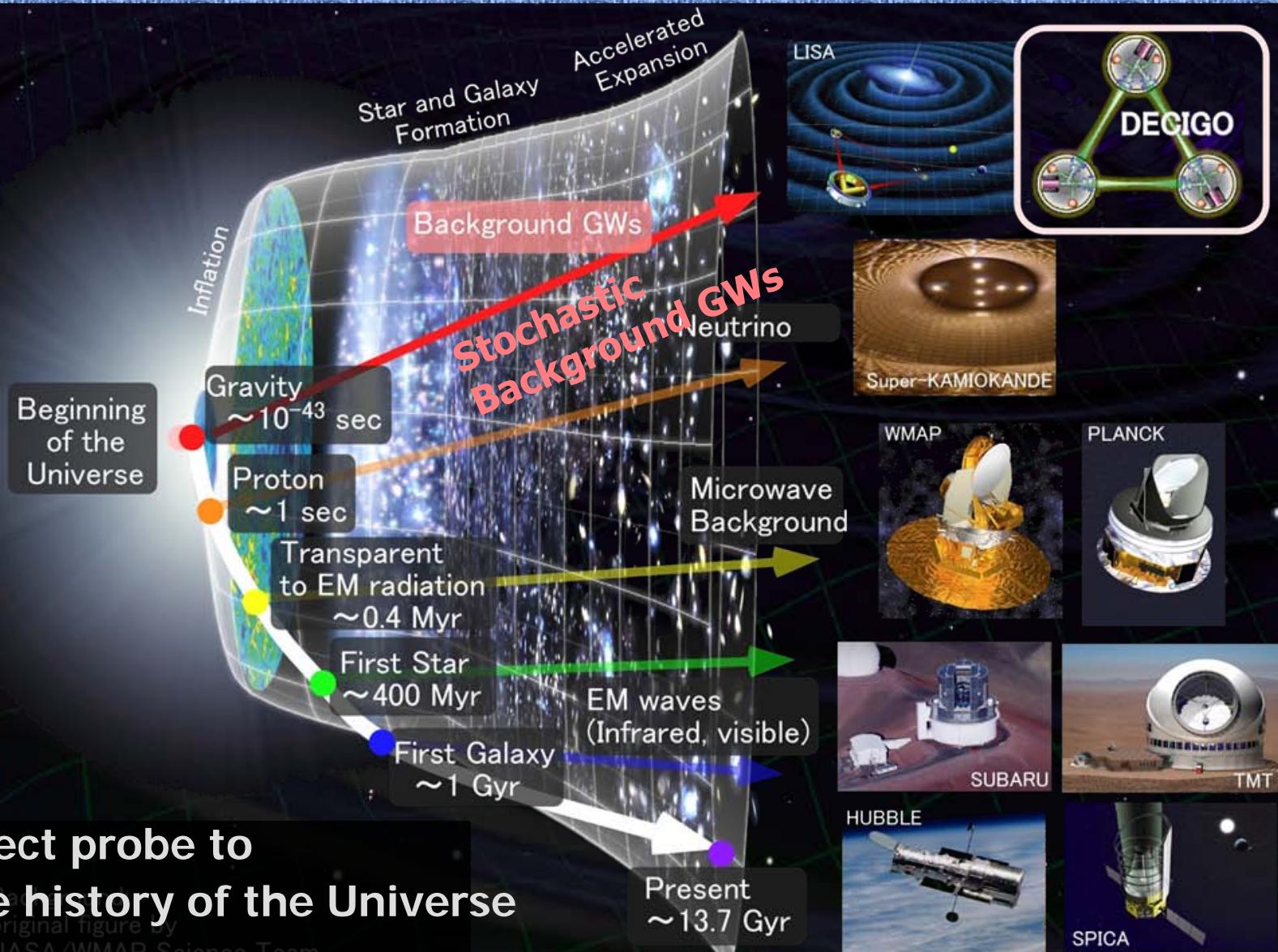
Cosmology (Inflation, Dark energy)

Fundamental physics



Characterization of inflation

DECIGO



Dark energy



DECIGO will observe

5×10^4 NS binaries for $z < 1$



Precise 'clock' at cosmological distance

'Standard Siren'

Relationship between
distance and redshift

Distance: chirp waveform

Redshift: host galaxy

→ Information on acceleration
of expansion of the universe

Determine cosmological parameters

Absolute and independent measurement



Seto, Kawamura, Nakamura,
PRL 87, 221103 (2001)

Angular resolution
 $\sim 10 \text{arcmin}^2$ (1 detector)
 $\sim 10 \text{arcsec}^2$ (3 detectors)

at $z=1$

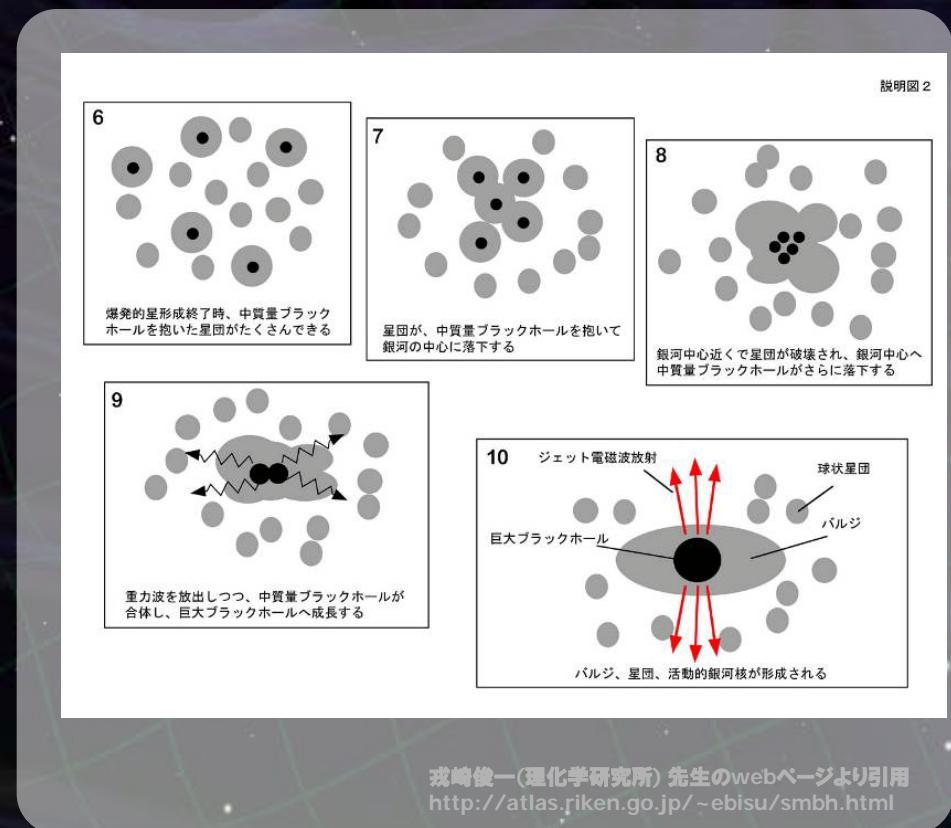
Galaxy formation



DECIGO will observe
Intermediate-mass BH (IMBH)
binary merger with
 $\text{SNR} > 10^3$ for $z \sim 10$ source



Information on the
formation of
Supermassive BHs
at the center of galaxies



- Verification of the alternative theories of gravity
Test Brans-Dicke theory by NS/BH binary evolution
→ Stronger constraint by 10^4 times

K. Yagi and T. Tanaka, Prog. Theor. Phys. 123, 1069 (2010)

- Black hole dark matter
Gravitational collapse of the primordial density fluctuations
→ Primordial black holes (PBHs)
as a candidate of dark matter

R. Saito and J. Yokoyama, Phys. Rev. Lett. 102 161101 (2009)

- Neutron-star physics
Determine mass of 10^5 NSs per year
→ Constrain the EOS of NS
Formation process of NS from the spectrum

