

Possibility of contribution to ESA L3 from Japan

- ※ This is not an officially qualified report from JAXA. This is presented from Ando's personal viewpoint as a representative of the Japanese space GW community.

Masaki Ando (Univ. of Tokyo / NAOJ)

- ※ Previous JGWC chairperson, DPF WG leader,
Member of JAXA space science committee

- JAXA-ISAS encourages international collaboration in general.
 - Sending person to ESA L1-L3 discussions.
- Informal discussion with **Saku Tsuneta** (Director General of JAXA-ISAS) on possibility to join ESA L3 (July 31th).
 - - Encourages bottom-up proposals from the community.
 - Suggested a presentation from community view points in GOAT #5.
- Japanese space GW community has been working on DECIGO and its precursor missions. Redefined the strategy in 2014. → Possibility to be a **LISA Junior partner**.

- **JAXA Mission Opportunities**
- **Activities in Japan**
- **Conclusion**

- Three categories in JAXA's

'Space Science and Exploration program' (2014)

- (1) Medium-scale missions

(Astro-H, Solar-C, Hayabusa2, ESA JUICE, ATHENA, ...)

- Cost ~\$300M, Development term 5-7 years.
- AO every 4 years.

- (2) Small-scale missions (Hisaki, ERG, ...)

- Cost \$100M~\$150M, Development term 4 years.
- AO every 2 years.

- (3) Various small projects

- Small rocket, Balloon, International Junior partner, Science mission using ISS, ...
- ~\$10M/year.

Mission Plan by JAXA



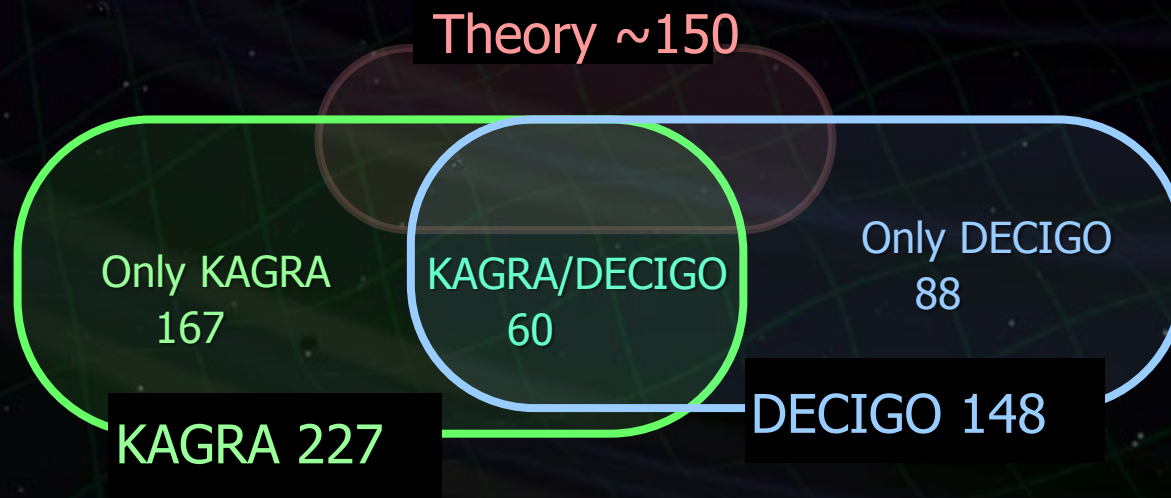
From file submitted to the government by ISAS/JAXA

(内閣府・宇宙政策委員会・宇宙科学・探査部会 2013年9月19日)

分類	ミッション・事業名称	状況	第2期中期計画				第3期中期計画				第4期中期計画				第5期中期計画				備考						
			FY2008	FY2009	FY2010	FY2011	FY2012	FY2013	FY2014	FY2015	FY2016	FY2017	FY2018	FY2019	FY2020	FY2021	FY2022	FY2023		FY2024	FY2025	FY2026	FY2027		
戦略的に実施する 中型計画	はやぶさ2	開発中	▲ P/J移行 ▲ P/J移行				▲ 打上				▲ 小衛星到着 ▲ 地球帰還														
	ASTRO-H	開発中	▲ P/J移行 ▲ PDR ▲ CONT.?				▲ 打上				▲ ...														
	将来計画 (仮称:M1-M4) 4年に1回AO発出 開発期間6年 (5~7年)	計画中	★ 公募(AO) ▲ P/J移行				▲ 打上				▲ ...				FY2021(20-22)▲										
			★ 公募(AO) ▲ P/J移行				▲ 打上				▲ ...				FY2025(26-28)▲ FY2029(29-30)▲ FY2033(32-34)▲										
公募型 小型計画	感星光学衛星	開発中	▲ SDR/P/J移行 ▲ 打上				▲ ...																		
	ジオスペース探査衛星	開発中	▲ MDR/SDR ▲ DRP ▲ SDR PDR ▲ 打上				▲ ...																		
	BepiColombo	開発中	▲ CSR ▲ 打上				▲ ...				▲ 火星到着 ▲ ...														
	将来計画 (仮称:S1-S7) 2年に1回AO発出 開発期間4年	計画中	★ 公募(AO) ▲ P/J移行				▲ 打上				▲ ...				FY2018▲ FY2020▲ FY2022▲ FY2024▲ FY2026▲ FY2028▲ FY2030▲										
多様な小規模 プロジェクト群	計画中	★ 計画決定				▲ 具体的な提案の状況に応じて、随時AOの発出・計画決定・実行する。																			
基盤的 活動費	継続的に 実施中	将来のミッション創出につながる学術研究・実験等の推進や衛星運用、施設維持の実施に必要な活動費。 従前より効率化努力を行ってきたこと、要する効率的な執行に努める。																							

- **JAXA Mission Opportunities**
- **Activities in Japan**
- **Conclusion**

- JGWC (Japan Gravitational Wave Community) :
 - 325 members (in 2014)
 - Information exchange
(conferences, announcement of new position, ...)
 - Discuss future strategy



