

# DECIGO and Pathfinder Missions

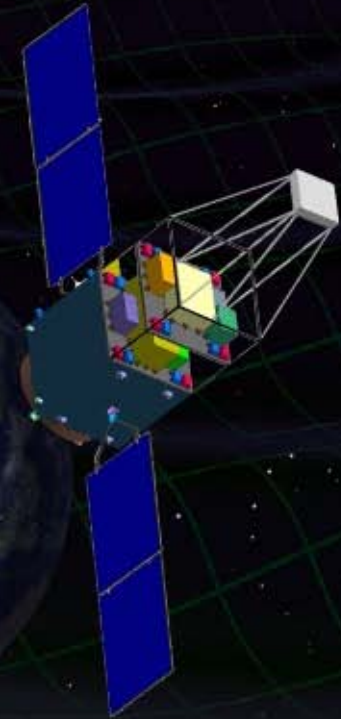


Original  
Picture : Sora

**Masaki Ando**

(Department of Physics, Kyoto University)

**Seiji Kawamura**, Takashi Nakamura, Kimio Tsubono, Takahiro Tanaka, Ikkoh Funaki, Naoki Seto, Kenji Numata, Shuichi Sato, Nobuyuki Kanda, Takeshi Takashima, Kunihiro Ioka, Kazuhiro Agatsuma, Tomotada Akutsu, Tomomi Akutsu, Koh-suke Aoyanagi, Koji Arai, Yuta Arase, Akito Araya, Hideki Asada, Yoichi Aso, Takeshi Chiba, Toshikazu Ebisuzaki, Motohiro Enoki, Yoshiharu Eriguchi, Masa-Katsu Fujimoto, Ryuichi Fujita, Mitsuhiro Fukushima, Toshifumi Futamase, Katsuhiko Ganzu, Tomohiro Harada, Tatsuaki Hashimoto, Kazuhiro Hayama, Wataru Hikida, Yoshiaki Himemoto, Hisashi Hirabayashi, Takashi Hiramatsu, Feng-Lei Hong, Hideyuki Horisawa, Mizuhiko Hosokawa, Kiyotomo Ichiki, Takeshi Ikegami, Kaiki T. Inoue, Koji Ishidoshiro, Hideki Ishihara, Takehiko Ishikawa, Hideharu Ishizaki, Hiroyuki Ito, Yousuke Itoh, Shogo Kamagasako, Nobuki Kawashima, Fumiko Kawazoe, Hiroyuki Kirihara, Naoko Kishimoto, Kenta Kiuchi, Shiho Kobayashi, Kazunori Kohri, Hiroyuki Koizumi, Yasufumi Kojima, Keiko Kokeyama, Wataru Kokuyama, Kei Kotake, Yoshinori Kozai, Hideaki Kudoh, Hiroo Kunimori, Hitoshi Kuninaka, Kazuaki Kuroda, Kei-ichi Maeda, Hideo Mitsuhashi, Yasushi Mino, Osamu Miyakawa, Shinji Miyoki, Mutsuko Y. Morimoto, Tomoko Morioka, Toshiyuki Morisawa, Shigenori Moriwaki, Shinji Mukohyama, Mitsuru Musha, Shigeo Nagano, Isao Naito, Noriyasu Nakagawa, Kouji Nakamura, Hiroyuki Nakano, Kenichi Nakao, Shinichi Nakasuka, Yoshinori Nakayama, Erina Nishida, Kazutaka Nishiyama, Atsushi Nishizawa, Yoshito Niwa, Masatake Ohashi, Naoko Ohishi, Masashi Ohkawa, Akira Okutomi, Kouji Onozato, Kenichi Oohara, Norichika Sago, Motoyuki Saijo, Masaaki Sakagami, Shin-ichiro Sakai, Shihori Sakata, Misao Sasaki, Takashi Sato, Masaru Shibata, Hisaaki Shinkai, Kentaro Somiya, Hajime Sotani, Naoshi Sugiyama, Yudai Suwa, Hideyuki Tagoshi, Kakeru Takahashi, Keitaro Takahashi, Tadayuki Takahashi, Hirotaka Takahashi, Ryuichi Takahashi, Ryutaru Takahashi, Takamori Akiteru, Tadashi Takano, Keisuke Taniguchi, Atsushi Taruya, Hiroyuki Tashiro, Mitsuru Tokuda, Masao Tokunari, Morio Toyoshima, Shinji Tsujikawa, Yoshiki Tsunesada, Ken-ichi Ueda, Masayoshi Utashima, Hiroshi Yamakawa, Kazuhiro Yamamoto, Toshitaka Yamazaki, Jun'ichi Yokoyama, Chul-Moon Yoo, Shijun Yoshida, Tajoh Yoshino





---

# **1. DECIGO**

Overview and Science

Pre-conceptual Design

# **2. DECIGO Pathfinder**

Overview and Science

Design and Status

# **3. Summary**



# **1. DECIGO**

**Overview and Science**

**Pre-conceptual Design**

# **2. DECIGO Pathfinder**

**Overview and Science**

**Design and Status**

# **3. Summary**



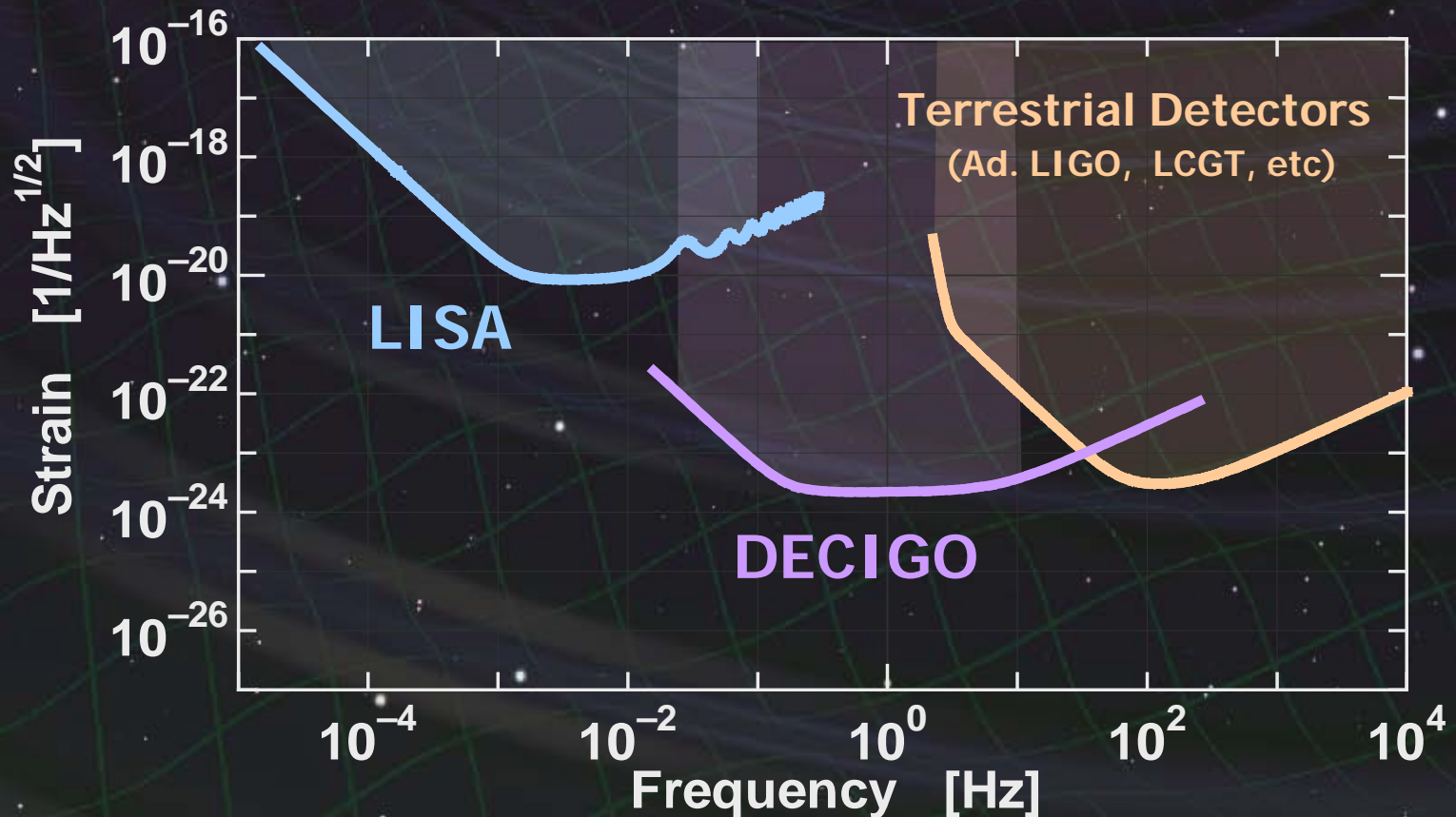
# DECIGO

**DECIGO** (Deci-hertz interferometer Gravitational wave Observatory)

Space GW antenna (~2024)  
Obs. band around 0.1 Hz



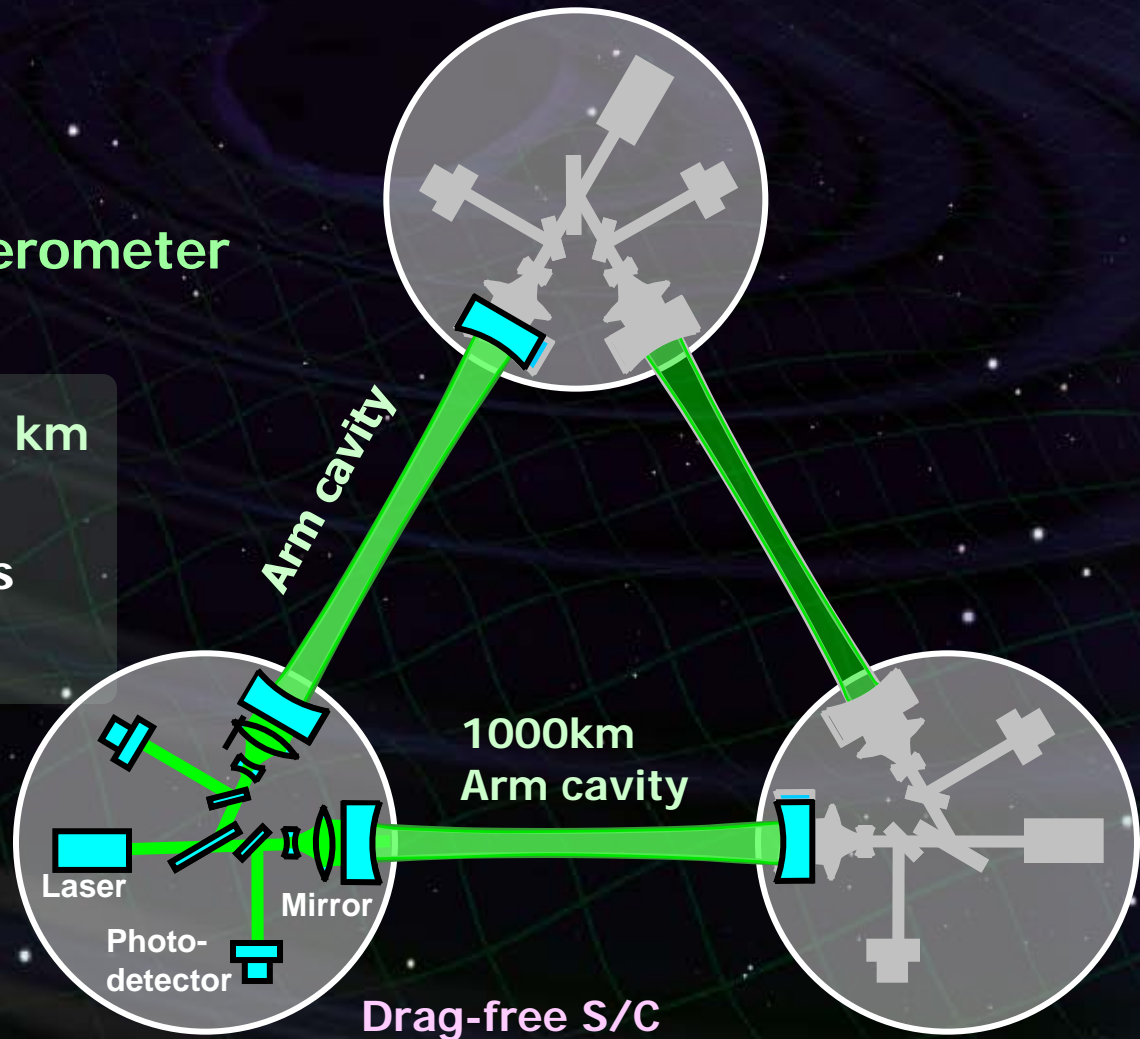
'Bridge' the obs.gap between  
**LISA** and **Terrestrial detectors**



# DECIGO Interferometer

Interferometer Unit:  
Differential FP interferometer

Baseline length: 1000 km  
3 S/C formation flight  
3 FP interferometers  
Drag-free control