1. DECIGO

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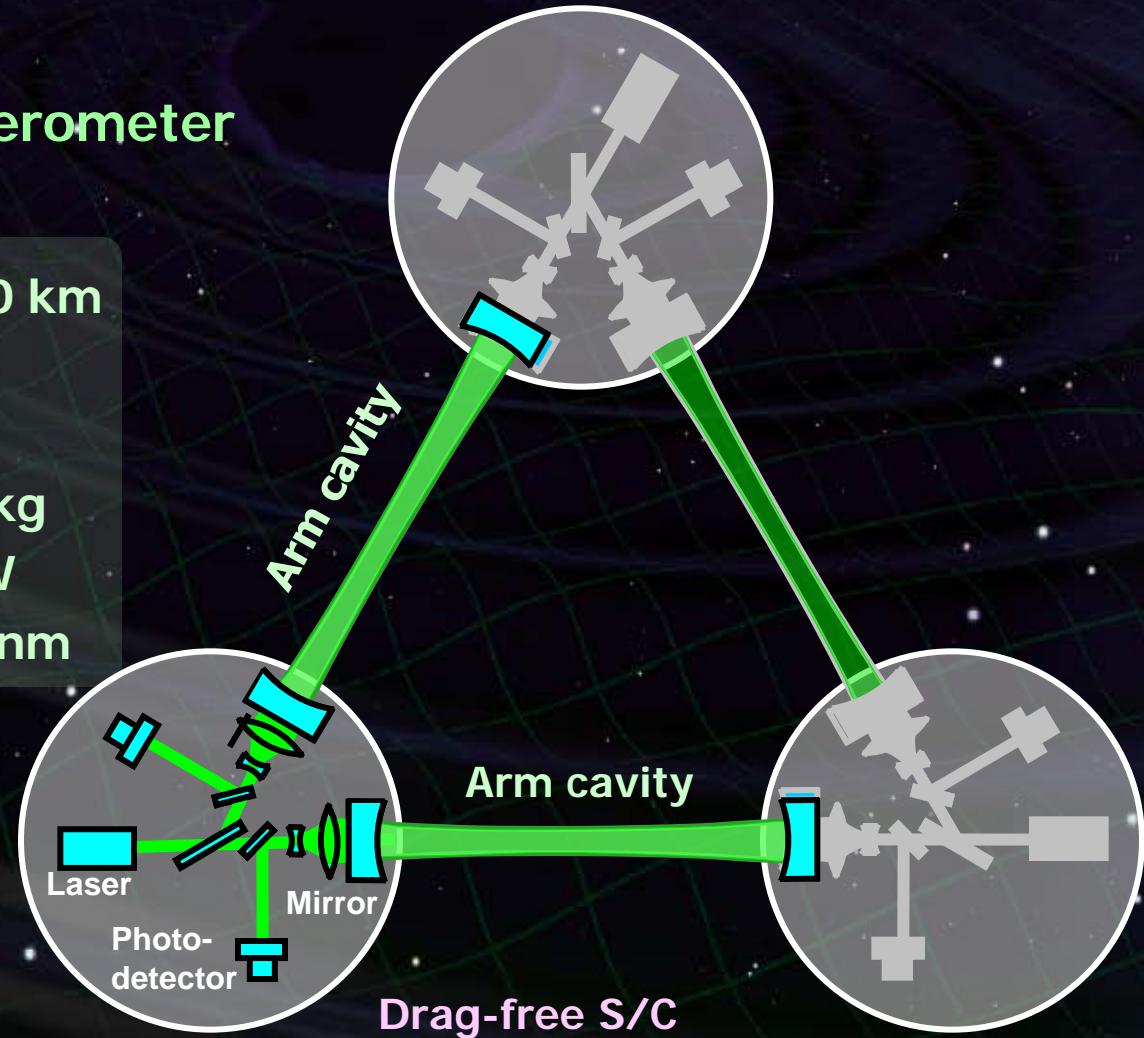
Pre-Conceptual Design



Interferometer Unit: Differential FP interferometer

Arm length: 1000 km
Finesse: 10
Mirror diameter: 1 m
Mirror mass: 100 kg
Laser power: 10 W
Laser wavelength: 532 nm

S/C: drag free
3 interferometers

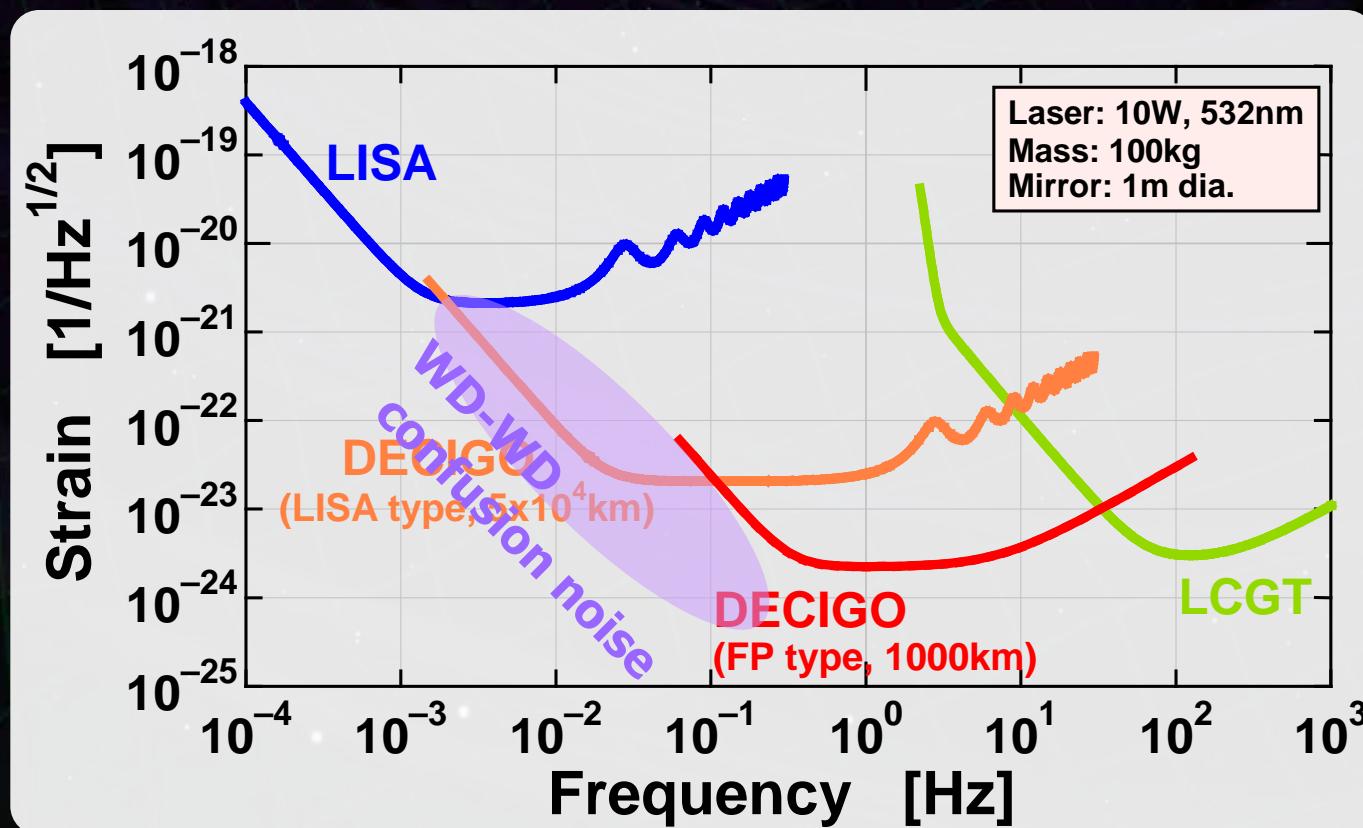


Transponder type vs Direct-reflection type

Compare : Sensitivity curves and Expected Sciences



Decisive factor: Binary confusion noise



Arm length

Cavity arm length : Limited by diffraction loss

Effective reflectivity ($\text{TEM}_{00} \rightarrow \text{TEM}_{00}$)