KAGRA and DECIGO

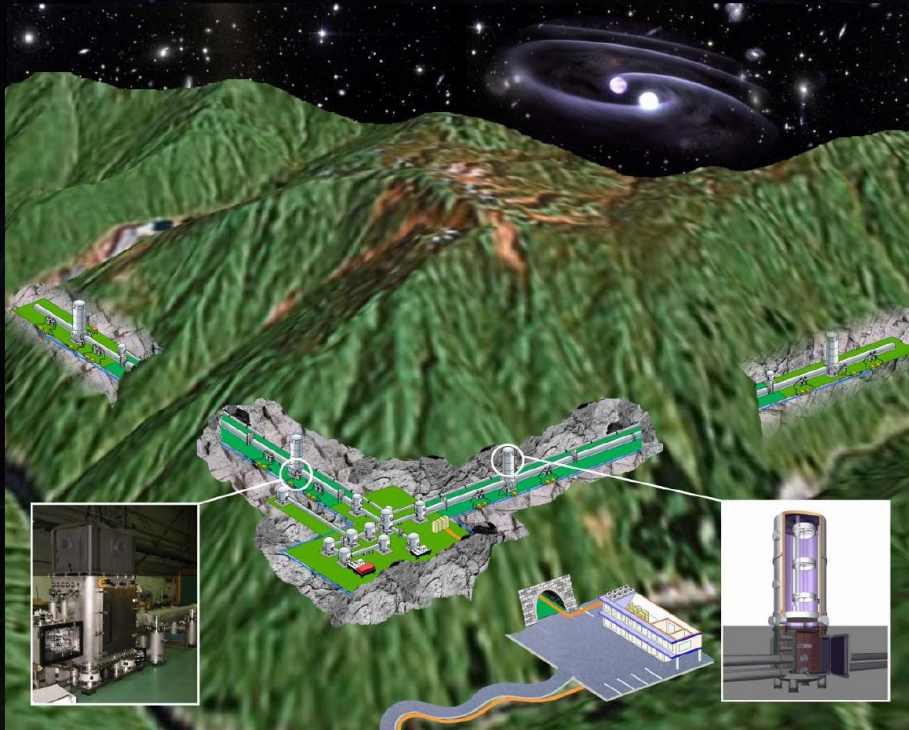


KAGRA (~2017)

Ground-based Detector

→ High-freq. (100Hz) GW event

Target: Detection and Astronomy

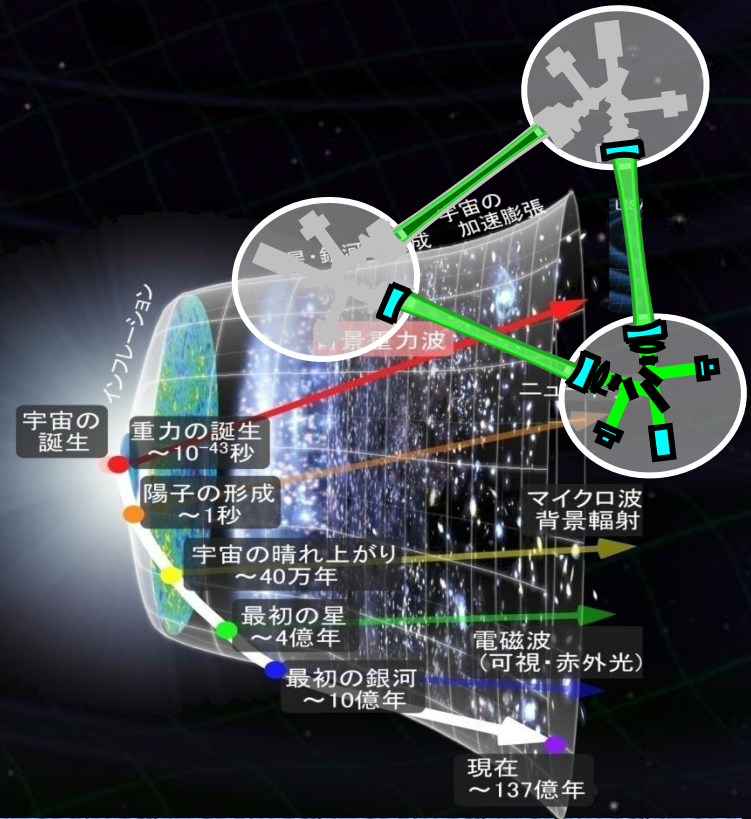


DECIGO (~2030)

Space observatory

→ Low-freq. (0.1Hz) GW signal

Target: Cosmology



Space GW Antenna DECIGO



DECIGO (DECI-hertz interferometer Gravitational wave Observatory)

Purpose: To Obtain Cosmological Knowledge.

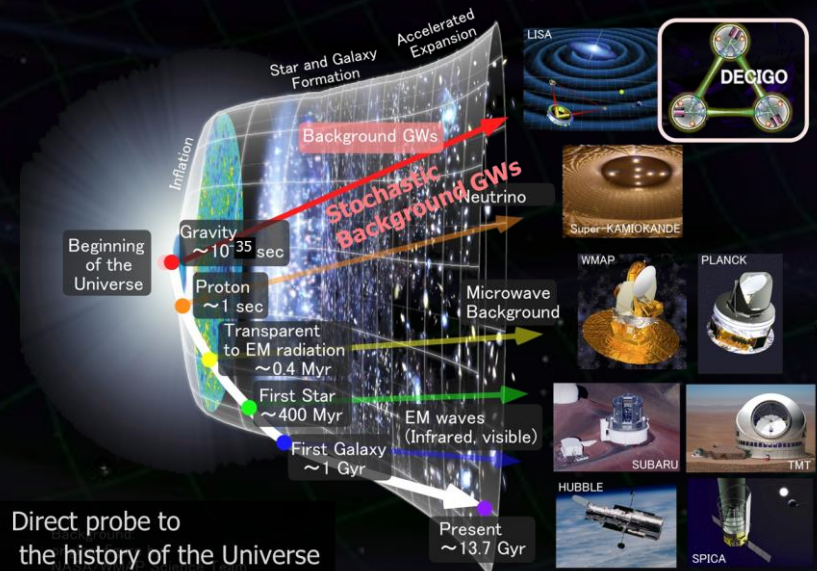
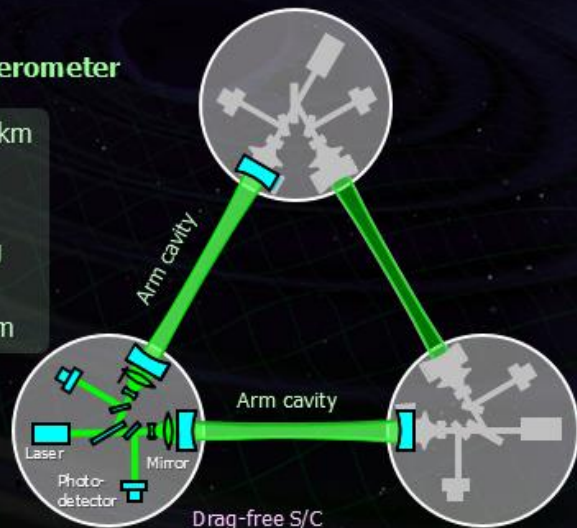
Direct observation of the origin of space-time and matter in Big-bang Universe.