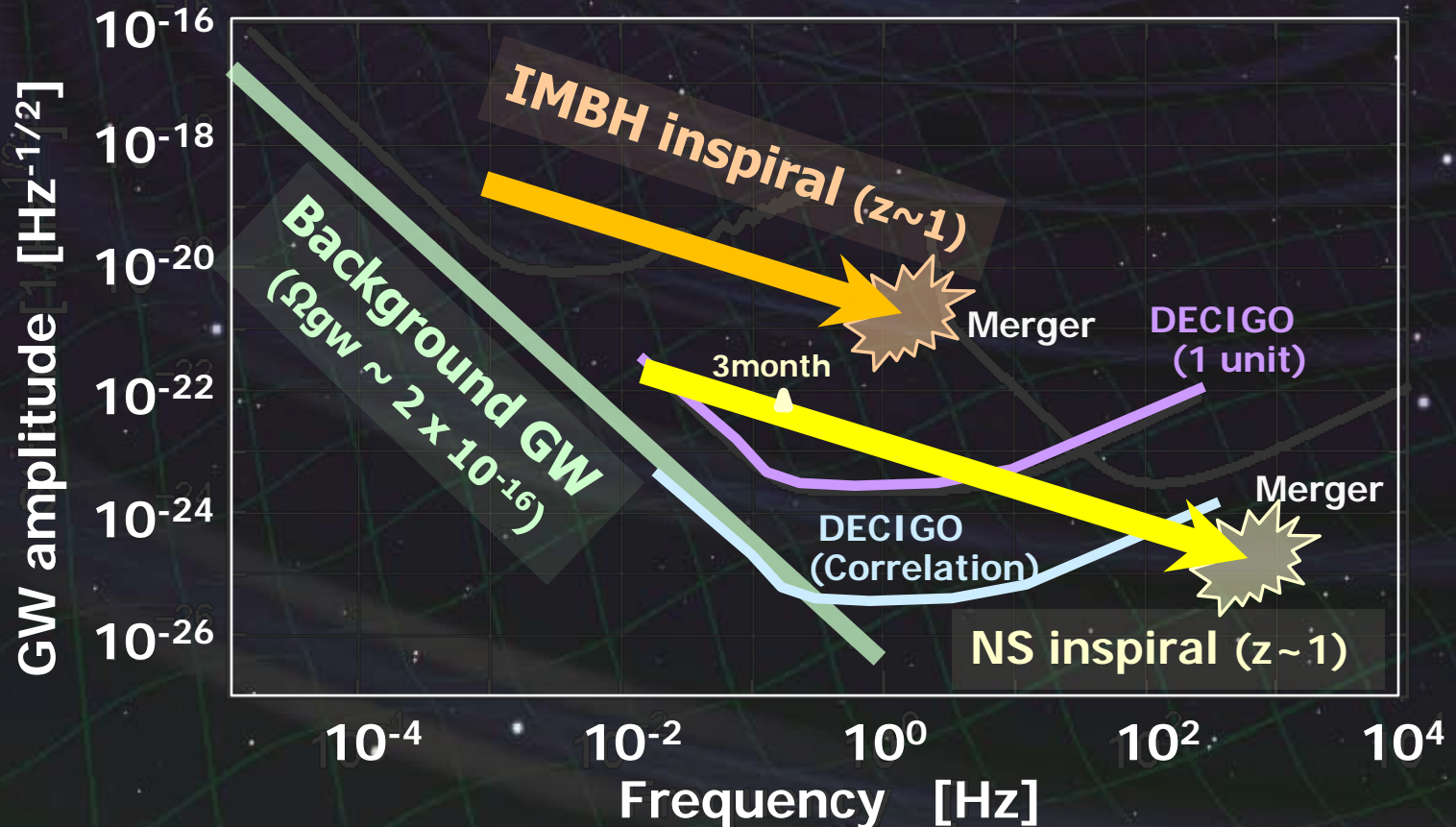


# Targets and Science

**IMBH** binary inspiral  
**NS** binary inspiral  
**Stochastic background**



Galaxy formation (Massive BH)  
Cosmology  
(Inflation, Dark energy)



# Constraint on dark energy

DECIGO will observe

$10^{4-5}$  NS binaries at  $z \sim 1$

↳ Precise 'clock' at cosmological distance

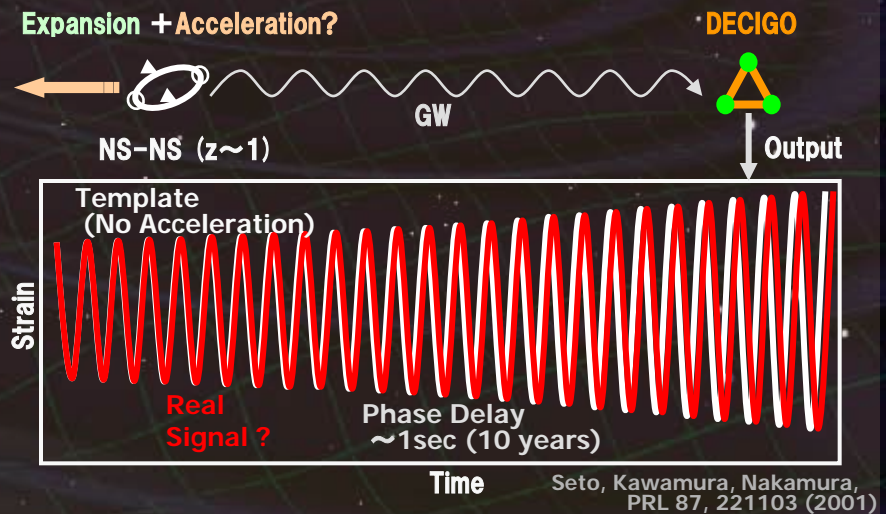
## 'Standard Siren'

Relationship between distance and redshift

Distance: chirp waveform

Redshift: host galaxy

→ Information on **acceleration of expansion of the universe**



Determine cosmological parameters

$$\Delta\Omega_m, \Delta\Omega_w, \Delta w \approx 1\%$$

**Absolute and independent measurement**

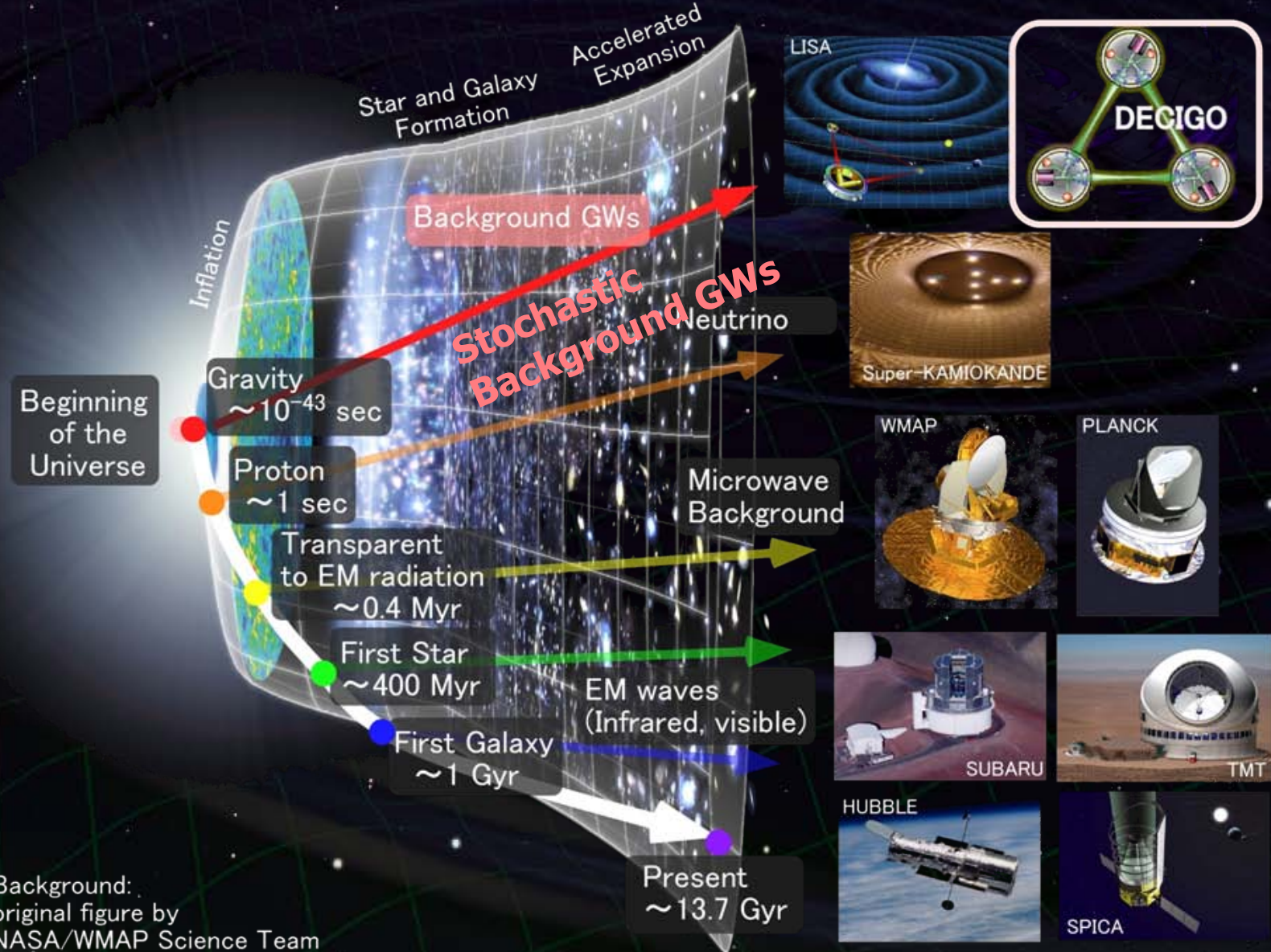
Angular resolution

$\sim 10\text{arcmin}$  (1 detector)  
 $\sim 10\text{arcsec}$  (3 detectors)

at  $z=1$



# Stochastic Background GWs





# 1. DECIGO

Overview and Science



Pre-conceptual Design

# 2. DECIGO Pathfinder

Overview and Science

Design and Status

# 3. Summary

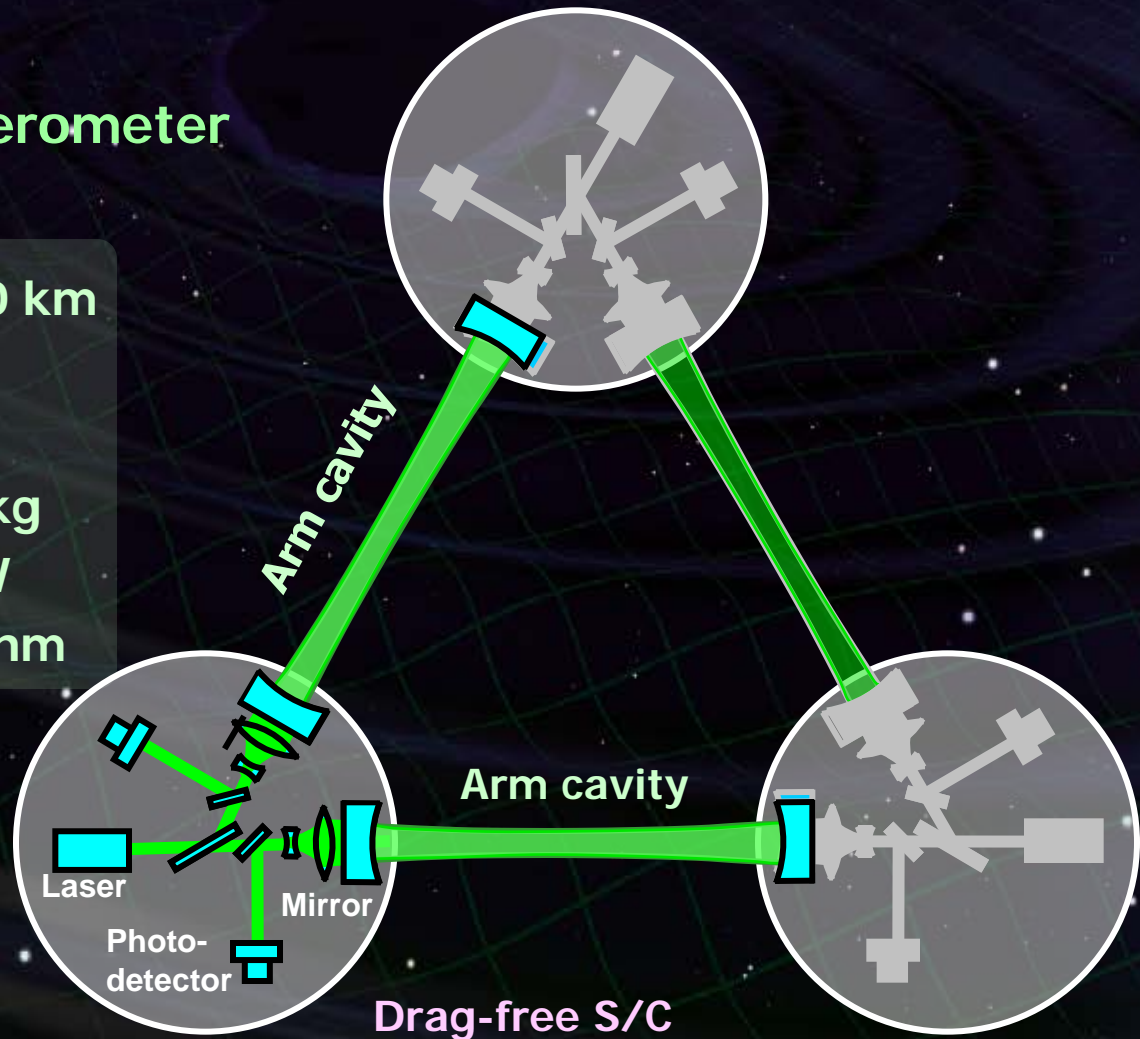
# Pre-Conceptual Design

## Interferometer Unit:

### Differential FP interferometer

Arm length:	1000 km
Finesse:	10
Mirror diameter:	1 m
Mirror mass:	100 kg
Laser power:	10 W
Laser wavelength:	532 nm