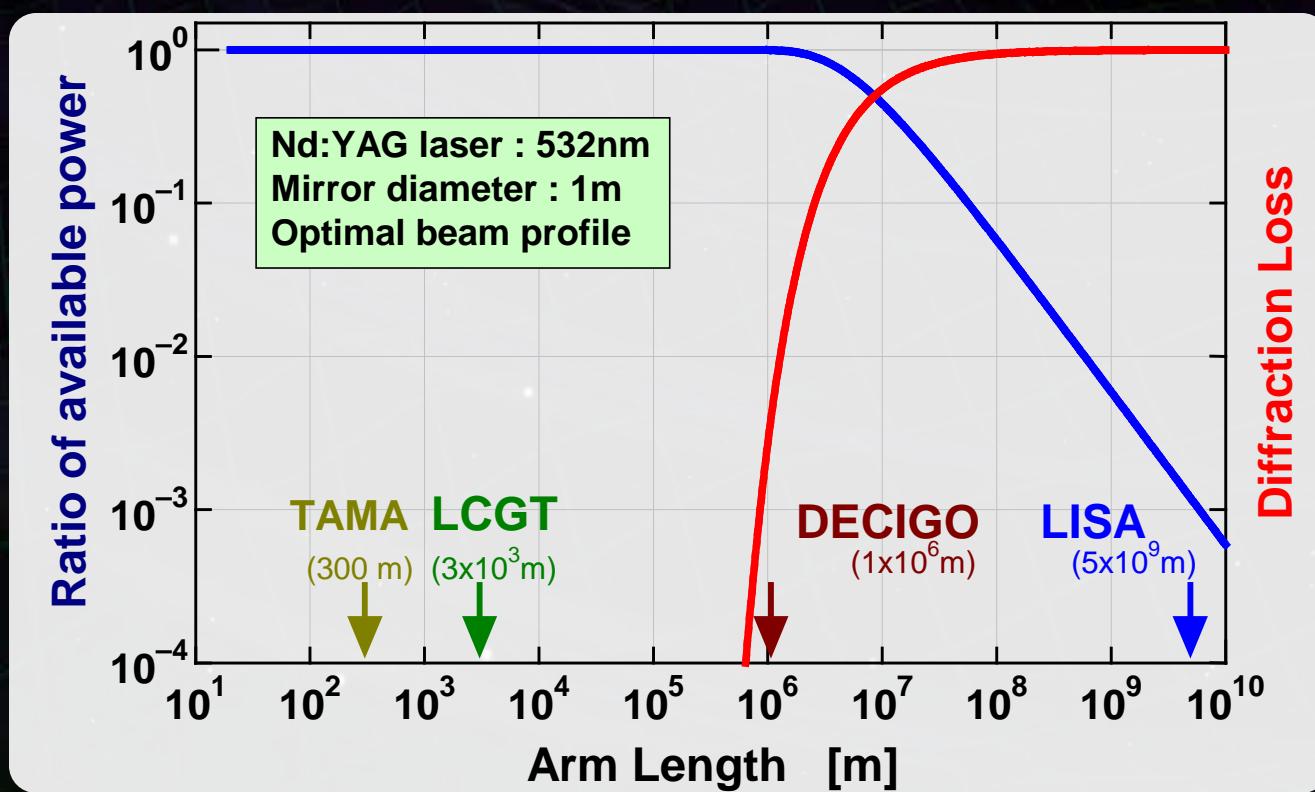
Laser wavelength : 532nm

Mirror diameter: 1m

Optimal beam size



1000 km
is almost max.



Foreground Cleaning



DECIGO obs. band: free from WD binary foreground
→ Open for cosmological observation

DECIGO will watch
 $\sim 10^5$ NS binaries

➡ Foreground for GWB

In principle, possible
to remove them.

Require accurate waveform
→ $\Delta m/m < \sim 10^{-7} \%$

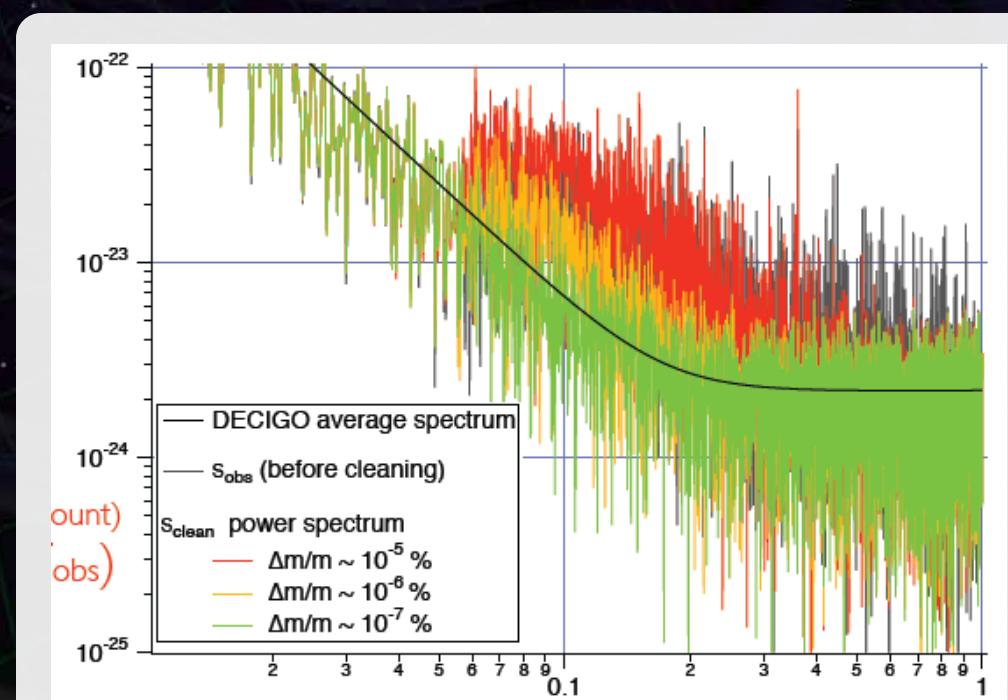


Fig: N. Kanda

Cavity and S/C control

Cavity length change

PDH error signal → Mirror position (and Laser frequency)