Interferometer Unit:

Differential FP interferometer

Arm length: 1000 km
Finesse: 10
Mirror diameter: 1 m
Mirror mass: 100 kg
Laser power: 10 W
Laser wavelength: 532 nm

S/C: drag free
3 interferometers

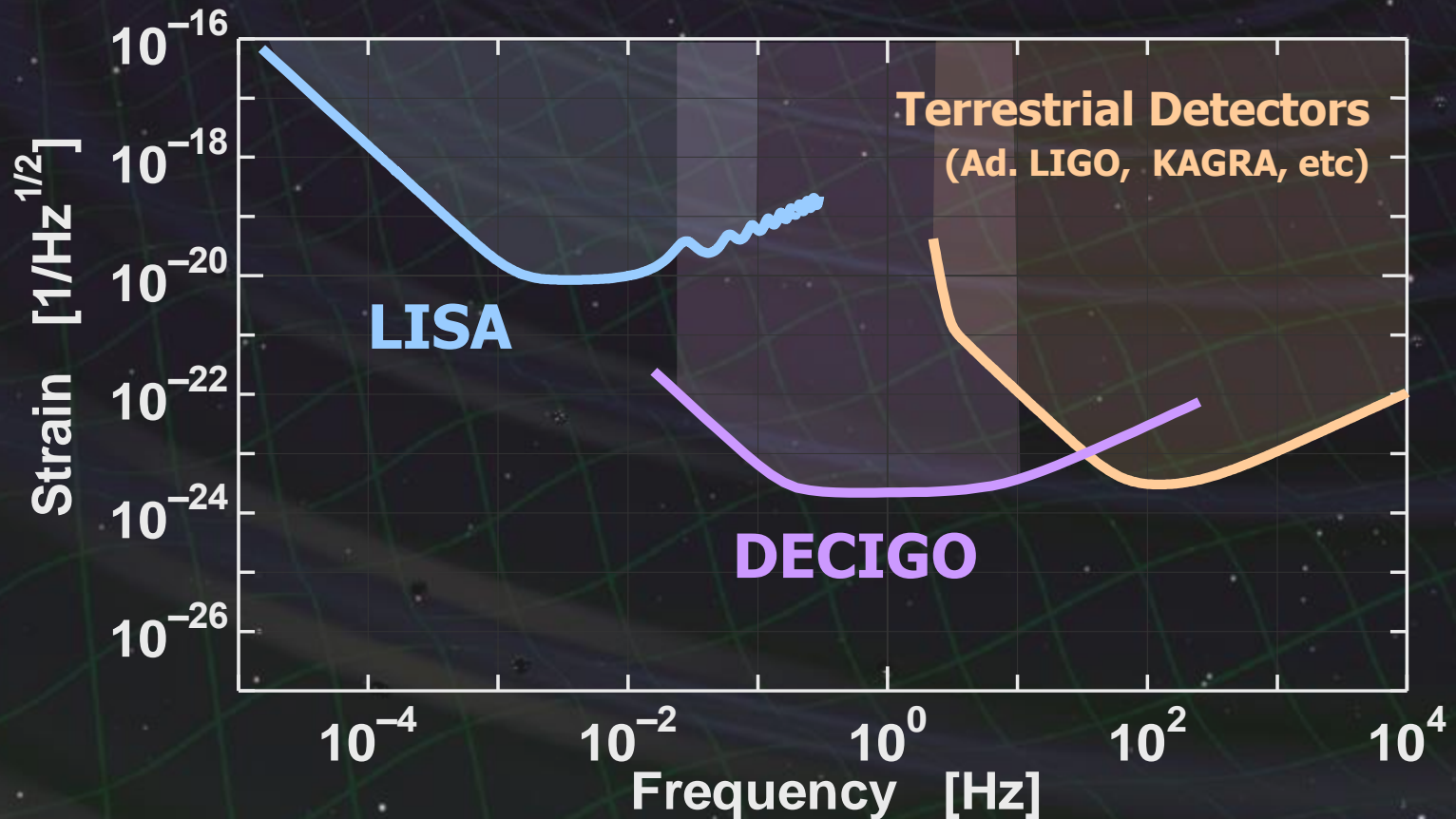


DECIGO (Deci-hertz interferometer Gravitational wave Observatory)

Space GW antenna (~2030)
Obs. band around 0.1 Hz



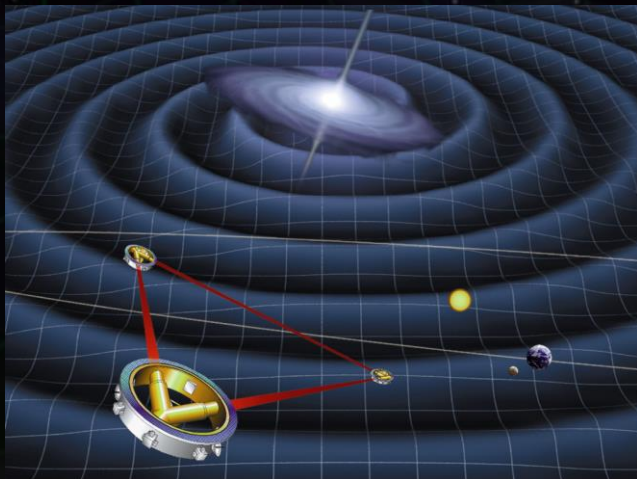
'Bridge' the obs.gap between
LISA and Terrestrial detectors



eLISA

(Laser Interferometer Space Antenna)