S/C: drag free  
3 interferometers



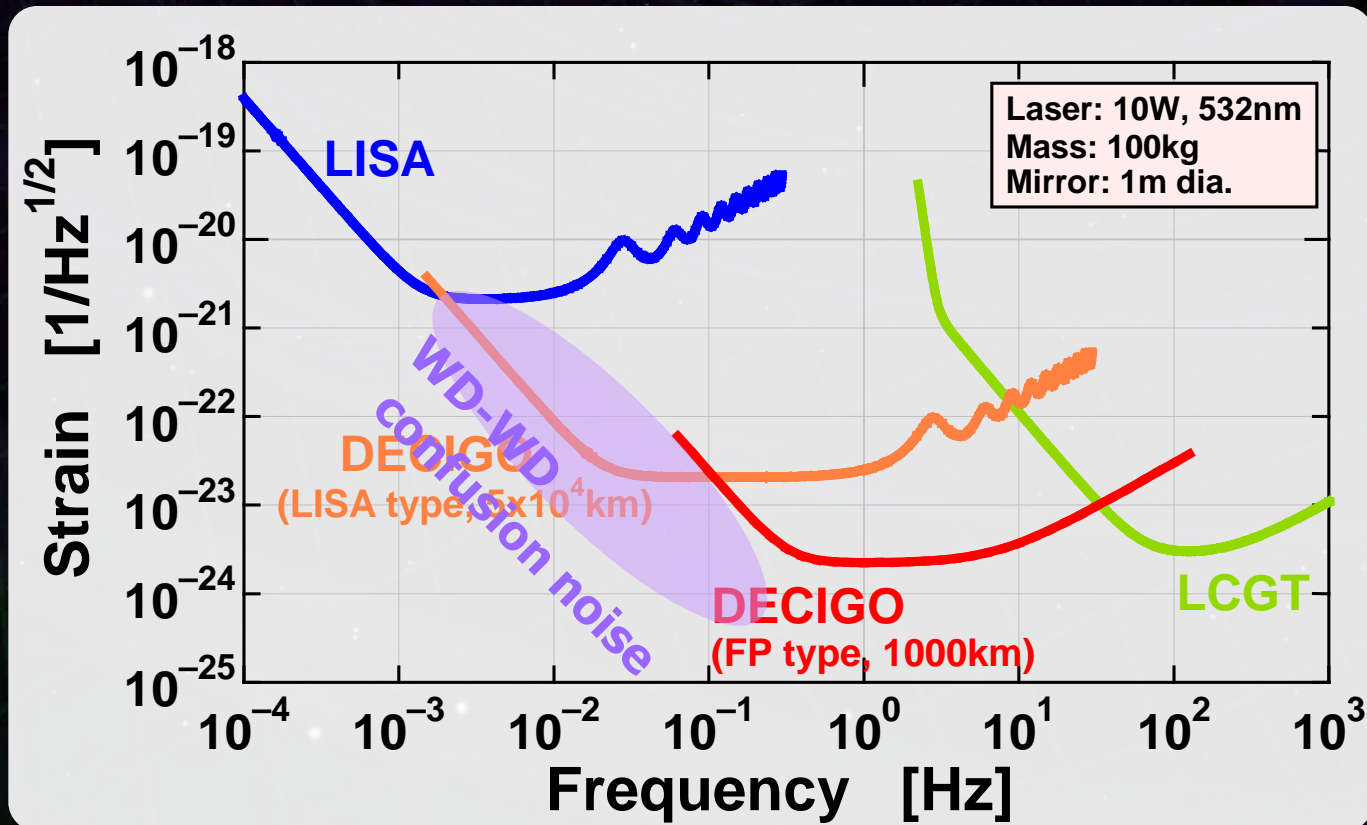


# Interferometer Design

## Transponder type vs Direct-reflection type

Compare : Sensitivity curves and Expected Sciences

⇨ Decisive factor: Binary confusion noise



# Cavity and S/C control

## Cavity length change

PDH error signal  $\rightarrow$  Mirror position (and Laser frequency)

Relative motion between mirror and S/C

Local sensor  $\rightarrow$  S/C thruster

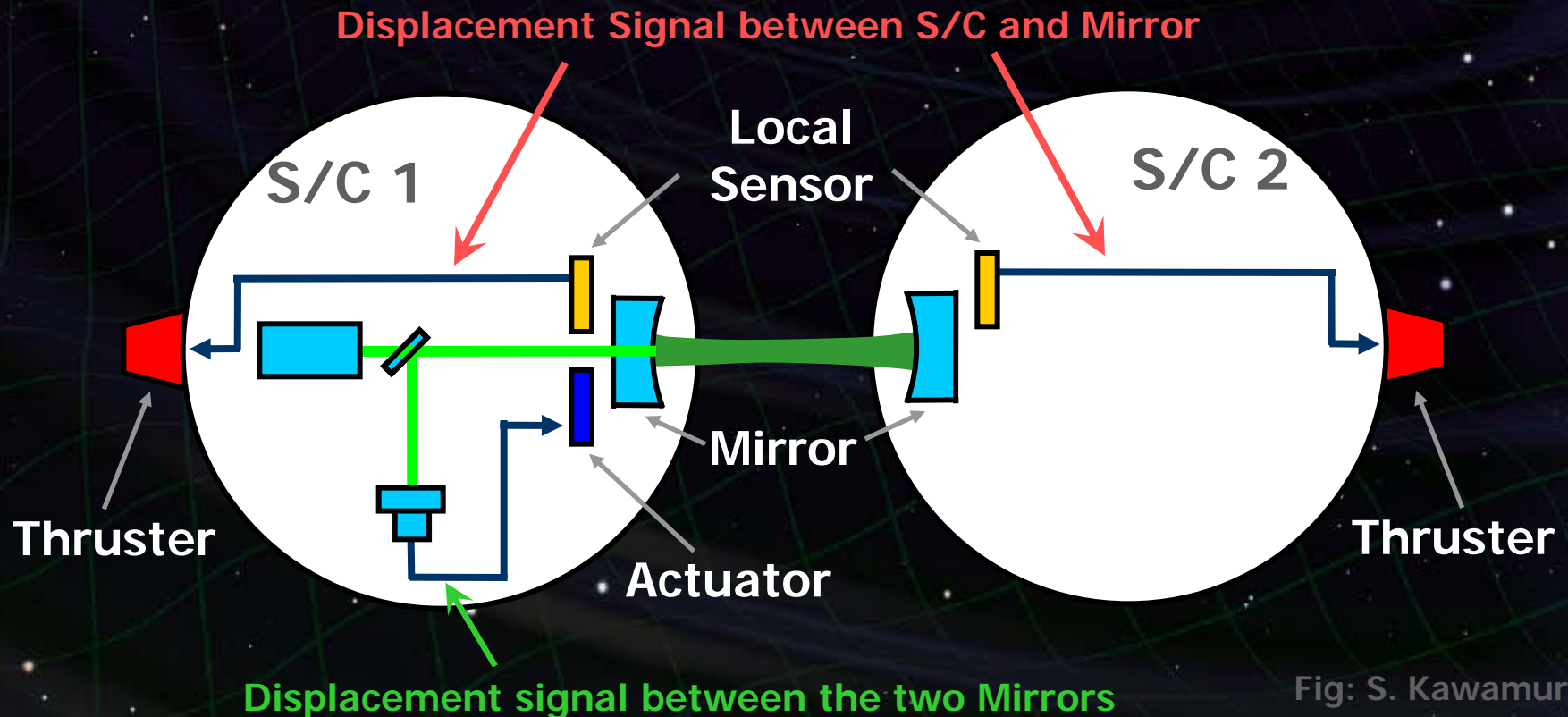


Fig: S. Kawamura



# Requirements

## Sensor Noise

Shot noise  $3 \times 10^{-18} \text{ m/Hz}^{1/2}$  (0.1 Hz)

⇒ x 10 of LCGT in phase noise

Other noises should be well below the shot noise

Laser freq. noise:  $1 \text{ Hz/Hz}^{1/2}$  (1Hz)

Stab. Gain  $10^5$ , CMRR  $10^5$

## Acceleration Noise

Force noise  $4 \times 10^{-17} \text{ N/Hz}^{1/2}$  (0.1 Hz)

⇒ x 1/50 of LISA

External force sources

Fluctuation of magnetic field, electric field,  
gravitational field, temperature, pressure, etc.

# Orbit and Constellation

Candidate of orbit:

Record-disk orbit around the Sun

Relative acc.  $4 \times 10^{-12} \text{ m/s}^2$   
(Mirror force  $\sim 10^{-9} \text{ N}$ )

Halo orbit around L2 (or L1)

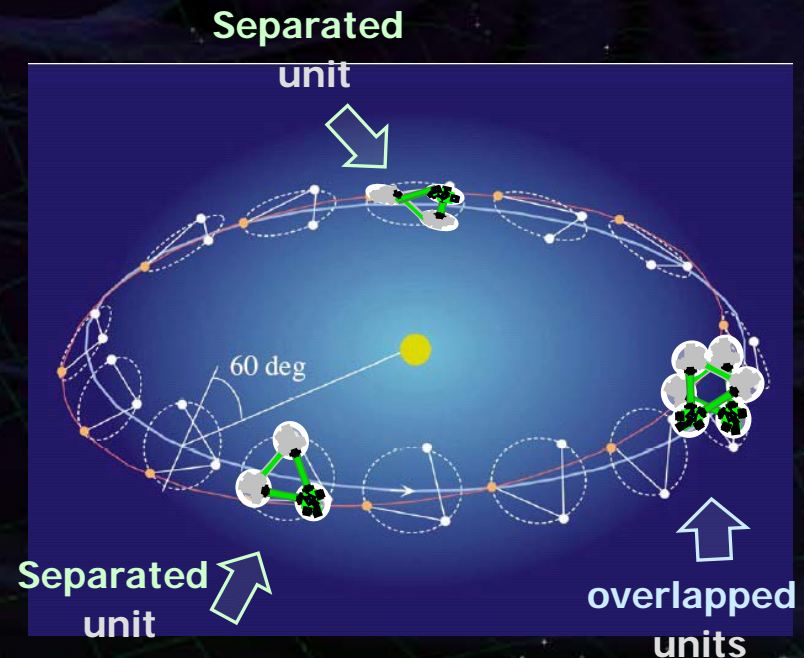
Relative acc.  $4 \times 10^{-7} \text{ m/s}^2$   
(Mirror force  $\sim 10^{-4} \text{ N}$ )

Constellation

4 interferometer units

2 overlapped units  $\rightarrow$  Cross correlation

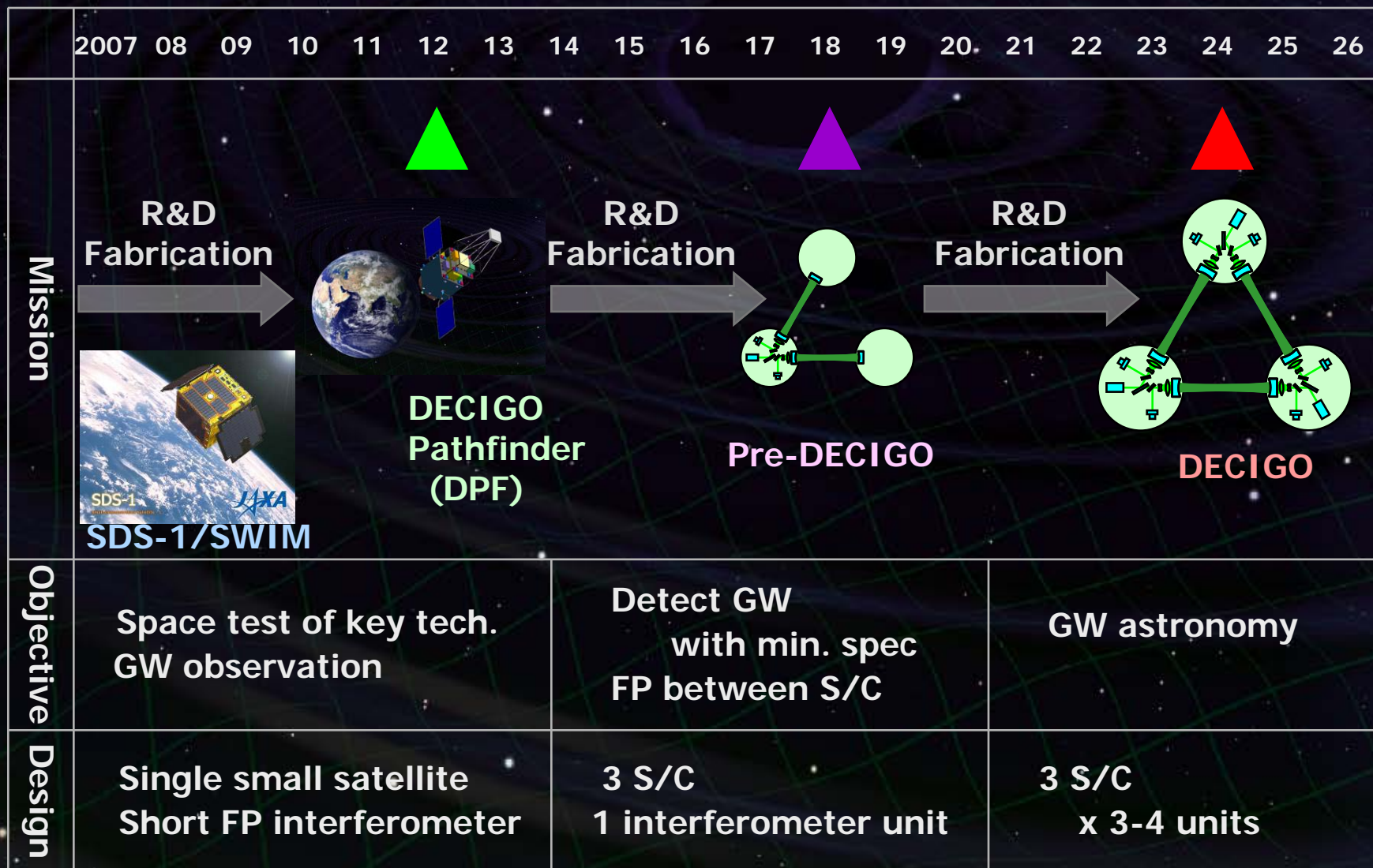
2 separated units  $\rightarrow$  Angular resolution