Relative motion between mirror and S/C

Local sensor → S/C thruster

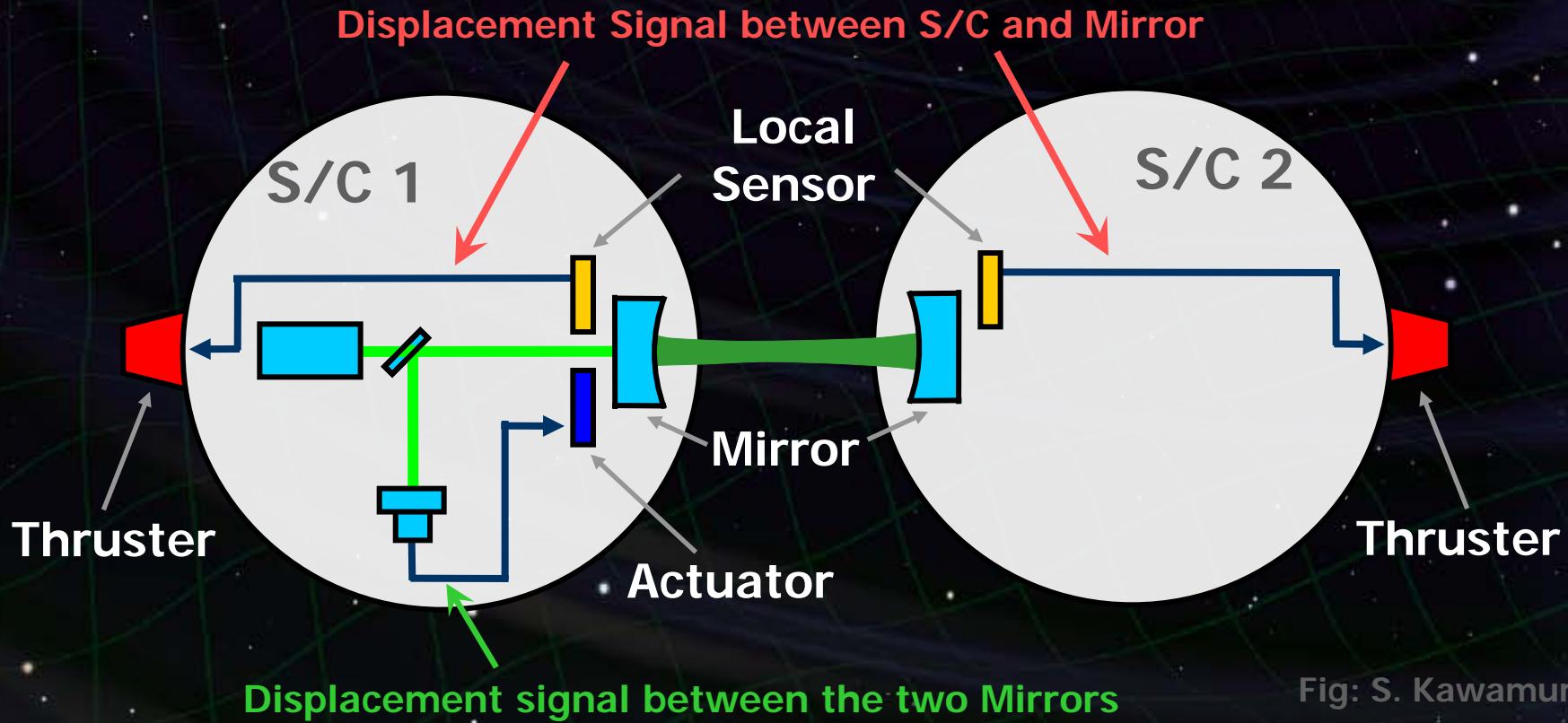


Fig: S. Kawamura

Requirements



Sensor Noise

Shot noise $3 \times 10^{-18} \text{ m/Hz}^{1/2}$ (0.1 Hz)

⇒ $\times 10$ of LCGT in phase noise

Other noises should be well below the shot noise

Laser freq. noise: $1 \text{ Hz/Hz}^{1/2}$ (1Hz)

Stab. Gain 10^5 , CMRR 10^5

Acceleration Noise

Force noise $4 \times 10^{-17} \text{ N/Hz}^{1/2}$ (0.1 Hz)

⇒ $\times 1/50$ of LISA

External force sources

Fluctuation of magnetic field, electric field, gravitational field, temperature, pressure, etc.

Orbit and Constellation

Candidate of orbit:

Record-disk orbit around the Sun

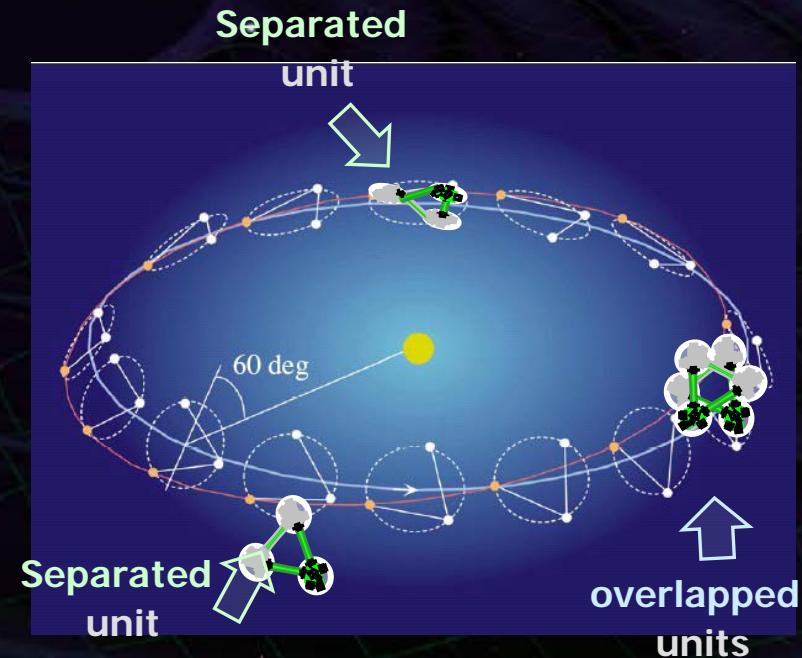
Relative acc. $4 \times 10^{-12} \text{ m/s}^2$
(Mirror force $\sim 10^{-9} \text{ N}$)

Constellation

4 interferometer units

2 overlapped units → Cross correlation

2 separated units → Angular resolution



Roadmap



Figure: S.Kawamura

	2010	11	12	13	14	15	16	17	18	19	20	21	22	23.	24	25	26	27	28	29
Mission	SDS-1/SWIM					DECIGO Pathfinder (DPF)						Pre-DECIGO					DECIGO			
Objective	Space test of key tech. GW observation						Detect GW with min. spec FP between S/C										GW astronomy			
Design	Single small satellite Short FP interferometer						3 S/C 1 interferometer unit										3 S/C x 3-4 units			

The diagram illustrates the progression of the DECIGO mission over time. It starts with the SDS-1/SWIM mission in 2010, which involved a single small satellite in low Earth orbit. This was followed by the DECIGO Pathfinder (DPF) mission in 2015, which involved a single satellite with a green triangle icon. The Pre-DECIGO phase began around 2020, featuring a cluster of three satellites connected by green lines, with a purple triangle icon. Finally, the full DECIGO mission is shown for the years 2026-29, consisting of a triangular arrangement of four satellites, each with a red triangle icon, connected by green lines.

Organization



PI: Kawamura (NAOJ)
Deputy: Ando (Kyoto)

Executive Committee

Kawamura (NAOJ), Ando (Kyoto), Seto (Kyoto), Nakamura (Kyoto), Tsubono (Tokyo), Tanaka (Kyoto), Funaki (ISAS), Numata (Maryland), Sato (Hosei), Kanda (Osaka city), Takashima (ISAS), Ioka (KEK), Yokoyama (Tokyo)

Pre-DECIGO
Sato (Hosei)

Detector
Akutsu (NAOJ)
Numata (Maryland)

Science, Data
Tanaka (Kyoto)
Seto (Kyoto)
Kanda (Osaka city)

Satellite
Funaki (ISAS)

DECIGO pathfinder
Leader: Ando (Kyoto)

Design phase

Mission phase

Detector
Sato (Hosei)
Ueda (NAOJ)
Aso (Tokyo)

Laser
Musha (ILS)
Ueda (ILS)

Drag free
Moriwaki (Tokyo)
Sakai (ISAS)

Thruster
Funaki (ISAS)

Bus
Takashima (ISAS)

Data
Kanda (Osaka city)

Collaboration and support



- Supports from LISA

- Technical advices from LISA/LPF experiences

- Support Letter for DECIGO/DPF, Joint workshop (2008.11)

- Collab. with Stanford univ. group

- Drag-free control of DECIGO/DPF

- UV LED Charge Management System for DPF

- Collab. with NASA/GSFC

- Fiber Laser , started discussion

- Collab. with JAXA navigation-control section

- formation flight of DECIGO, DPF drag-free control

- Research Center for the Early Universe (RESCEU), Univ. of Tokyo

- Support DECIGO as ones of main projects (2009.4-)

- Advanced technology center (ATC) of NAOJ

- Will make it a main nucleus of DPF

LCGT and DECIGO



LCGT (~2016)