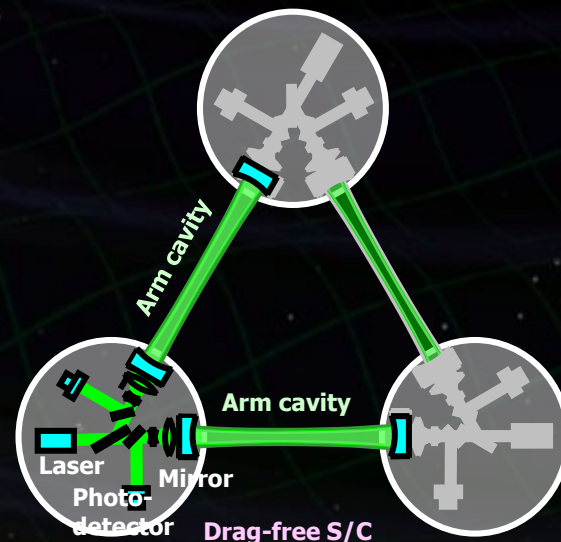
- Target: SMBH, Binaries.
~1mHz GW signals.
- Baseline : 100-500 Mkm.
Constellation of 3 S/C.
- Config.: Optical transponder.



DECIGO

(Deci-hertz Interferometer
Gravitational Wave Observatory)

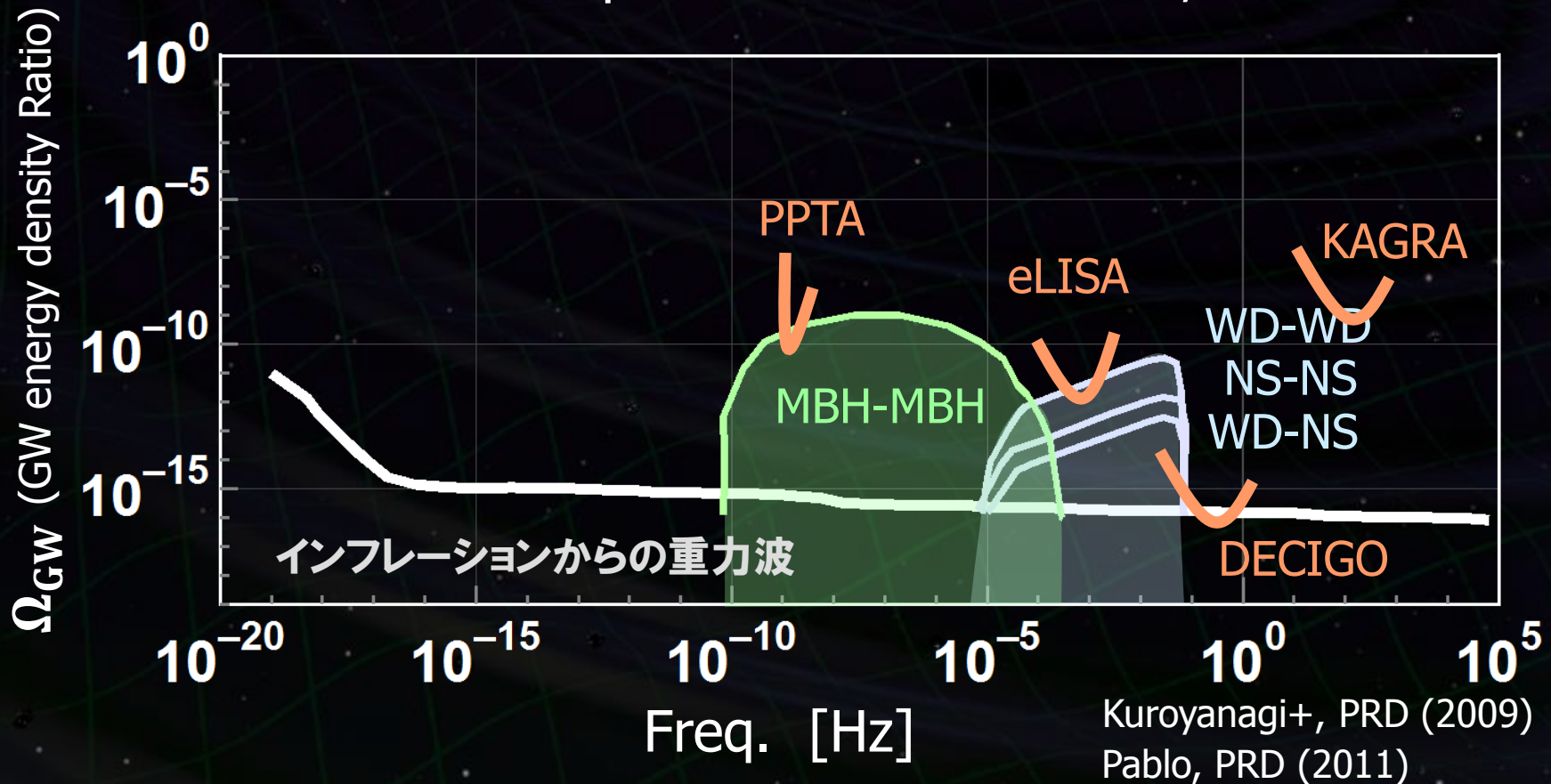
- Target: IMBH, NS, GW Background.
~0.1Hz GW signals.
- Baseline : 1000 km.
Formation Flight by 3 S/C.
- Config.: FP Interferometer.



Foreground GW

GWs from many binaries \rightarrow Unresolvable noises.

\Rightarrow Foreground noise for primordial GW observation
at freq. bands $10^{-10} - 0.1$ Hz,



- Activity for DECIGO and DPF.

- * 2001 First proposal paper on DECIGO.

- * 2005 Establishment of the first roadmap.

- DPF → Pre-DECIGO → DECIGO

- * 2007 DPF working group approved by JAXA-ISAS.

- * 2009 DPF Mission proposal → Fail (in final two).
(2010 the KAGRA project started)

- * 2014 DPF Mission proposal → Fail (in the first stage).

- 2014 Revision of the DECIGO roadmap.

- Ground test/Piggy back → Pre-DECIGO → DECIGO

- Pre-DECIGO mission design development of key tech.