# Roadmap

Figure: S.Kawamura



# Organization

**PI: Kawamura (NAOJ)**  
**Deputy: Ando (Kyoto)**

**Executive Committee**  
Kawamura (NAOJ), Ando (Kyoto), Seto (Kyoto), Nakamura (Kyoto),  
Tsubono (Tokyo), Tanaka (Kyoto), Funaki (ISAS), Numata (Maryland),  
Sato (Hosei), Kanda (Osaka city), Takashima (ISAS), Ioka (KEK)

**Pre-DECIGO**  
Sato (Hosei)

**Detector**  
Numata  
(Maryland)  
Ando (Kyoto)

**Science, Data**  
Tanaka (Kyoto)  
Seto (Kyoto)  
Kanda (Osaka city)

**Satellite**  
Funaki (ISAS)

**Design phase**

**DECIGO pathfinder**  
**Leader: Ando (Kyoto)**  
**Deputy: Takashima (ISAS)**

**Mission phase**

**Detector**  
Ando  
(Kyoto)

**Laser**  
Ueda  
(ILS)  
Musya  
(ILS)

**Housing**  
Sato  
(Hosei)

**Drag free**  
Moriwaki  
(Tokyo)  
Sakai  
(ISAS)

**Thruster**  
Funaki  
(ISAS)

**Bus**  
Takashim  
a (ISAS)

**Data**  
Kanda  
(Osaka  
city)



# Collaboration and support

- Supports from **LISA**  
Technical advises from LISA/LPF experiences  
Support Letter for DECIGO/DPF  
LISA-DECIGO workshop (2008.11)
- Collab. with **Stanford univ. group**  
Drag-free control of DECIGO/DPF  
UV LED Charge Management System for DPF
- Collab. with **JAXA navigation-control section**  
→ formation flight of DECIGO, DPF drag-free control
- Research Center for the Early Universe (**RESCEU**), Univ. of Tokyo  
Support DECIGO as ones of main projects (2009.4-)
- Collab. with **UNISEC** (University Space Engineering Consortium)  
Call for active young engineers

# 1. DECIGO

Overview and Science

Pre-conceptual Design



# 2. DECIGO Pathfinder

Overview and Science

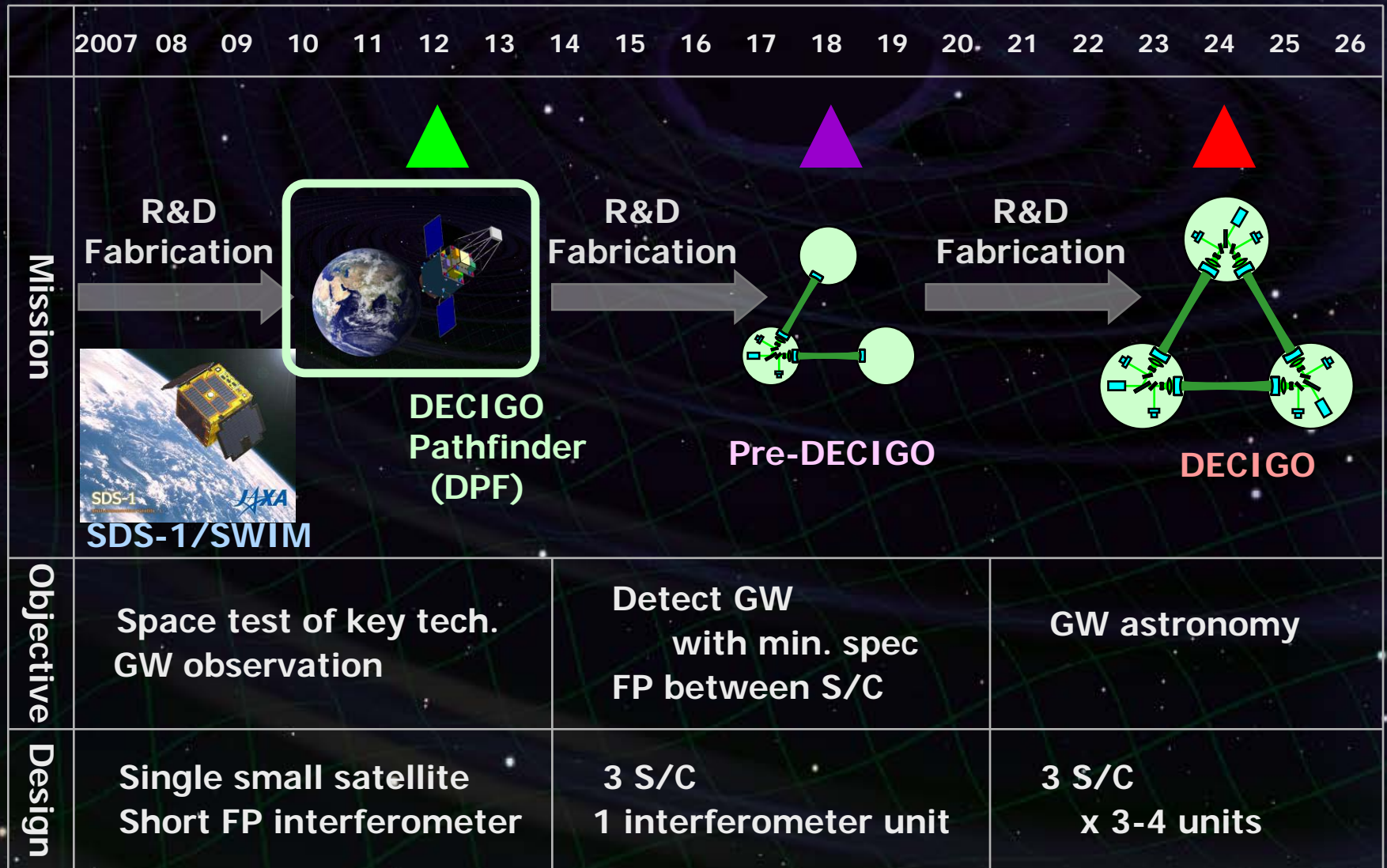
Design and Status

# 3. Summary



# Roadmap

Figure: S.Kawamura



# DECIGO-PF

## DECIGO Pathfinder (DPF)

First milestone mission for DECIGO

Shrink arm cavity

DECIGO 1000km  $\rightarrow$  DPF 30cm

### Single satellite

(Payload  $\sim 1\text{m}^3$ , 350kg)

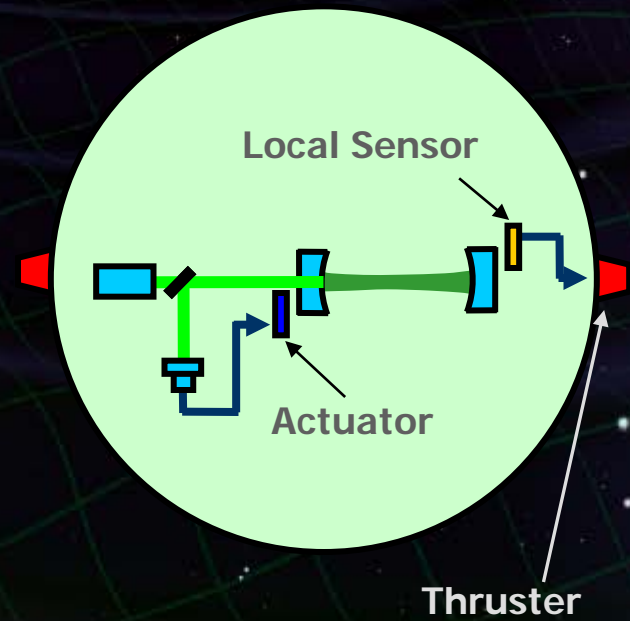
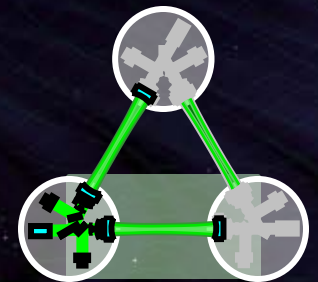
### Low-earth orbit

(Altitude 500km, sun synchronous)

30cm FP cavity with 2 test masses

Stabilized laser source

Drag-free control





# DPF satellite

## DPF Payload

Size : 950mm cube  
Weight : 150kg  
Power : 130W  
Data Rate: 800kbps  
Mission thruster x12

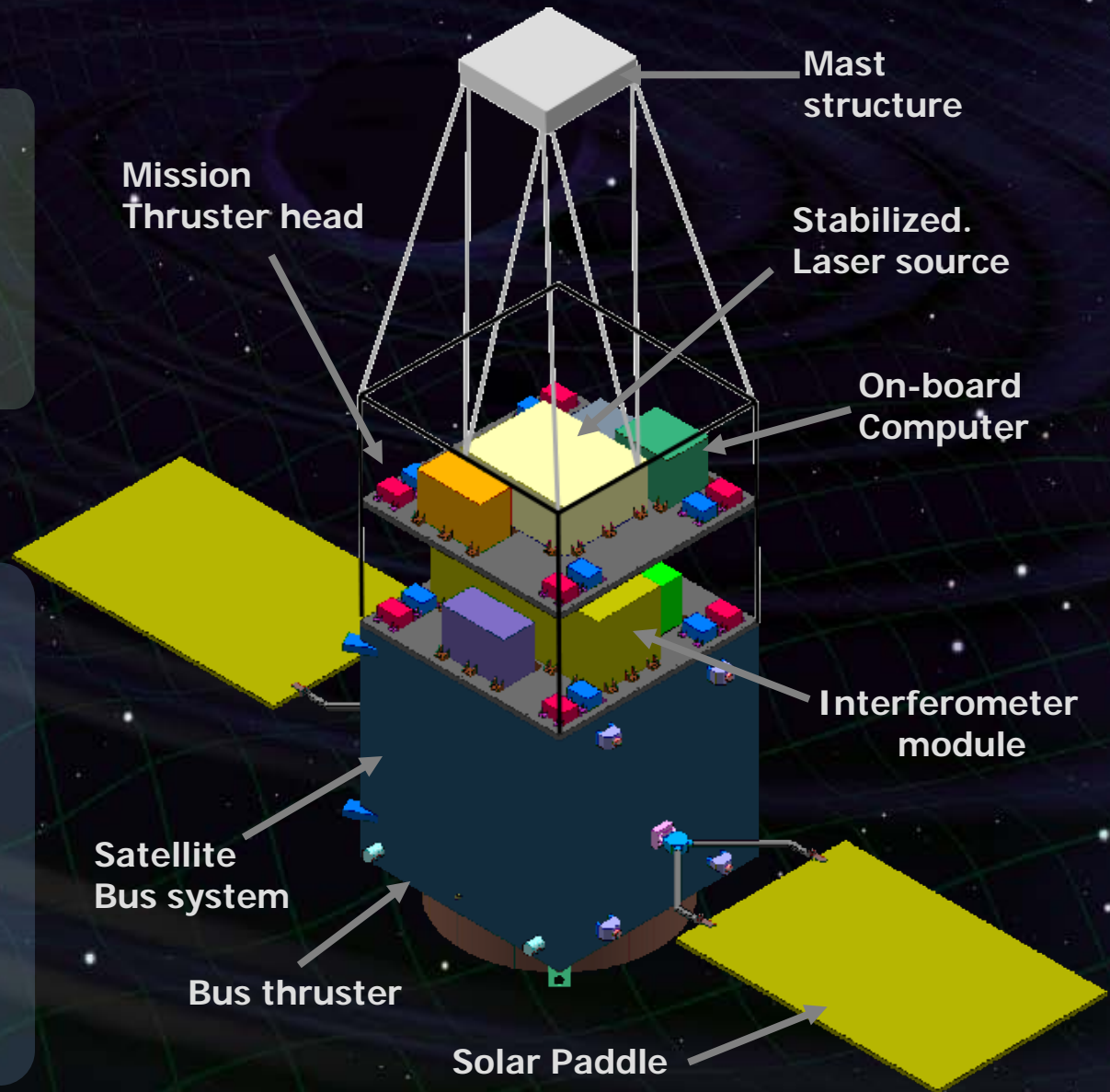
Power Supply  
SpW Comm.