Terrestrial Detector

→ High frequency events

Target: GW detection

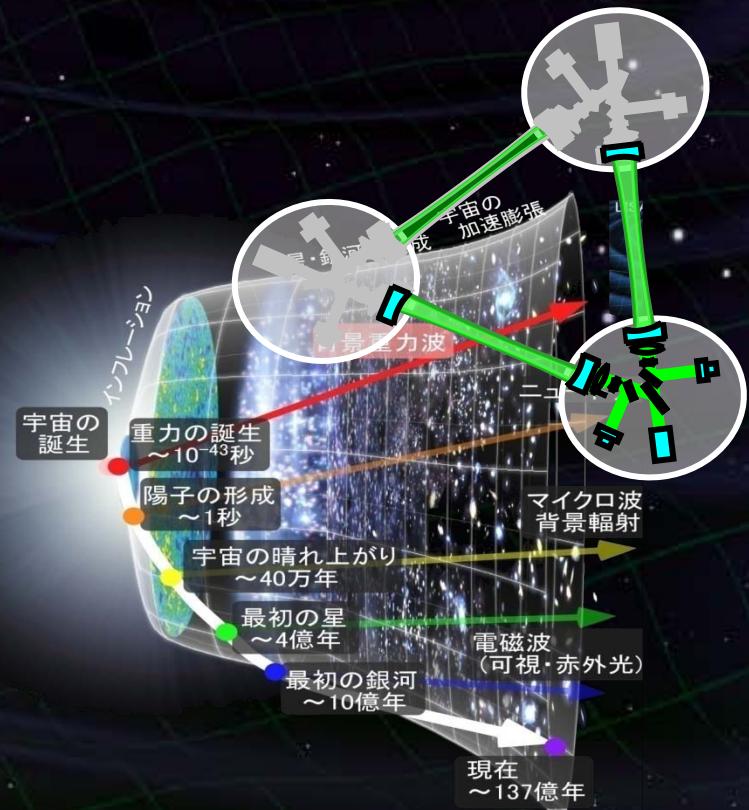


DECIGO (~2027)

Space observatory

→ Low frequency sources

Target: GW astronomy



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Figure: S.Kawamura

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	Single small satellite Short FP interferometer												3 S/C 1 interferometer unit							

DECIGO Pathfinder (DPF)

First milestone mission for DECIGO
Shrink arm cavity

DECIGO 1000km → DPF 30cm

Single satellite

(Payload ~1m³ , 350kg)

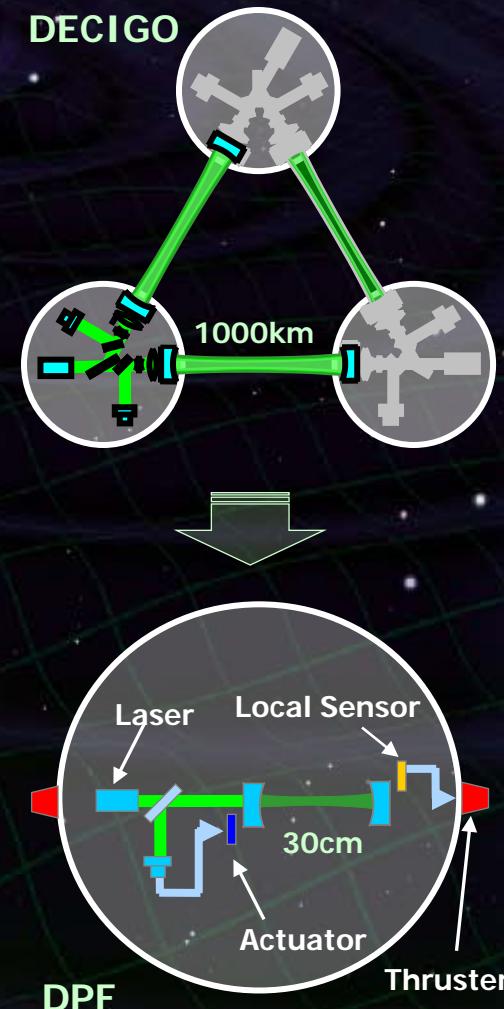
Low-earth orbit

(Altitude 500km, sun synchronous)

30cm FP cavity with 2 test masses

Stabilized laser source

Drag-free control



DPF satellite



DPF Payload

Size : 950mm cube

Weight : 150kg

Power : 130W

Data Rate: 800kbps

Mission thruster x12

Power Supply
SpW Comm.

Satellite Bus

('Standard bus' system)

Size :