- Possibility of international collaboration

Previous Roadmap for DECIGO

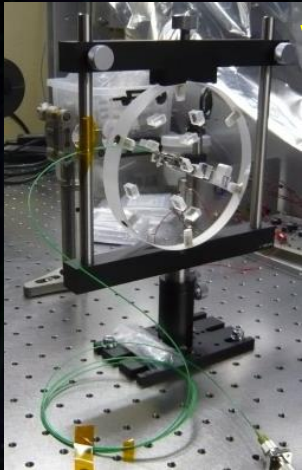
Figure: S.Kawamura

	2014	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30	31	32	33
Mission																				
Purpose	Interferometer in Space (Obs. Of GW and Earth Gravity)						Long-baseline Precise Formation Flight (GW Observation)						GW Astronomy and Cosmology							
Design	One Small Satellite Short FP cavity + Drag-free						FF with 3 S/C 1 IFO unit						FF with 3 S/C 3-4 IFO units							

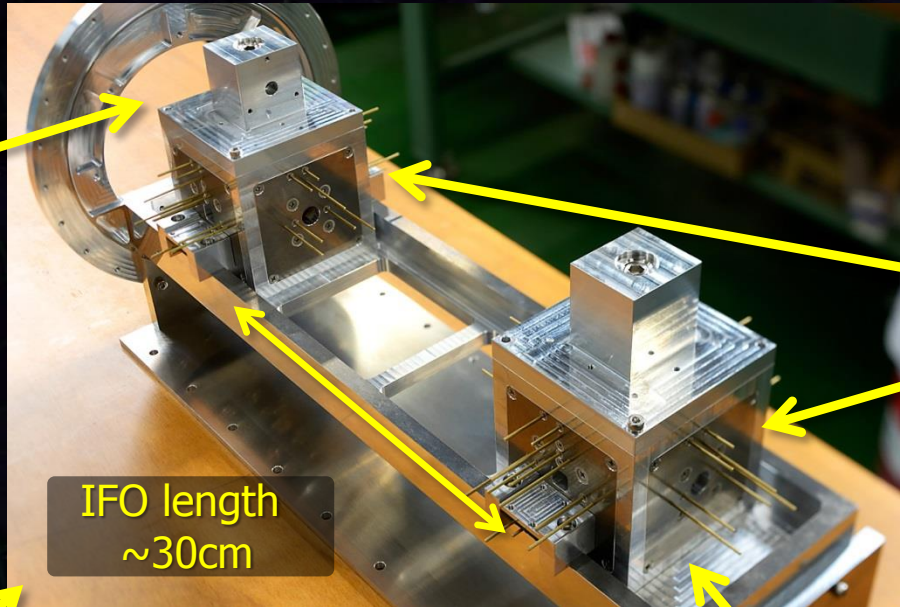
Interferometer Module

IO Optics

Monolithic opt. bench by silicate bonding



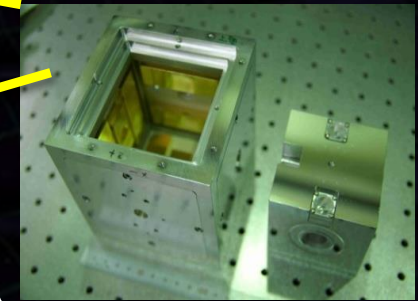
Interferometer Module



IFO length
~30cm

Test mass module

TM, Capacitive
Sensor/Actuator,
Launch lock



Quad-RFPD

Quadrant PD +
Demod. circuits for
length and alignment
control signals



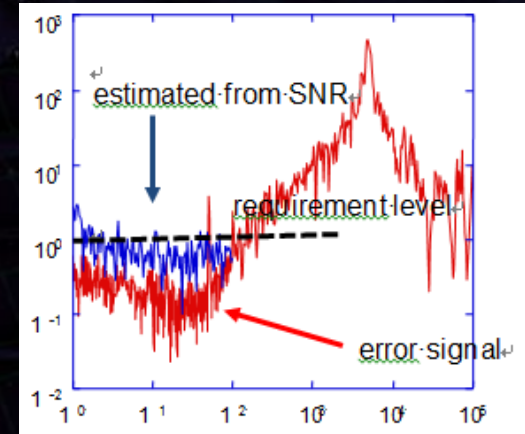
SpW signal- processing board

SpW FPGA +
16bit AD/DA

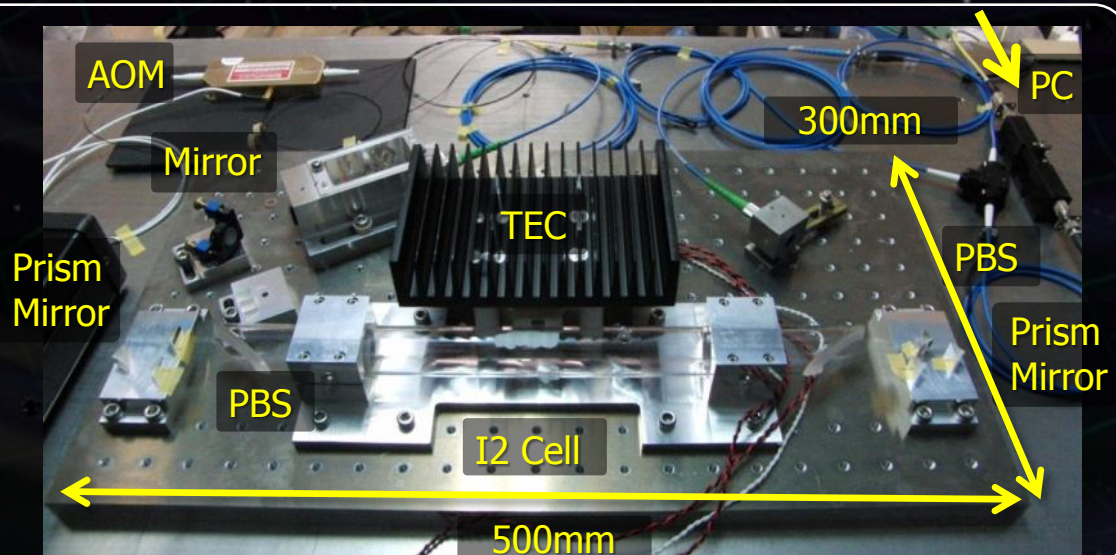
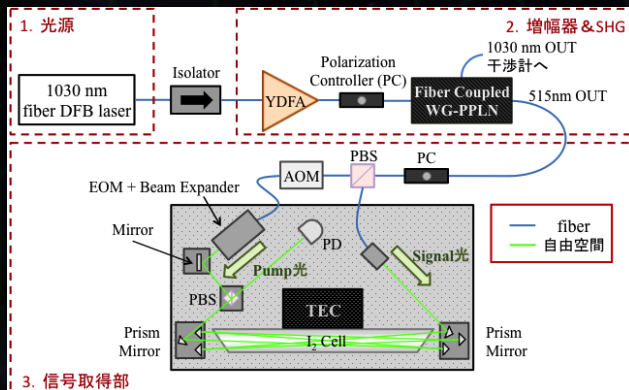


Frequency Stabilization Module

- Frequency Stabilization module BBM2 (at UEC)
 - Use absorption line of Iodine molecule.
 - Satisfy requirement ($0.5 \text{ Hz/Hz}^{1/2}$) in error-signal measurement.
 - Preparing one-more module for relative stability evaluation.