## Satellite Bus

('Standard bus' system)

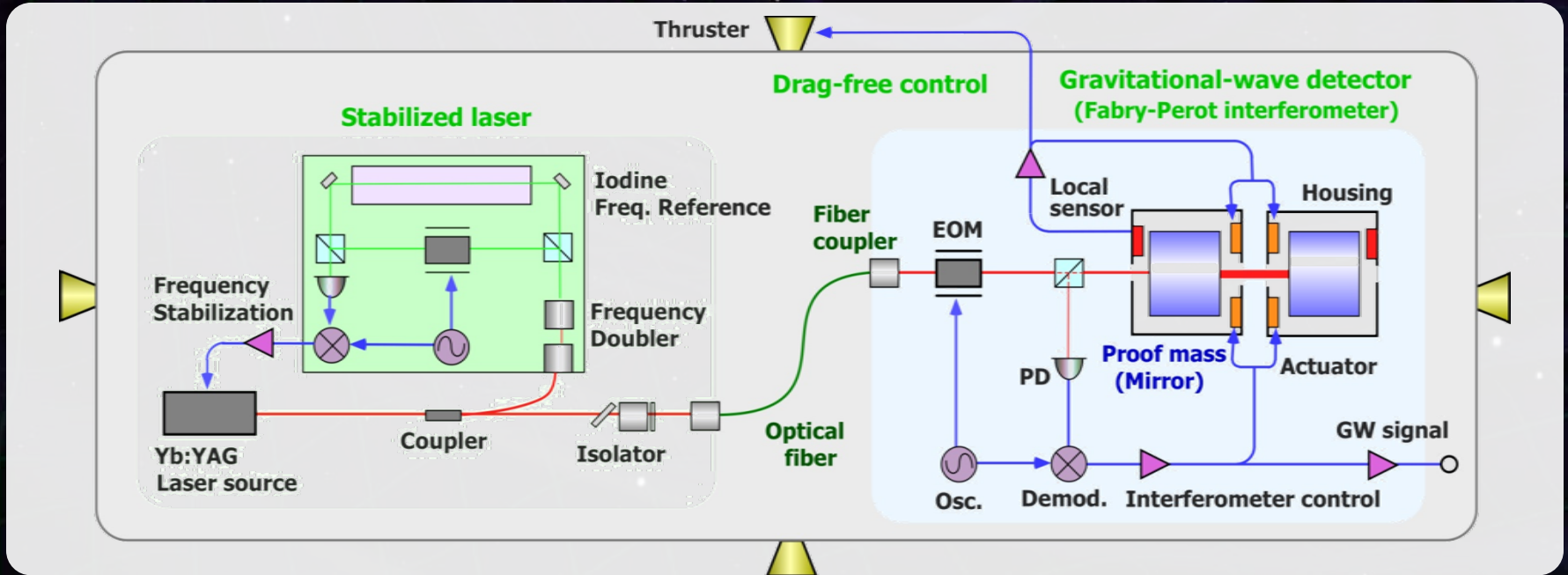
Size :  
950x950x1100mm  
Weight : 200kg  
SAP : 960W  
Battery: 50AH  
Downlink : 2Mbps  
DR: 1GByte  
3N Thrusters x 4



# DPF mission payload

Mission weight : ~150kg  
Mission space : ~90 x 90 x 90 cm

Drag-free control  
Local sensor signal  
→ Feedback to thrusters



## Laser source

Yb:YAG laser (1030nm)  
Power : 25mW  
Freq. stab. by Iodine abs. line

## Fabry-Perot interferometer

Finesse : 100  
Length : 30cm  
Test mass : 1kg  
Signal extraction by PDH

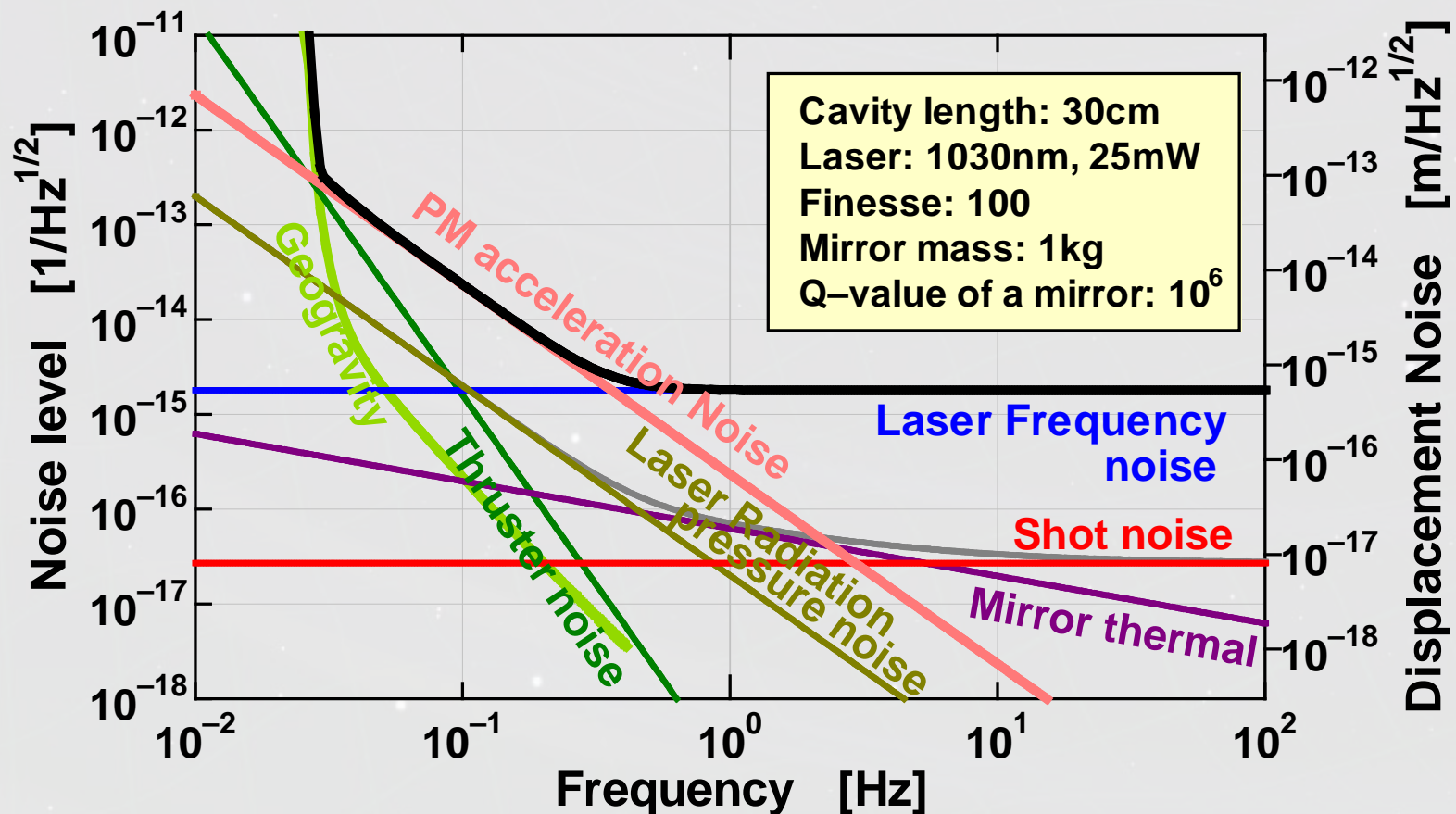


# DPF Sensitivity

Laser source : 1030nm, 25mW  
IFO length : 30cm  
Finesse : 100, Mirror mass : 1kg  
Q-factor :  $10^5$ , Substrate: TBD  
Temperature : 293K

Satellite mass : 350kg, Area: 2m<sup>2</sup>  
Altitude: 500km  
Thruster noise:  $0.1\mu\text{N}/\text{Hz}^{1/2}$

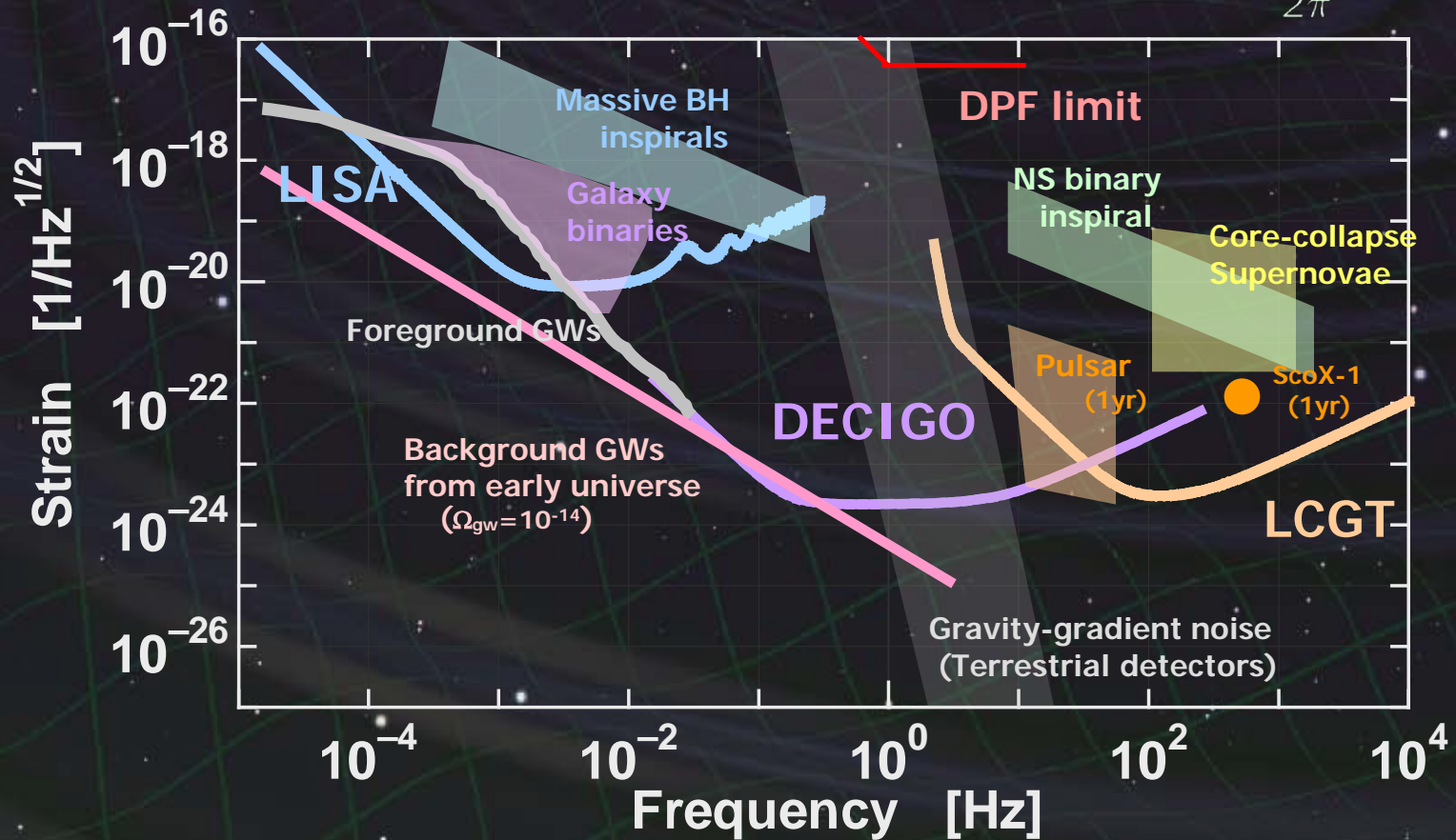
(Preliminary parameters)



# DPF sensitivity

DPF sensitivity  $h \sim 2 \times 10^{-15} \text{ Hz}^{1/2}$   
(x10 of quantum noises)

$$f \sim \frac{1}{2\pi} \sqrt{GM/R^3}$$





# GW target of DPF

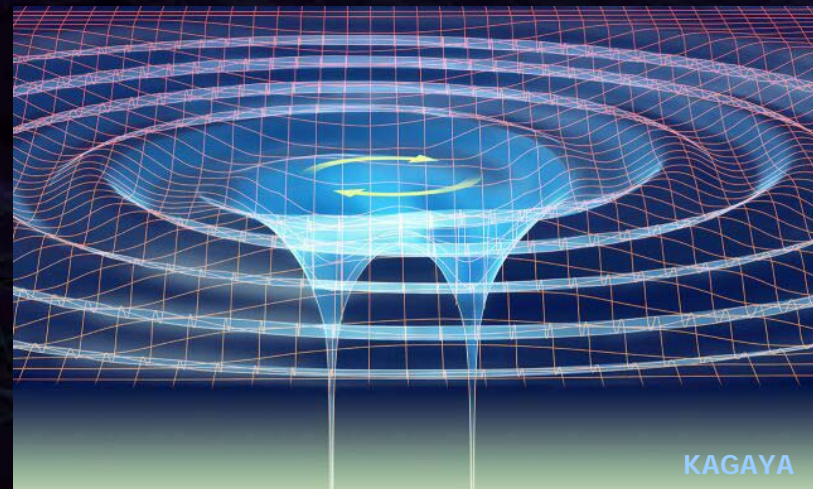
## Blackholes events in our galaxy

### IMBH inspiral and merger

$$h \sim 10^{-15}, f \sim 4 \text{ Hz}$$

$$\text{Distance } 10 \text{ kpc}, m = 10^3 M_{\text{sun}}$$

Obs. Duration ( $\sim 1000 \text{ sec}$ )