950x950x1100mm

Weight : 200kg

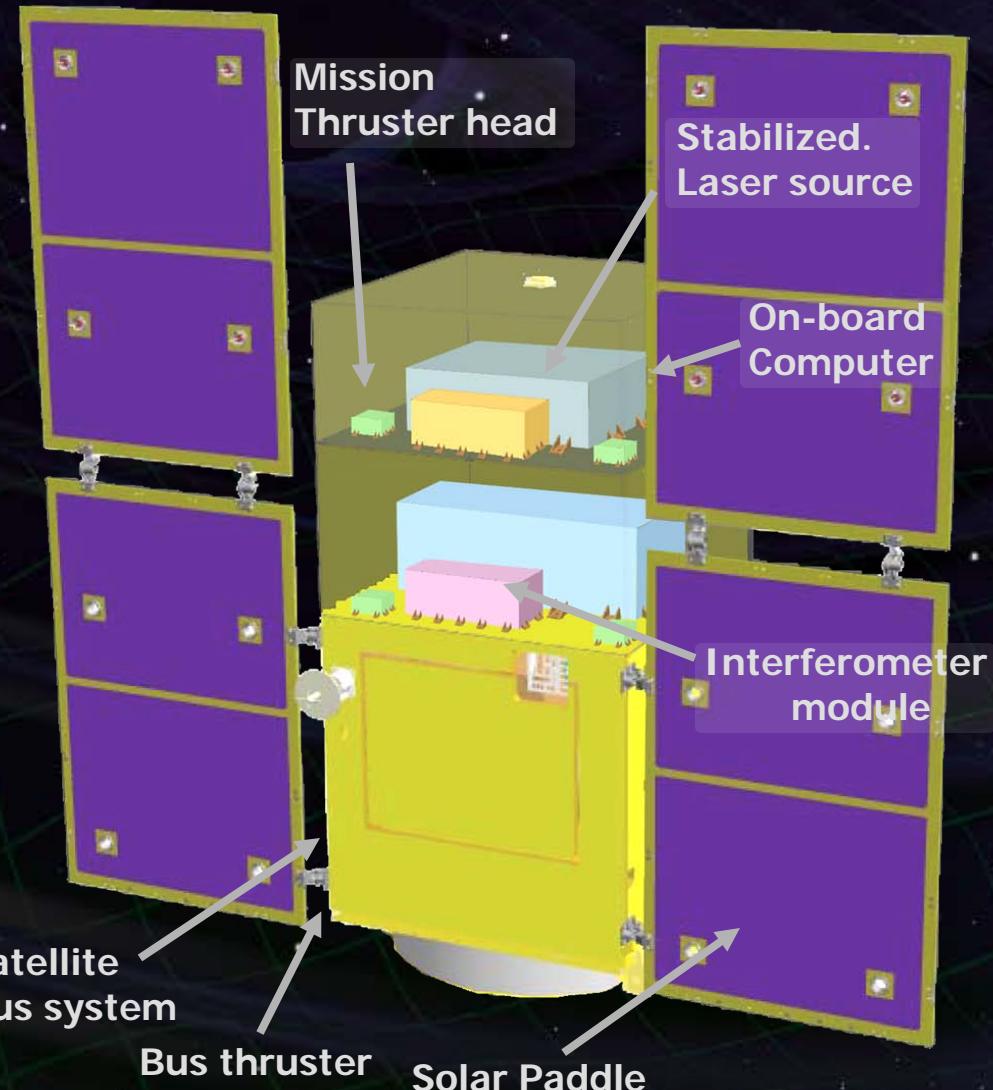
SAP : 960W

Battery: 50AH

Downlink : 2Mbps

DR: 1GByte

3N Thrusters x 4

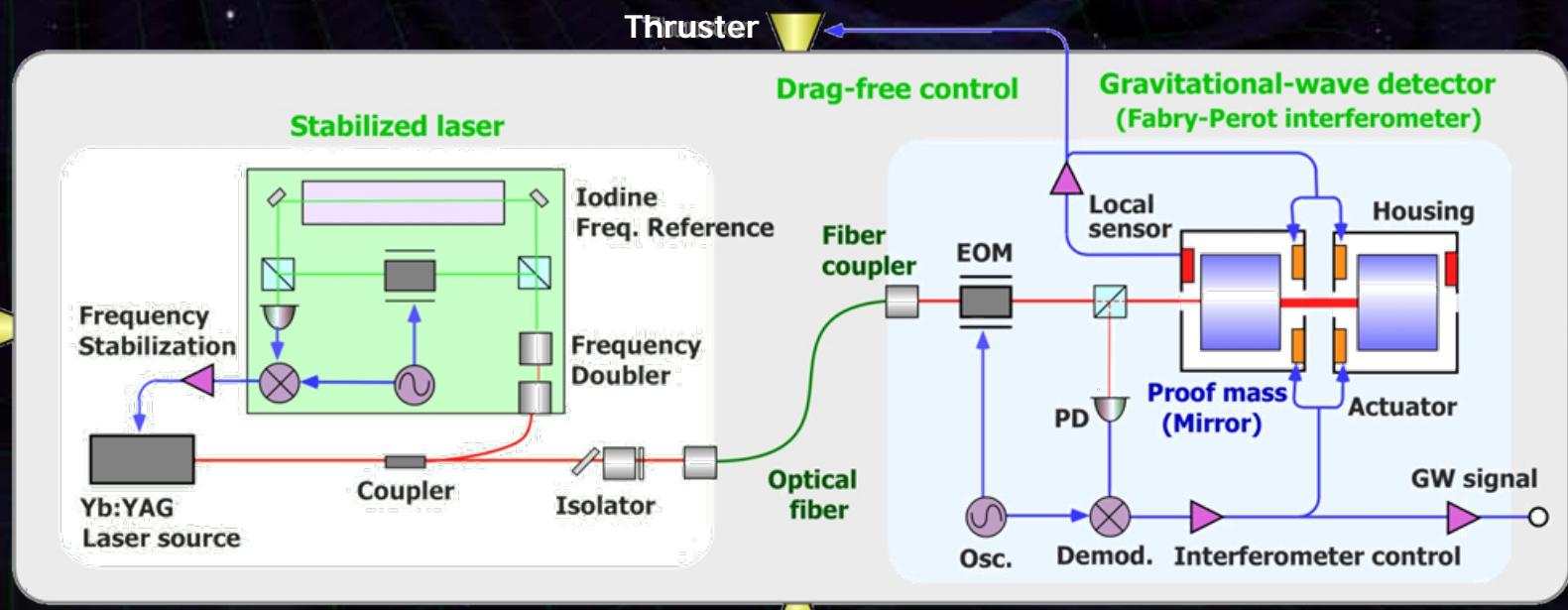


DPF mission payload



Mission weight : ~150kg
Mission space : ~95 x 95 x 90 cm

Drag-free control
Local sensor signal
→ Feedback to thrusters



Laser source
Yb:YAG laser (1030nm)
Power : 25mW
Freq. stab. by Iodine abs. line

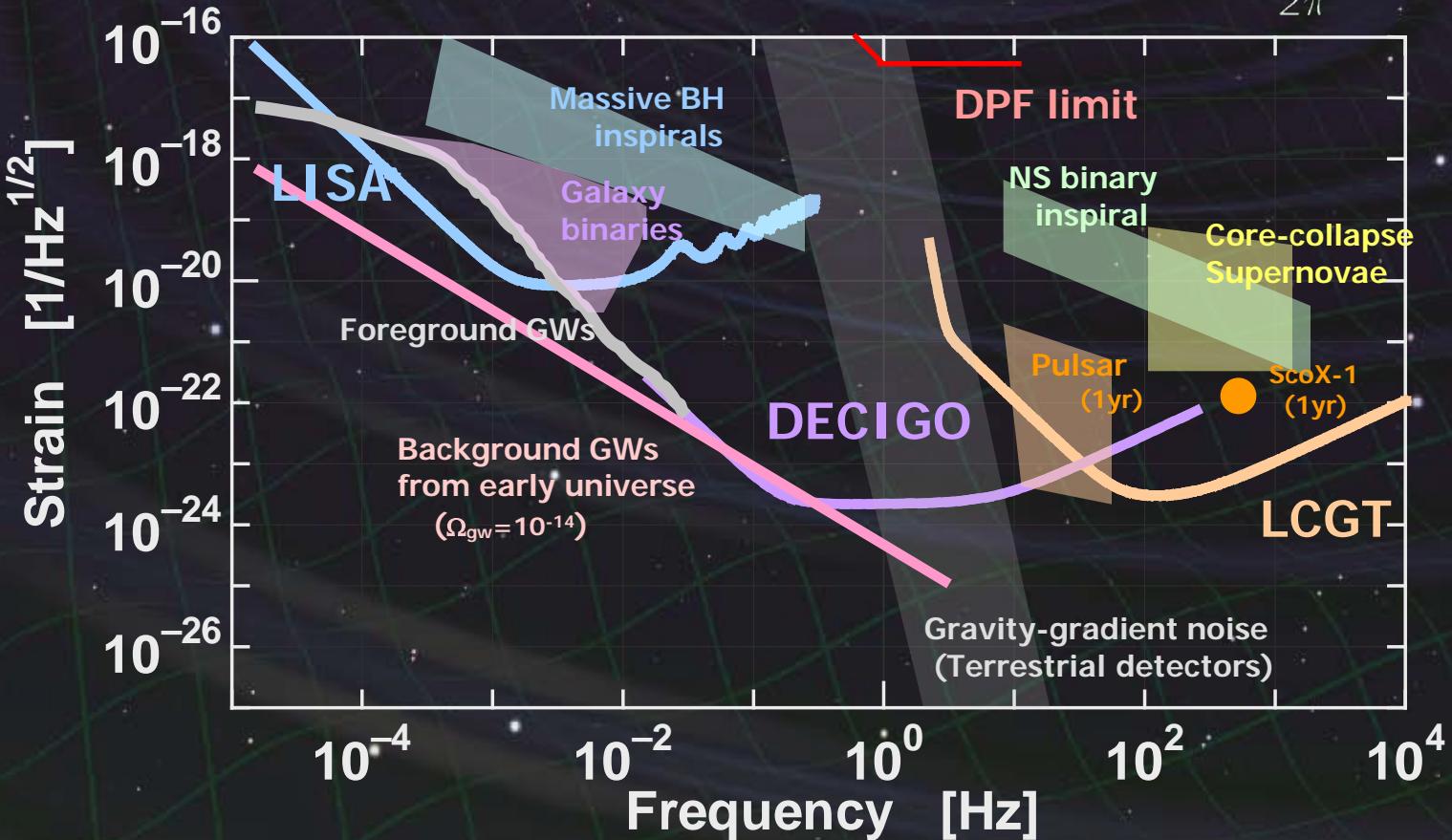
Fabry-Perot interferometer
Finesse : 100
Length : 30cm
Test mass : ~1kg
Signal extraction by PDH

DPF sensitivity

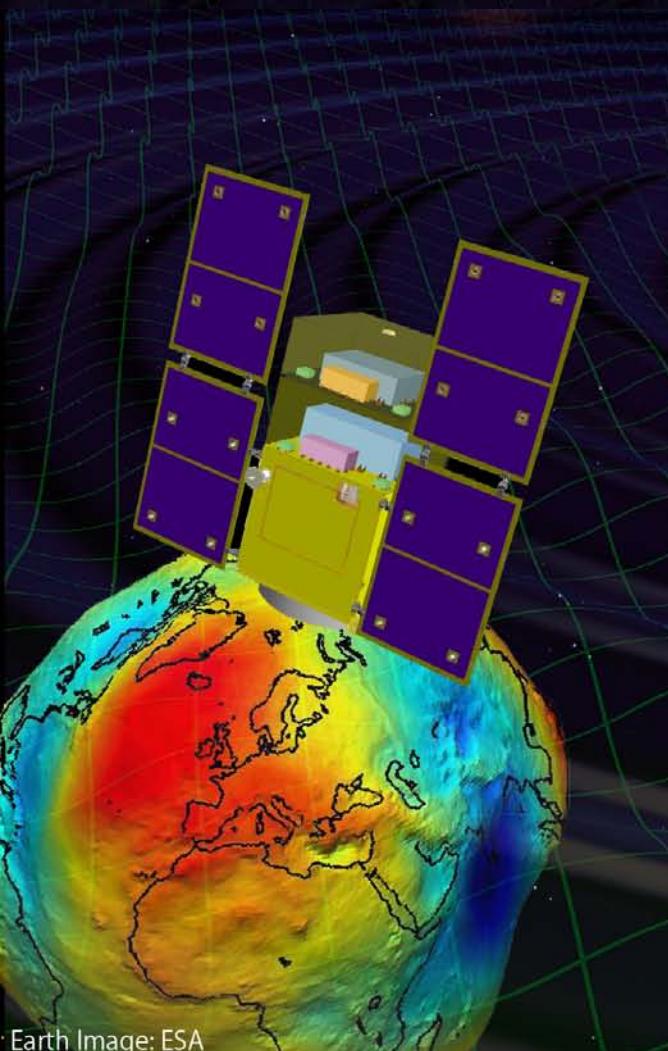
DPF sensitivity $h \sim 2 \times 10^{-15} \text{ Hz}^{1/2}$