Freq. Stab module



Space Demonstration by SWIM



Photo:
JAXA

SWIM (Space wire demonstration module) on SDS-1

Launched in Jan. 2009, 1.5 yr operation.

➔ First demonstration of a space GW detector



SpaceCube2: Space-qualified Computer

CPU: HR5000

(64bit, 33MHz)

System Memory:

2MB Flash Memory

4MB Burst SRAM

4MB Asynch. SRAM

Data Recorder:

1GB SDRAM

1GB Flash Memory

SpW: 3ch

Size: 71 x 221 x 171

Weight: 1.9 kg

Power: 7W



Photo by JAXA

SWIM_{μv} : User Module

Processor test board

GW+Acc. sensor

FPGA board

DAC 16bit x 8 ch

ADC 16bit x 4 ch

→ 32 ch by MPX

Torsion Antenna x2

~47g test mass

Data Rate : 380kbps

Size: 124 x 224 x 174

Weight: 3.5 kg

Power: ~7W

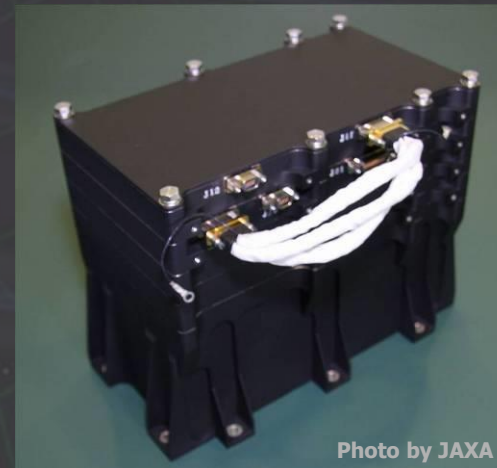


Photo by JAXA

**SDS-1
Bus System**

Power +28V
RS422 for CMD/TLM
GPS signal

Power ±15V, +5V
SpW x2 for CMD/TLM

Rotating TOBA : SWIM μ v



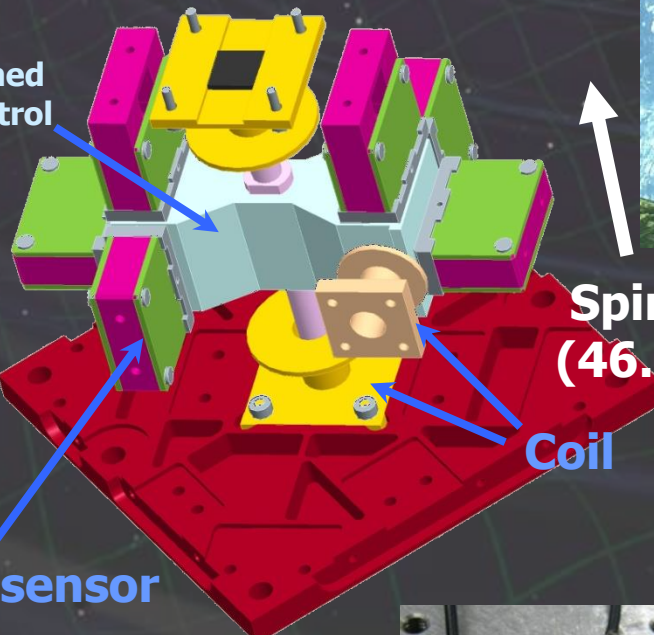
Small Module SWIM μ v on SDS-1

Launched Jan. 2009, Terminated Sept. 2010

TAM: Torsion Antenna Module with free-falling test mass
(Size : 80mm cube, Weight : ~500g)

Test mass

~47g Aluminum, Surface polished
Small magnets for position control



Spin Axis
(46.5mHz)

Coil

Photo sensor

Reflective-type optical displacement sensor
Separation to mass ~1mm
Sensitivity ~ 10^{-9} m/Hz^{1/2}
6 PSs to monitor mass motion

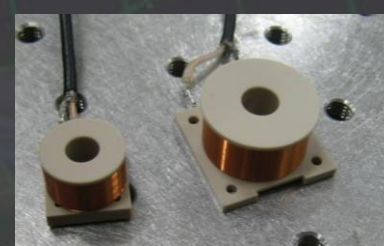
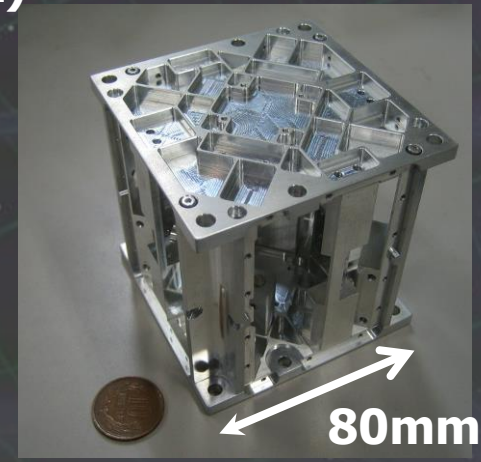