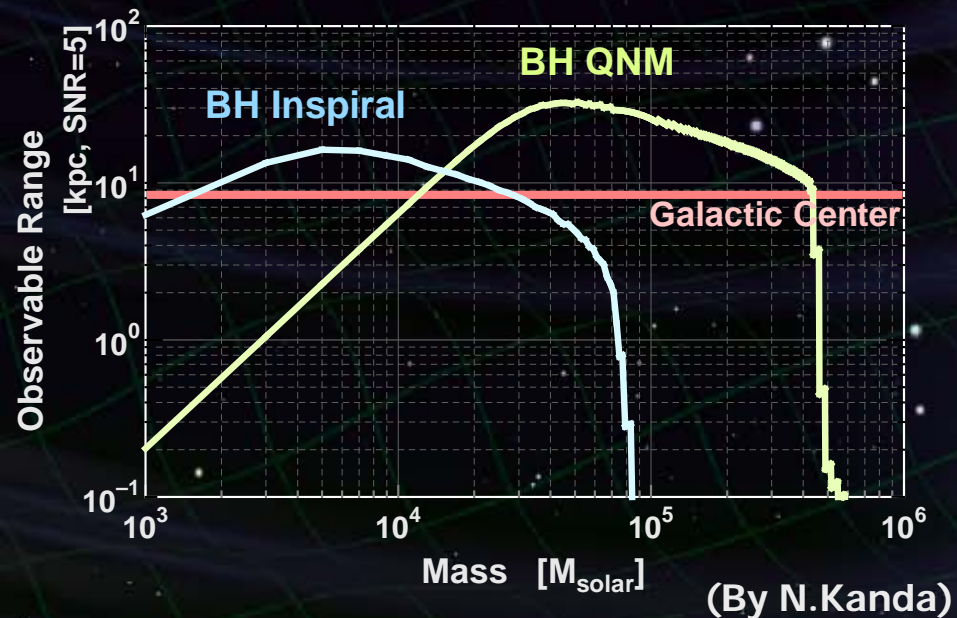
### BH QNM

$$h \sim 10^{-15}, f \sim 0.3 \text{ Hz}$$

$$\text{Distance } 1 \text{ Mpc}, m = 10^5 M_{\text{sun}}$$

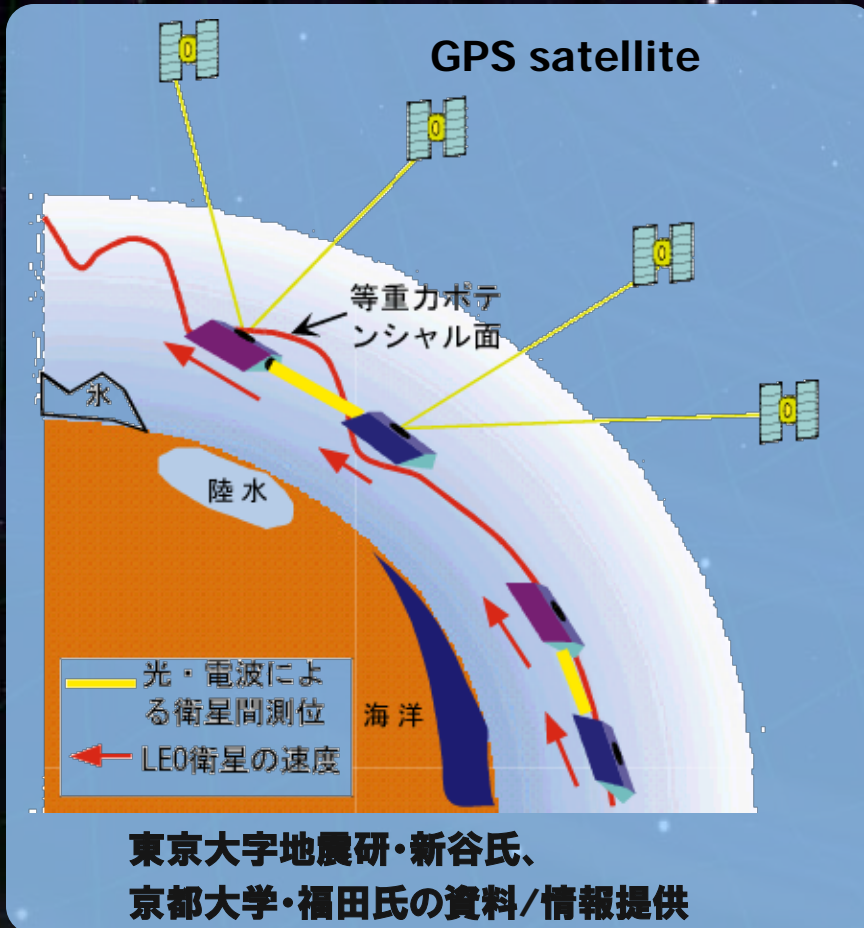
Observable range reaches  
the **Galactic center** (SNR  $\sim 5$ )

Hard to access by others  
→ Original observation



# Gravity of the Earth

## Measure gravity field of the Earth for Satellite Orbits



Determine global gravity field  
→ Density distribution  
Monitor of change in time  
Ground water motion  
Strains in crusts by  
earthquakes and volcanoes

Observation Gap  
between GRACE and GRACE-FO  
(2012-16)  
→ DPF contribution  
in international network



# 1. DECIGO

Overview and Science

Pre-conceptual Design

# 2. DECIGO Pathfinder

Overview and Science

**Design and Status**

# 3. Summary



# R&D for DPF (1)

## Stabilized Laser

BBM development

Yb:YAG (NPRO) source

Saturated absorption by  $I_2$

→ Stability test, Packaging

By  
M.Musha



## IFO and housing

BBM-EM development

→ Test of concepts

+ Earth gravity sensors

S.Sato's talk

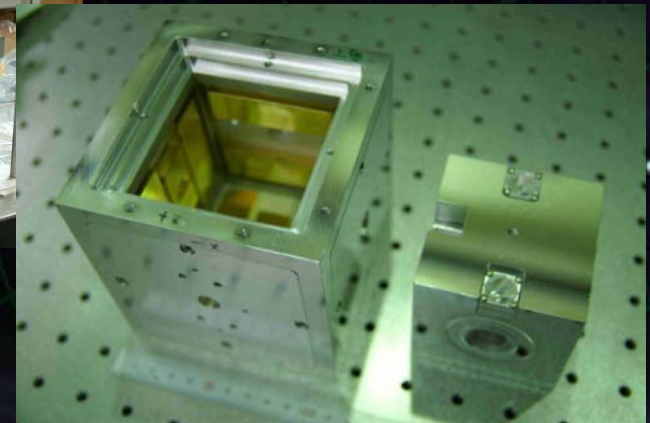
(P. Session #2, Today)

Y.Wakabayashi's poster



By  
S.Sato

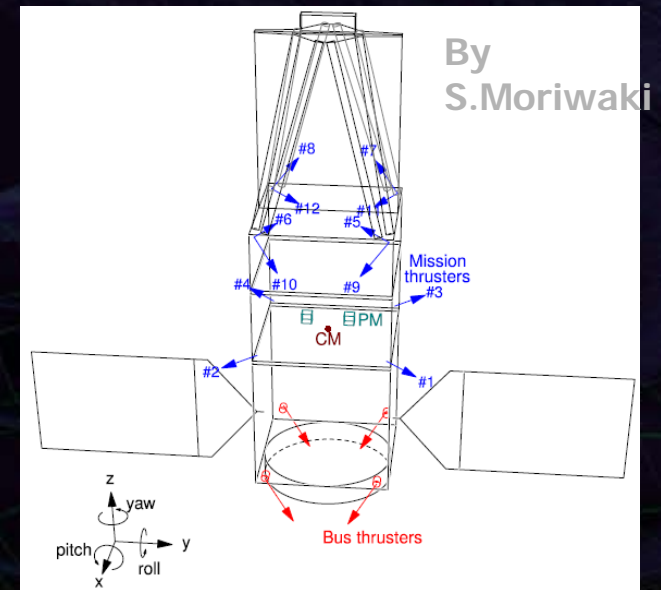
By  
A.Araya





# R&D for DPF (2)

Attitude control and Drag-free  
Satellite structure (mass distribution)  
Passive attitude stabilization  
by gravity gradient  
Mission thruster position  
Control topology



Thruster  
System design  
with existing tech.  
Noise meas. system  
(thruster stand)  
Development of Slit FEED

By  
I.Funaki



# SWIM launch and operation

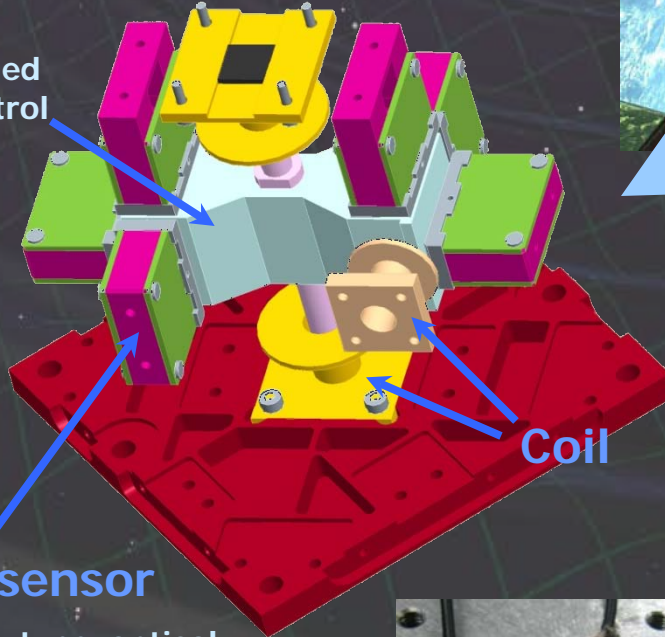
Tiny GW detector module  
Launched in Jan. 23, 2009

⇒ In-orbit operation

TAM: Torsion Antenna Module with free-falling test mass  
(Size : 80mm cube, Weight : ~500g)

## Test mass

~47g Aluminum, Surface polished  
Small magnets for position control



Coil

## Photo sensor

Reflective-type optical displacement sensor  
Separation to mass ~1mm  
Sensitivity ~  $10^{-9}$  m/Hz<sup>1/2</sup>  
6 PSs to monitor mass motion

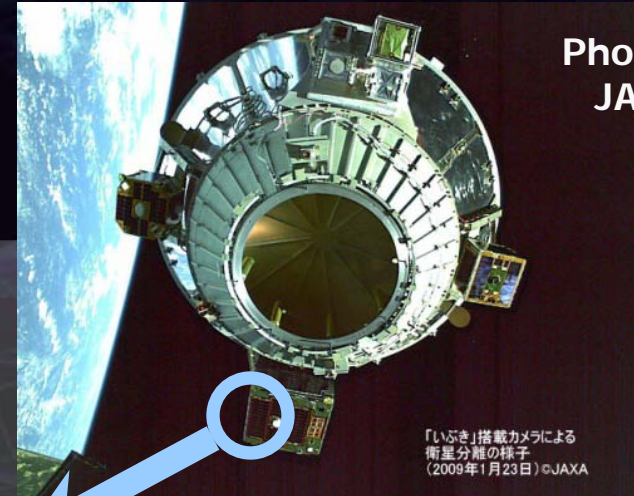
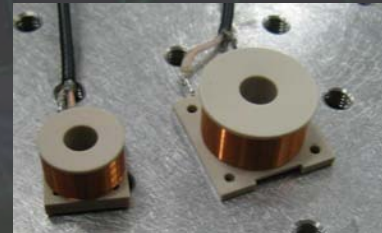


Photo:  
JAXA

「いぶき」搭載カメラによる  
衛星分離の様子  
(2009年1月23日) ©JAXA

W.Kokuyama's talk  
(P. Session #2,  
Today)



# DPF mission status

DPF : One of the candidate of  
JAXA's small satellite series



At least 3 satellite in 5 years with  
Standard Bus + M-V follow-on rocket

1<sup>st</sup> mission (2012): SPRINT-A/EXCEED

2<sup>nd</sup> mission (~2013) in selection

Candidates: 2 missions (ERG, DPF)

**Decision in this month**



Next-generation  
Solid rocket booster (M-V FO)  
Fig. by JAXA

# 1. DECIGO

Overview and Science

Pre-conceptual Design

# 2. DECIGO Pathfinder

Overview and Science

Design and Status



# 3. Summary



# Summary

---

## **DECIGO : Fruitful Sciences**

**Very beginning of the Universe**

**Dark energy**

**Galaxy formation**

## **DECIGO Pathfinder**

**Important milestone for DECIGO**

**Strong candidate of JAXA's satellite series**

**SWIM – under operation in orbit**

**first precursor to space!**

The image features a dark blue background with a grid of green lines that are distorted into a wavy pattern, representing gravitational waves. In the center, the word "End" is written in a bold, white, sans-serif font. The background is also filled with numerous small, white, star-like points of light. At the top and bottom of the image, there are horizontal bars with a repeating pattern of small, light blue and white squares.

**End**



# LCGT and DECIGO

LCGT (~2014)

Terrestrial Detector

→ High frequency events

Target: GW detection

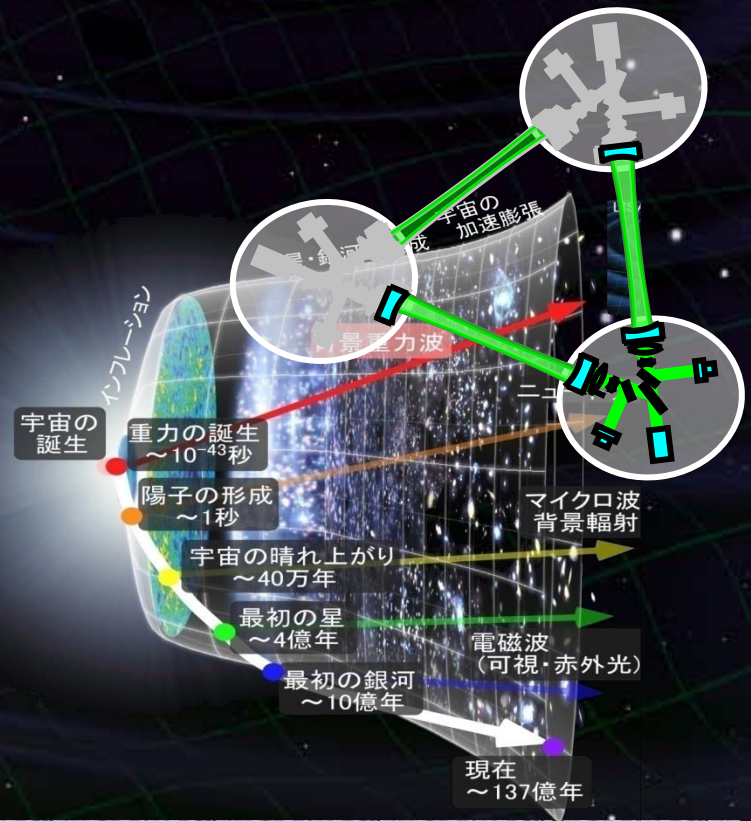


DECIGO (~2024)

Space observatory

→ Low frequency sources

Target: GW astronomy

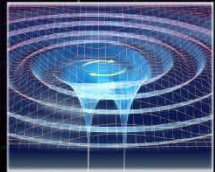




# Objectives of DPF

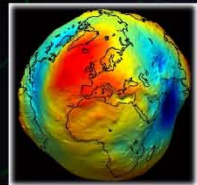
## Observation

### Gravitational wave

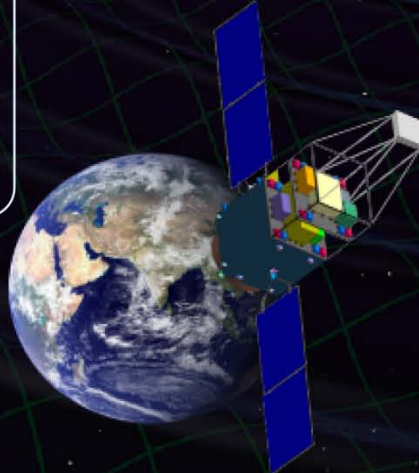


Intermediate-mass  
inspiral and merger

### Earth gravity



Environ. monitor  
Geoid resolution  
 $\sim 1\text{mm}$ .