(x10 of quantum noises)

$$f \sim \frac{1}{2\pi} \sqrt{GM/R^3}$$



Targets of DPF



Earth Image: ESA

Scientific observations

Gravitational Waves from BH mergers
→ BH formation mechanism

Gravity of the Earth
→ Geophysics, Earth environment

Science technology

Space demonstration for DECIGO
→ Most tech. with single satellite
(IFO, Laser, Drag-free)

Precision measurement in orbit
→ IFO measurement
under stable zero-gravity

DPF Targets



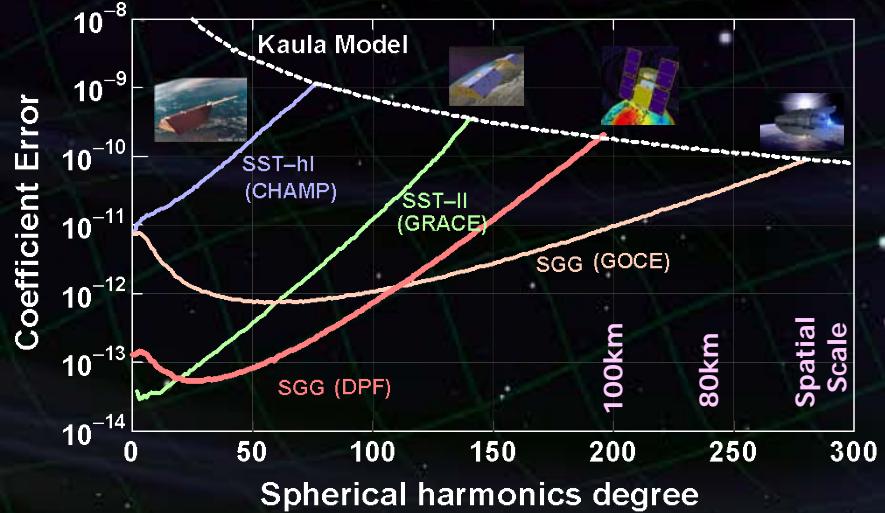
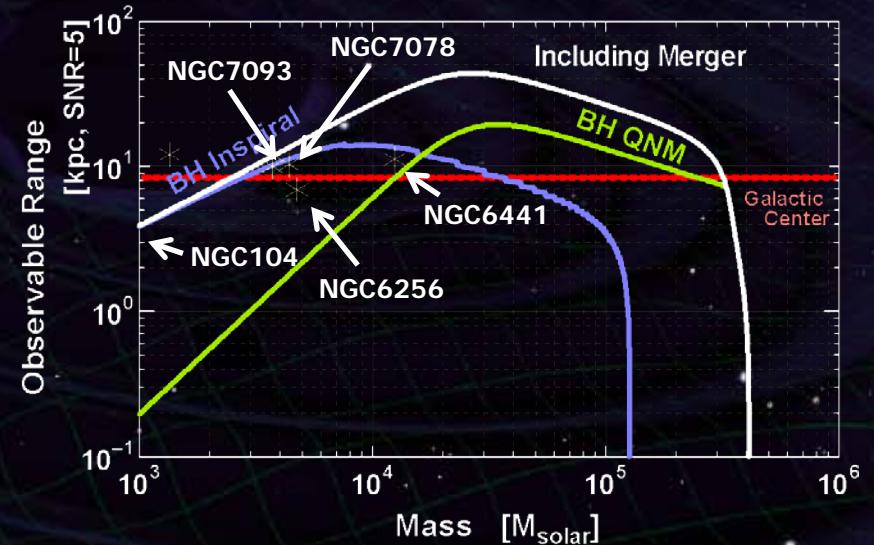
Astronomical observation

GW from merger of IMBHs
→ Formation mechanism
of supermassive BHs

~ 30 GCs within DPF range

Observation of the earth
Gravitational potential
→ Shape of the earth
Environment monitor

Comparable sensitivity
with other missions



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DPF mission status



DPF : One of the candidate of
JAXA's small satellite series



At least 3 satellite in 5 years with
Standard Bus + M-V follow-on rocket



SPRINT-A/EXCEED 撮像団(池下章裕氏作)

SPRINT-A /EXCEED
UV telescope mission

1st mission (2012) : SPRINT-A/EXCEED

2nd mission (~2013/14) : ERG

DPF survived until final two

3rd mission (~2015/16) : TBD



Next-generation
Solid rocket booster (M-V FO)
Fig. by JAXA

DPF is one of the strongest
candidates of the 3rd mission

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	Single small satellite Short FP interferometer												3 S/C 1 interferometer unit							

SWIM launch and operation

DECT GO

Tiny GW detector module
Launched in Jan. 23, 2009
⇒ In-orbit operation

TAM: Torsion Antenna Module with free-falling test mass
(Size : 80mm cube, Weight : ~500g)

Test mass

~47g Aluminum, Surface polished
Small magnets for position control

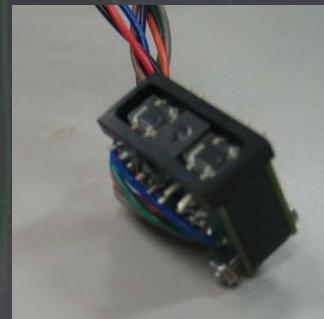
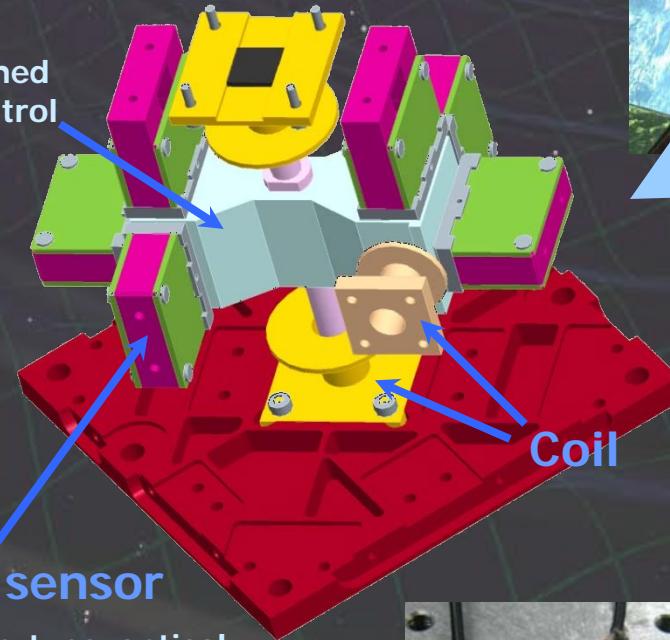


Photo sensor

Reflective-type optical displacement sensor
Separation to mass ~1mm
Sensitivity ~ 10^{-9} m/Hz $^{1/2}$
6 PSs to monitor mass motion