Photo:
JAXA

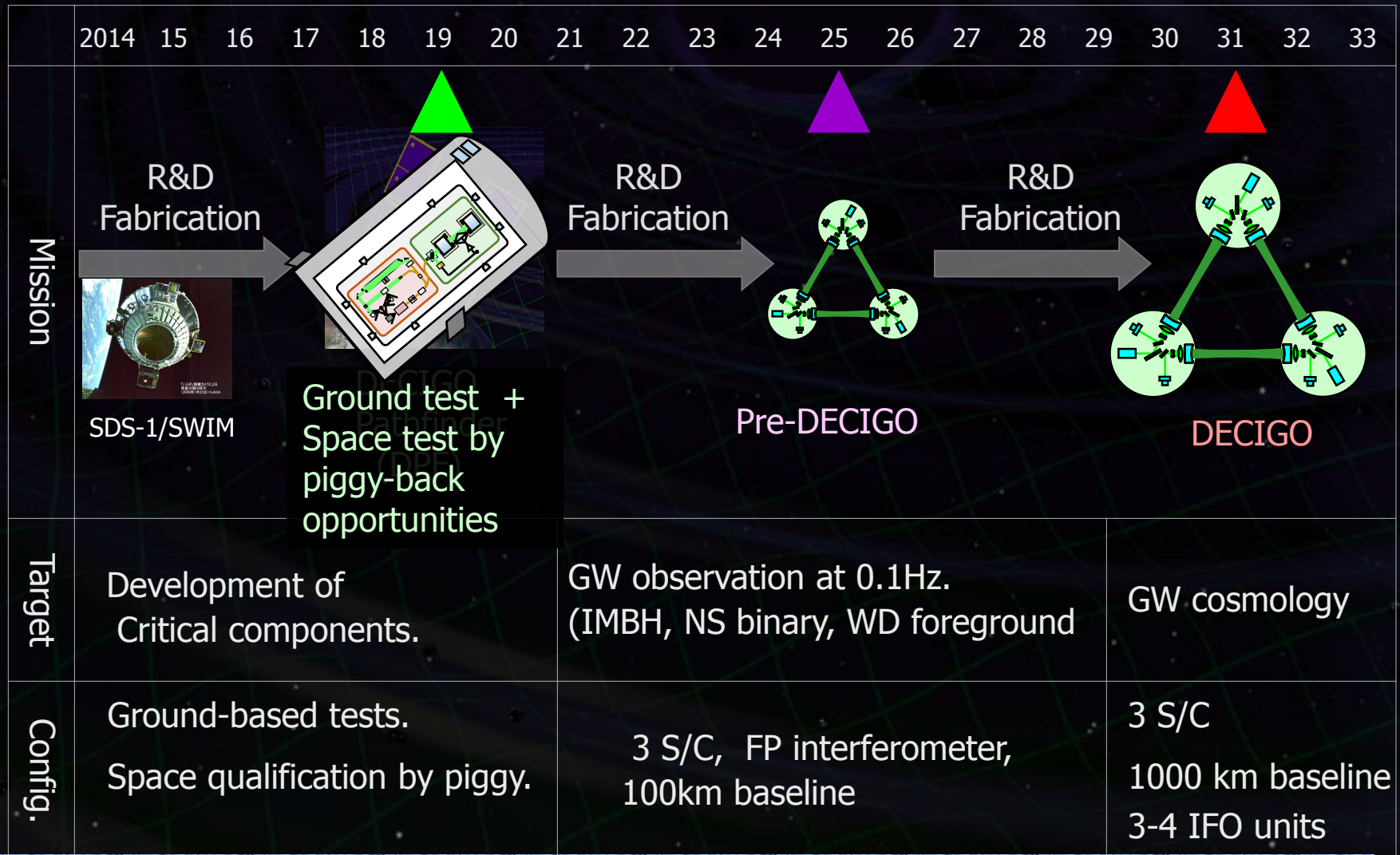
「いぶき」搭載カメラによる
衛星分離の様子
(2009年1月23日) ©JAXA



80mm

DECIGO Roadmap (Revised in 2014)

Figure: S.Kawamura



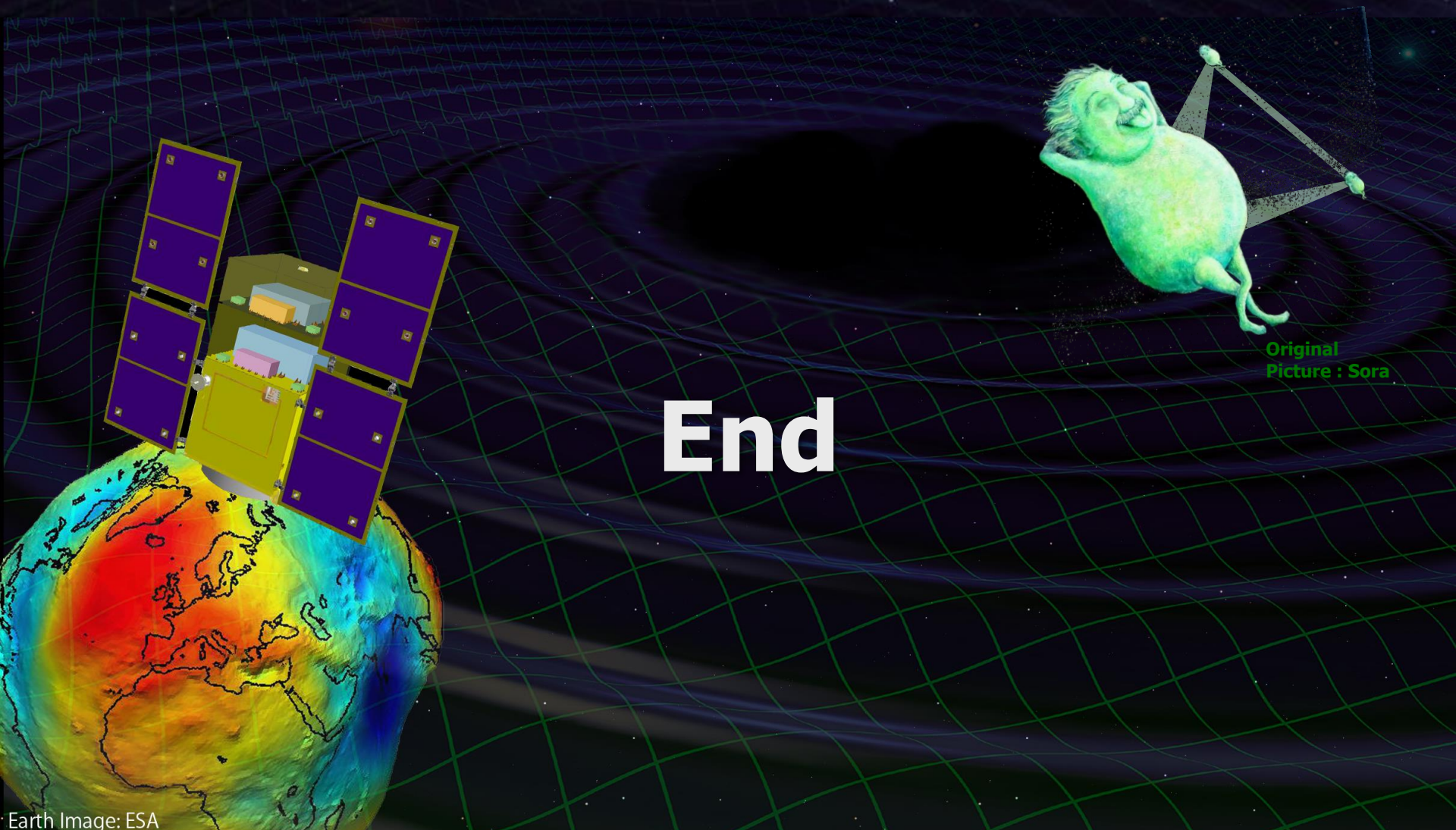
- **JAXA Mission Opportunities**
- **Activities in Japan**
- **Conclusion**