## Science Technology

### Space interferometer

Precise meas. in space

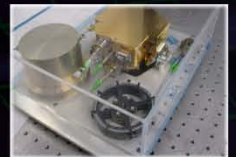
$$6 \times 10^{-16} \text{ m/Hz}^{1/2}$$



### Stabilized laser

High stability in Space

$$0.5 \text{ Hz/Hz}^{1/2}$$



### Drag-free control

Low-noise control  
with passive stab.





# DPF and DECIGO

## DPF requirements

Precise meas.  
by IFO



Disp. noise  
 $6 \times 10^{-16} \text{ m/Hz}^{1/2}$

$4 \times 10^{-18} \text{ m/Hz}^{1/2}$

Force noise  
 $10^{-14} \text{ N/Hz}^{1/2}$

$10^{-17} \text{ N/Hz}^{1/2}$

Stab. Laser



Freq. Stability  
 $0.5 \text{ Hz/Hz}^{1/2}$

$1 \text{ Hz/Hz}^{1/2}$

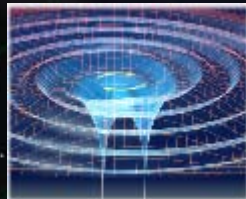
Drag-free  
control



Satellite disp.  
 $10^{-9} \text{ m/Hz}^{1/2}$

Thruster noise  
 $10^{-7} \text{ N/Hz}^{1/2}$

GW Obs.



0.1 Hz band  
Observation and  
Data analysis

## DECIGO requirements

1000km FP cavity  
IFO control in space  
Low external force  
Large optics

Ultra stable Laser  
Stabilization of source  
Stabilization by long arm

Formation flight  
Stable orbit  
Inter S/C Ranging  
Drag-free control  
Low-noise thruster

Observation  
Data procession  
Data analysis  
Triggered search

# Arm length

Cavity arm length : Limited by diffraction loss

Effective reflectivity ( $TEM_{00} \rightarrow TEM_{00}$ )

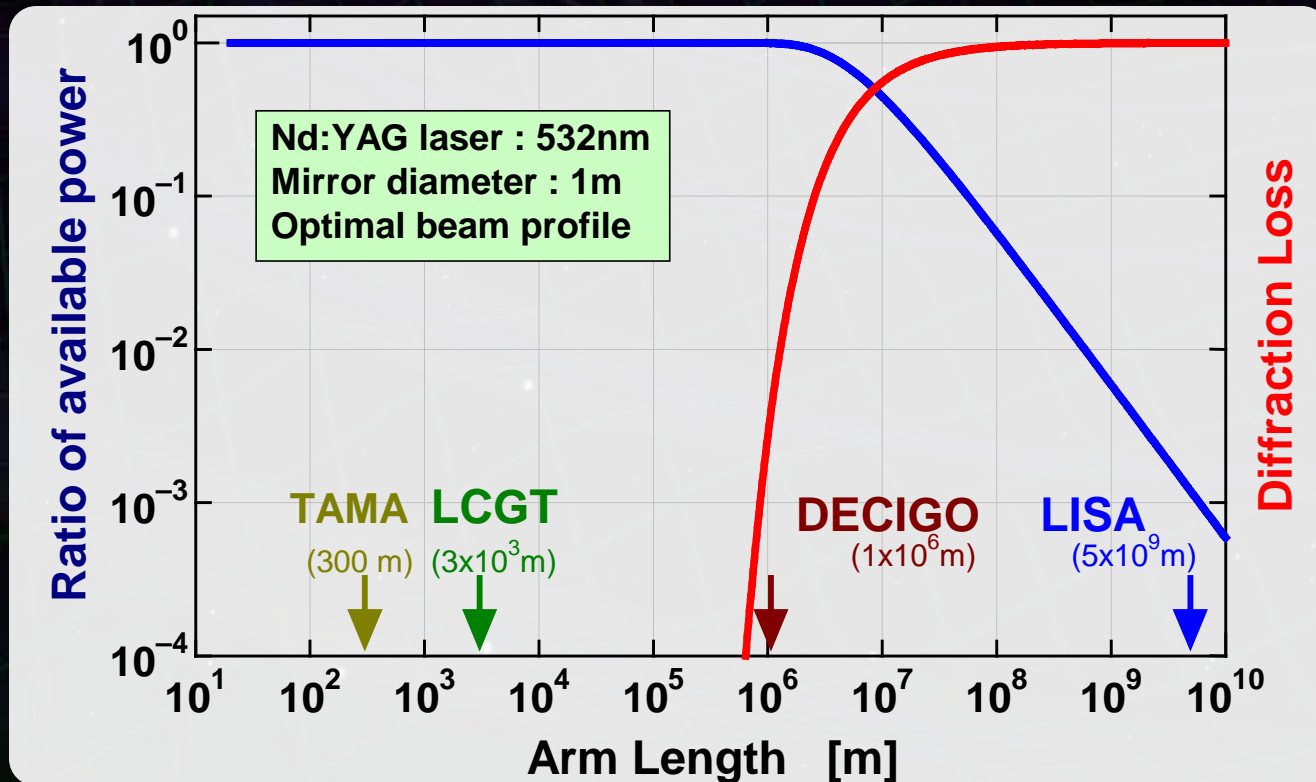
Laser wavelength : 532nm

Mirror diameter: 1m

Optimal beam size



1000 km  
is almost max.



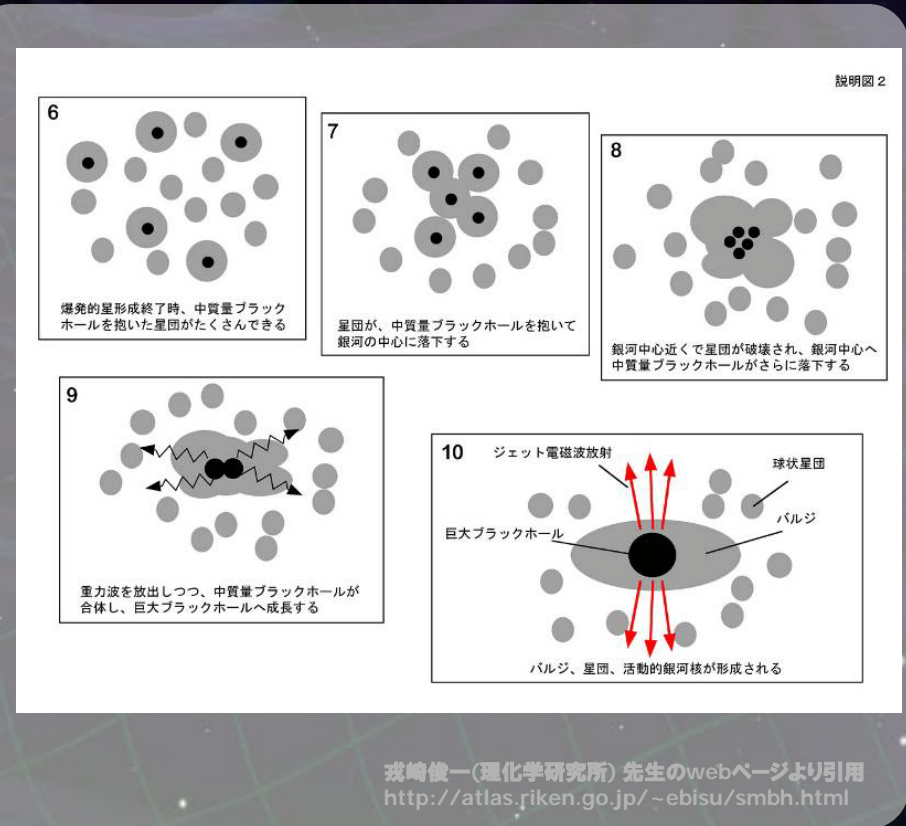


# IMBH inspiral and Merger

DECIGO will observe  
Intermediate-mass BH (IMBH)  
binary merger with  
SNR > 6000 for  $z \sim 1$  source

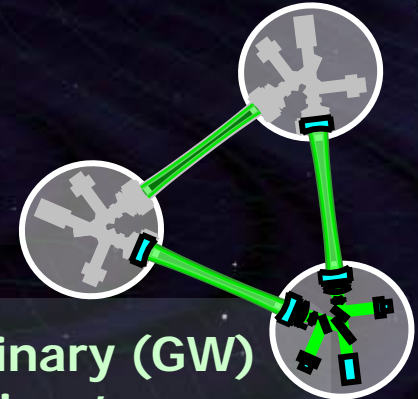


Information on the  
formation of  
Supermassive BHs  
at the center of galaxies



# Standard Sources

Fig. from  
SNAP  
web page



**Supernova (EM wave)**  
**'Standard Candle'**

**Neutron-star binary (GW)**  
**'Standard Siren'**

**Absolute power  
or amplitude**

Extrapolated from  
nearby events

<

General Relativity

**Event rate**

2000/yr (SNAP)

<

$10^{4-5}$ /yr (DECIGO)

**Error in distance**

~10%

≈

10% at  $z=1$

**Identification  
of host galaxy**

Easy?

>

Require multiple detectors  
or statistics

**Others**

Uncertainty by  
dust absorption

<

Negligible interaction  
with matters

R.Takahashi (2006)



# DPF targets

## BHs in Globular clusters

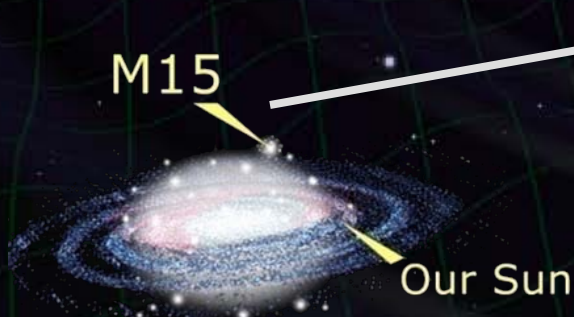
BH masses estimated from star motion

⇒ Estimate SNR of GW signals

Equal mass, Mass ratio 1:1/3, 100Msun BH capture

Credit: NASA, STScI

Globular clusters known to have black holes



Milky Way Galaxy  
(artist's concept)

(~ 150 Globular Clusters  
in our Galaxy)

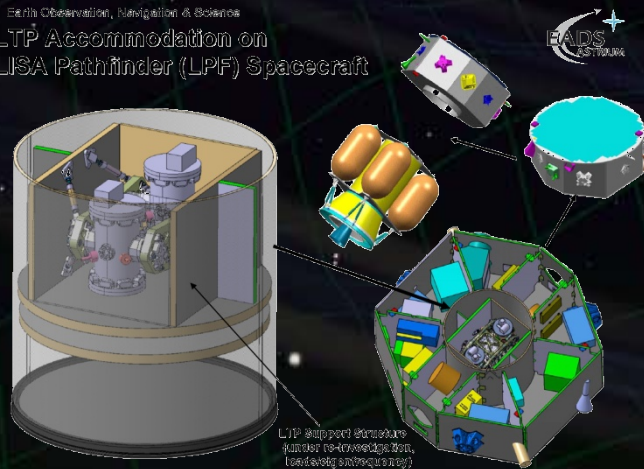
NGC#	BH質量 [Msun]	距離 [kpc]	SNR (同質量)	SNR (1:1/3)	SNR +100Msun	速度分散 [km/sec]
6441	12423.8	11.2	36.4	22.2	3.7	19.5
6256	4753.6	6.9	26.6	16.2	4.3	15.4
7078	4387.8	10.3	16.6	10.2	2.8	15.1
6093	3720.3	10.0	14.9	9.1	2.7	14.5
104	820.0	4.5	9.4	5.7	3.6	10
1851	1348.5	12.1	5.3	3.2	1.6	11.3
6681	820.0	9.0	4.7	2.9	1.8	10
6293	365.6	8.8	2.5	1.5	1.4	8.2
5286	443.8	11.0	2.3	1.4	1.2	8.6
6522	227.8	7.8	1.9	1.1	1.3	7.3
5904	142.0	7.5	1.3	0.8	1.1	6.5
6325	133.3	8.0	1.2	0.7	1.0	6.4
6752	45.0	4.0	0.9	0.6	1.3	4.9
7099	89.3	8.0	0.8	0.5	0.9	5.8
6284	170.7	15.3	0.7	0.5	0.6	6.8

(By N.Seto)