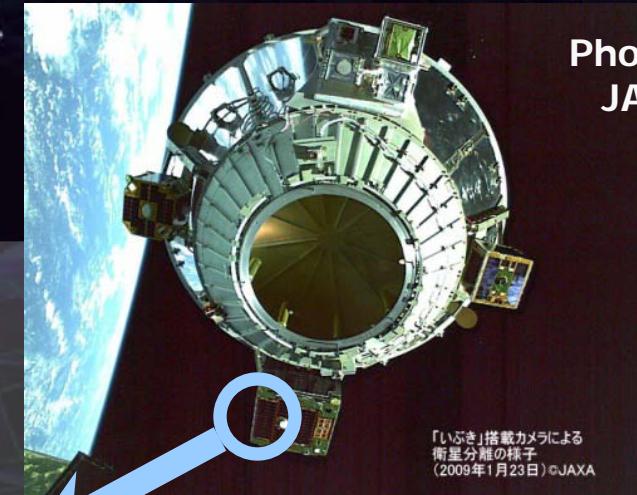
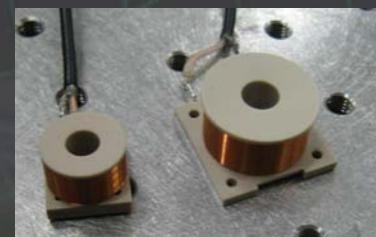


Photo:
JAXA

SWIM observation



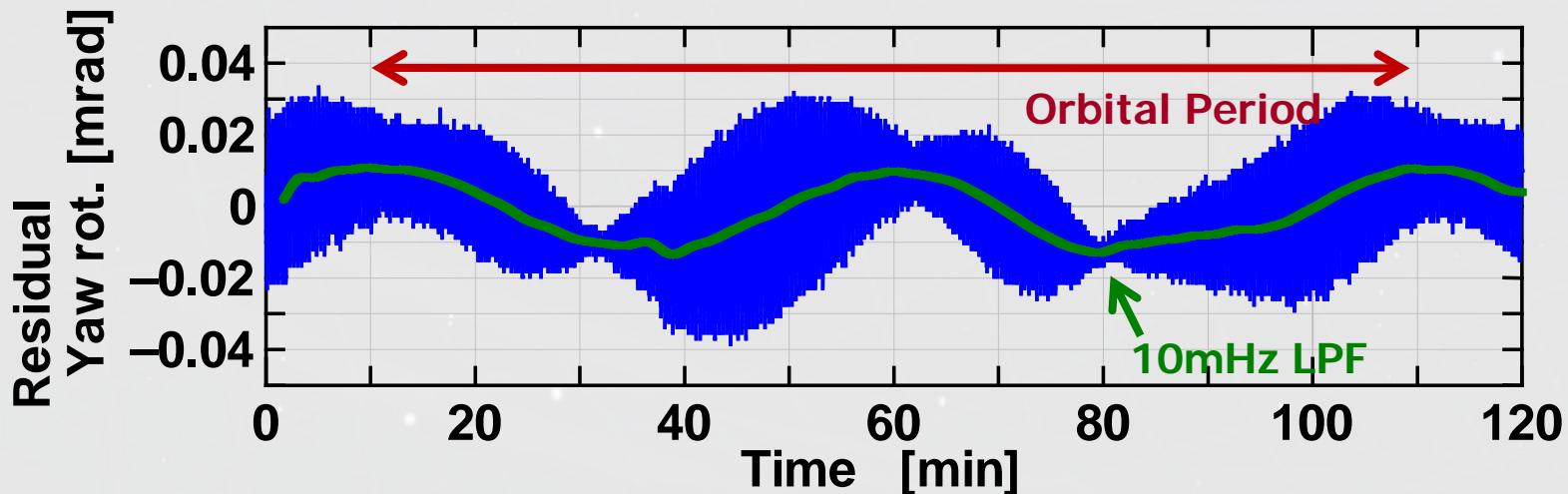
Observation by SWIM

Jun 17, 2010 ~120 min. operation

July 15, 2010 ~240 min. operation

Ground-based detectors were
operated at the same period.

⇒ Data analysis



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DECIGO : Fruitful Sciences

Very beginning of the Universe

Dark energy

Galaxy formation

DECIGO Pathfinder

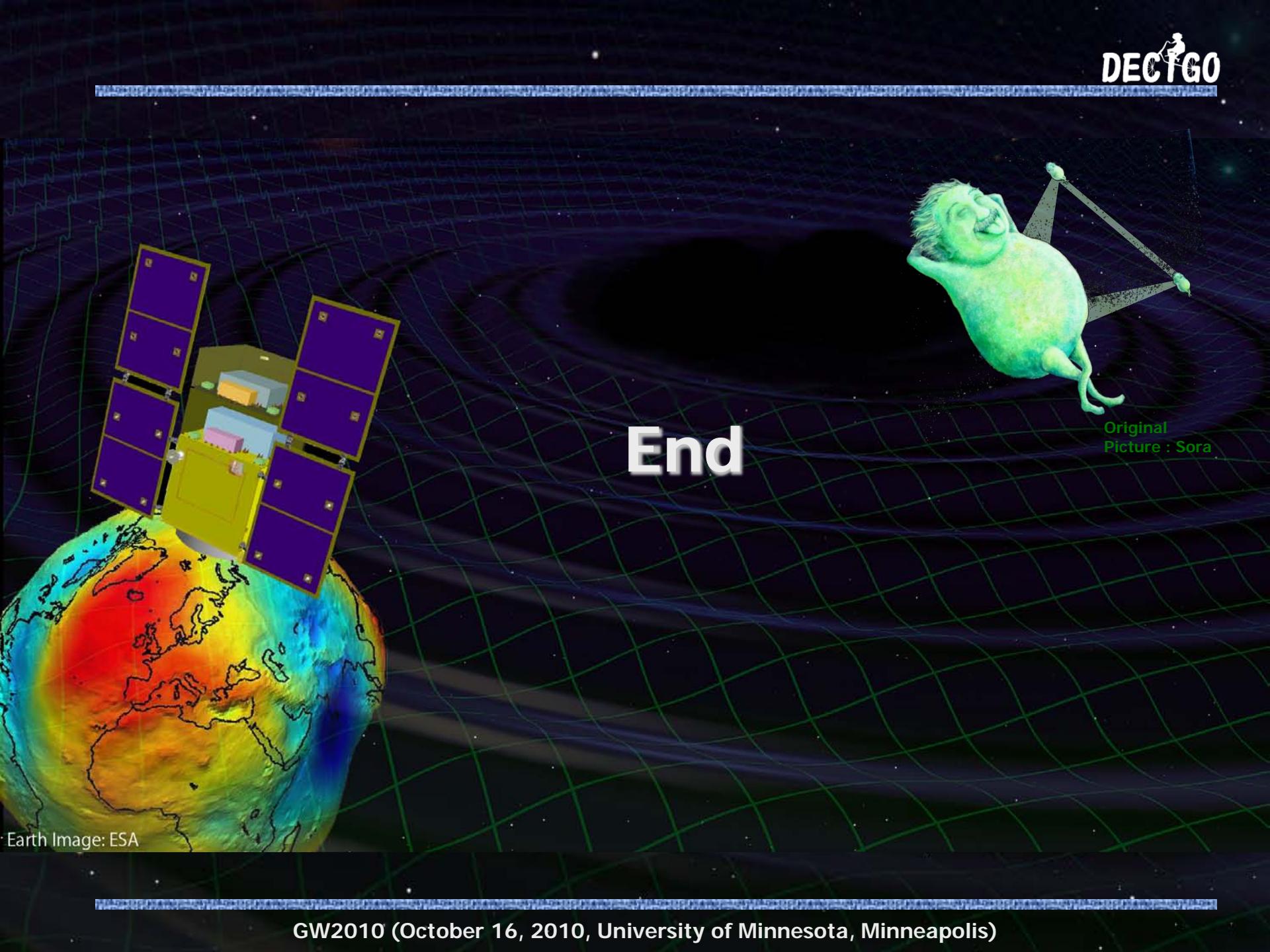
Important milestone for DECIGO

Strong candidate of JAXA's satellite series

**SWIM – under operation in orbit
first precursor to space!**

End

Original
Picture : Sora



Earth Image: ESA

GW2010 (October 16, 2010, University of Minnesota, Minneapolis)