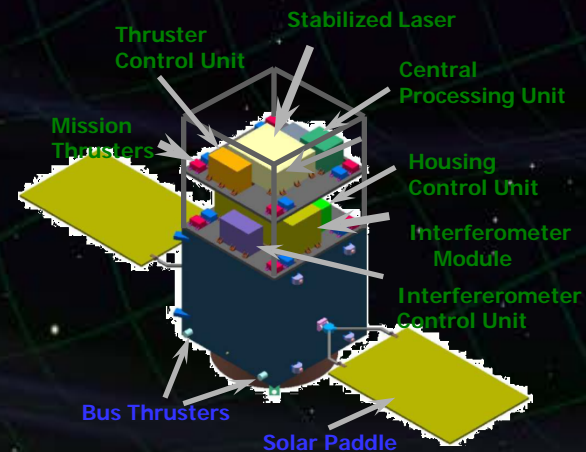
# Comparison with LPF

	LPF (LISA Pathfinder)	DPF (DECIGO Pathfinder)
Purpose	Demonstration for LISA	Demonstration for DECIGO GW observation
Launch	2010	~2013
Weight	1,900 kg	350 kg
Orbit	Halo orbit around L1 Drag-free attitude control	SSO altitude 500km Drag-free attitude control
Test Mass	Au-Pt alloy x2	TBD x2
Laser source	Nd:YAG (1064nm)	Yb:YAG (1030nm)
Interferometer	Mach-Zehnder	Fabry-Perot
Sensitivity	$3 \times 10^{-14} \text{ m/s}^2/\text{Hz}^{1/2}$ (1mHz)	$1 \times 10^{-15} \text{ m/s}^2/\text{Hz}^{1/2}$ (0.1Hz)

Earth Observation, Navigation & Science  
LTP Accommodation on  
LISA Pathfinder (LPF) Spacecraft

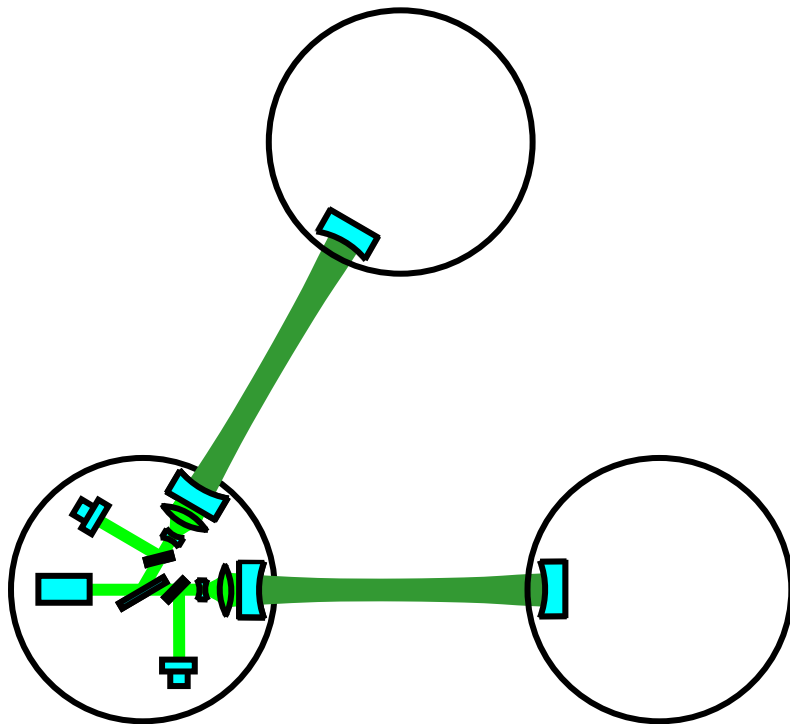


LTP instrument enclosure  
includes re-configuration,  
hardware and software





# Pre-DECIGO



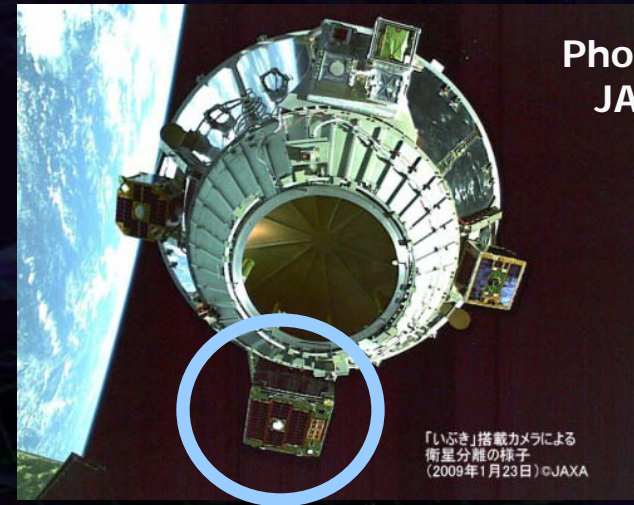
	<b>Pre-DECIGO</b>	<b>DECIGO</b>
<b>Arm length</b>	<b>100 km</b>	<b>1000 km</b>
<b>Mirror diameter</b>	<b>30 cm</b>	<b>1 m</b>
<b>Laser wavelength</b>	<b>0.532 <math>\mu\text{m}</math></b>	<b>0.532 <math>\mu\text{m}</math></b>
<b>Finesse</b>	<b>30</b>	<b>10</b>
<b>Laser power</b>	<b>1 W</b>	<b>10 W</b>
<b>Mirror mass</b>	<b>30 kg</b>	<b>100 kg</b>
<b># of interferometers in each cluster</b>	<b>1</b>	<b>3</b>
<b># of clusters</b>	<b>1</b>	<b>4</b>

•8th Edoardo Amaldi Conference on Gravitational Waves (June 24, 2009, New York, USA)

# SWIM launch

Test of signal processing  
and control system

SWIM (Space-wire Demonstration module)  
on SDS-1 satellite  
Launched in Jan. 23, 2009



## SpaceCube2: Space-qualified Computer

CPU: HR5000  
(64bit, 33MHz)

System Memory:  
2MB Flash Memory  
4MB Burst SRAM  
4MB Asynch. SRAM  
Data Recorder:  
1GB SDRAM  
1GB Flash Memory  
SpW: 3ch

Size: 71 x 221 x 171  
Weight: 1.9 kg  
Power: 7W

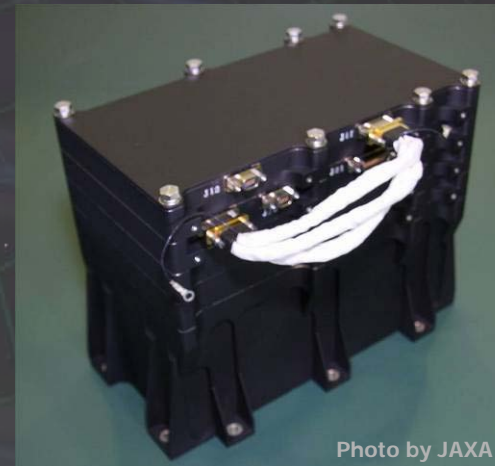


## SWIM $\mu$ v : User Module

Processor test board  
GW+Acc. sensor

FPGA board  
DAC 16bit x 8 ch  
ADC 16bit x 4 ch  
→ 32 ch by MPX  
Torsion Antenna x2  
~47g test mass

Data Rate : 380kbps  
Size: 124 x 224 x 174  
Weight: 3.5 kg  
Power: ~7W





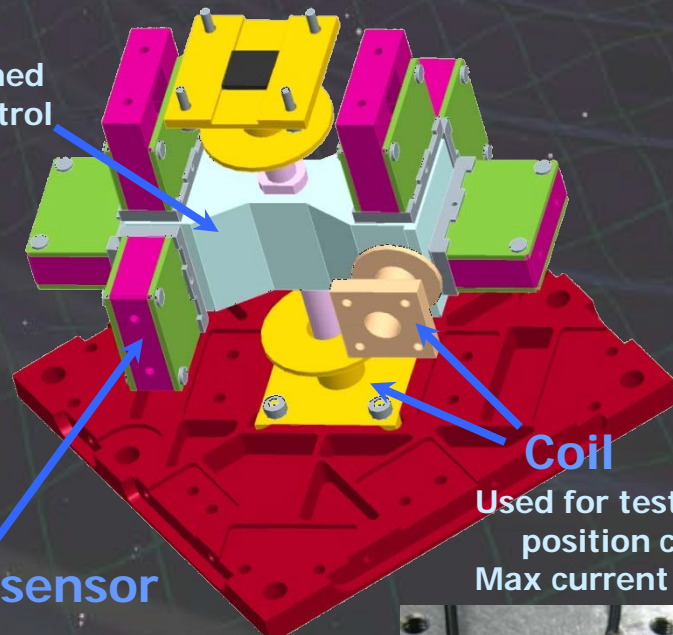
# SWIM $\mu$ v

Tiny GW detector ~47g test masses inside  
→ Levitated control in space

TAM: Torsion Antenna Module with free-falling test mass  
(Size : 80mm cube, Weight : ~500g)

## Test mass

~47g Aluminum, Surface polished  
Small magnets for position control

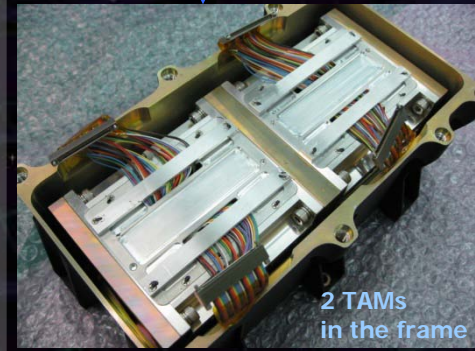
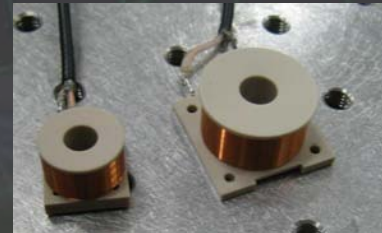


Coil

Used for test-mass  
position control  
Max current ~100mA

Photo sensor

Reflective-type optical  
displacement sensor  
Separation to mass ~1mm  
Sensitivity ~  $10^{-9}$  m/Hz<sup>1/2</sup>  
6 PSs to monitor mass motion





# Successful control

SWIM

In-orbit operation

Test mass controlled

Error signal  $\rightarrow$  zero

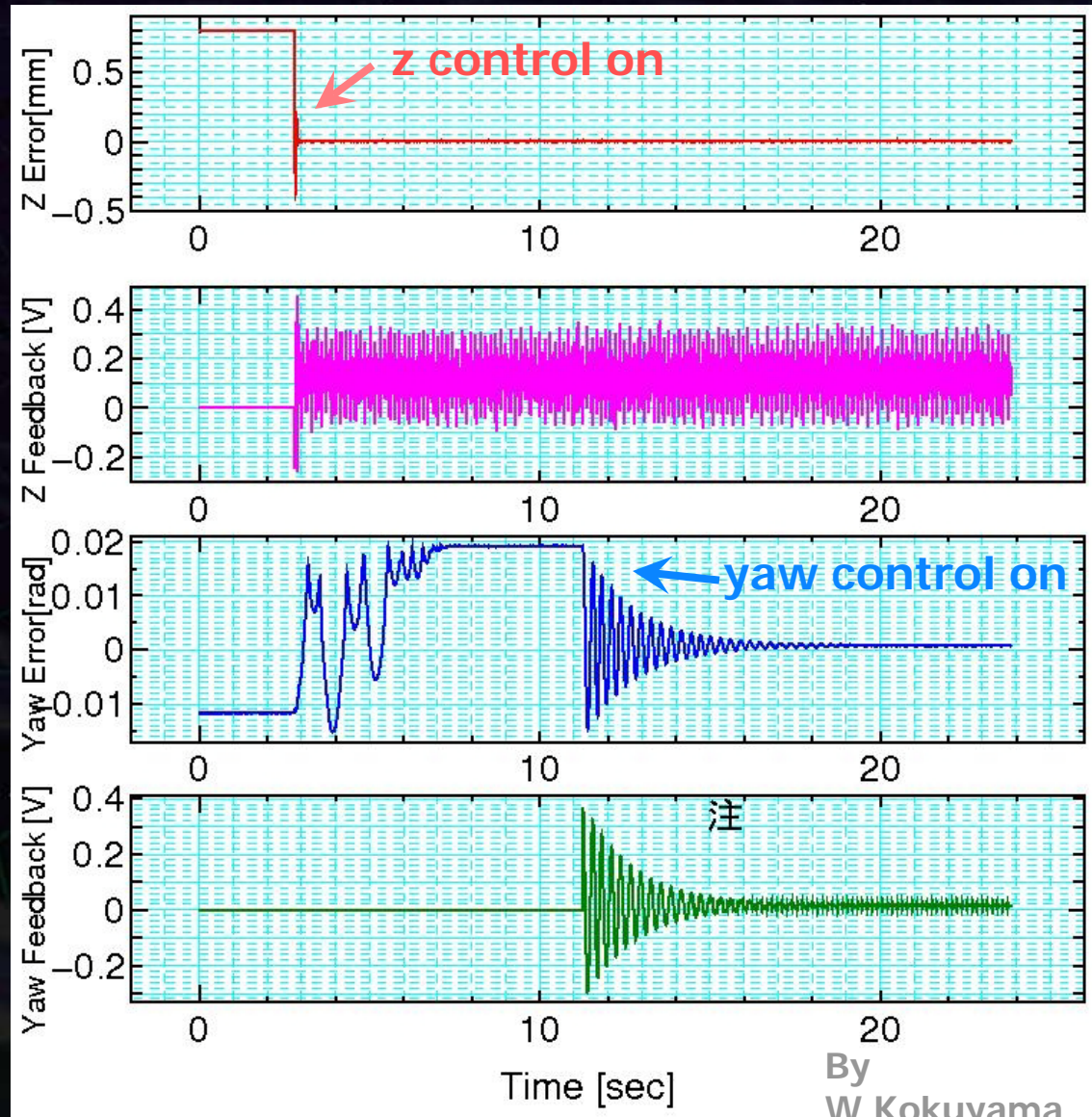
Damped oscillation  
(in pitch DoF)

Free oscillation  
in x and y DoF

Signal injection  
 $\rightarrow$  OL trans. Fn.

Operation: May 12, 2009

Downlink: ~ a week



By  
W. Kokuyama