Summary from Community PoV (1/2)

- GW community in Japan has ~45 years experiences on ground-based GW detectors, and ~15 years activities for space GW antenna. It also have a little experiences on space missions.
- Current first priority in Japanese GW community is KAGRA, and space GW antenna is the next step. Since JAXA highly requires first-priority support from the community, there will be a small chance to be selected as medium/small-scale mission. However there will be a **chance in the 'Various small project' category in an international framework.**

Summary from Community PoV (2/2)

- Since **DECIGO will provide fruitful science** which never obtained by EM observations, it will be realized at last. Pre-DECIGO will have a lot of original science, different from and complementary with those of LISA. It is important to keep the (minimum) activities: system design study and development of key components.
- Contribution to LISA will be a good opportunity to make solid basis for the space GW community in Japan. Junior partner will be a realistic choice. Candidates: Laser source, Freq. stabilization, Part of Interferometer, Data analysis, ...



End

Original
Picture : Sora

Earth Image: ESA

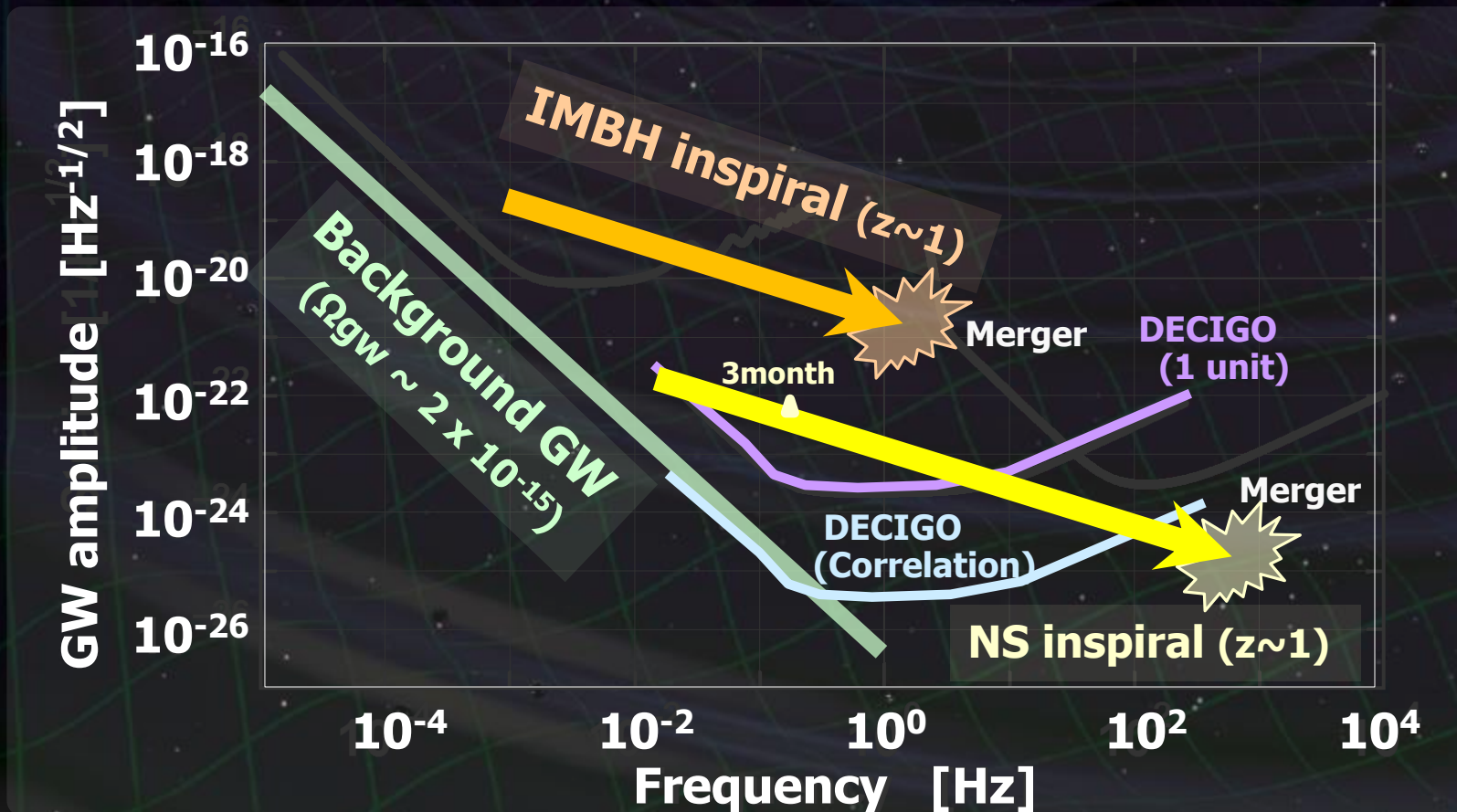
Supplementary Slides

Targets and Science

IMBH binary inspiral
NS binary inspiral
Stochastic background

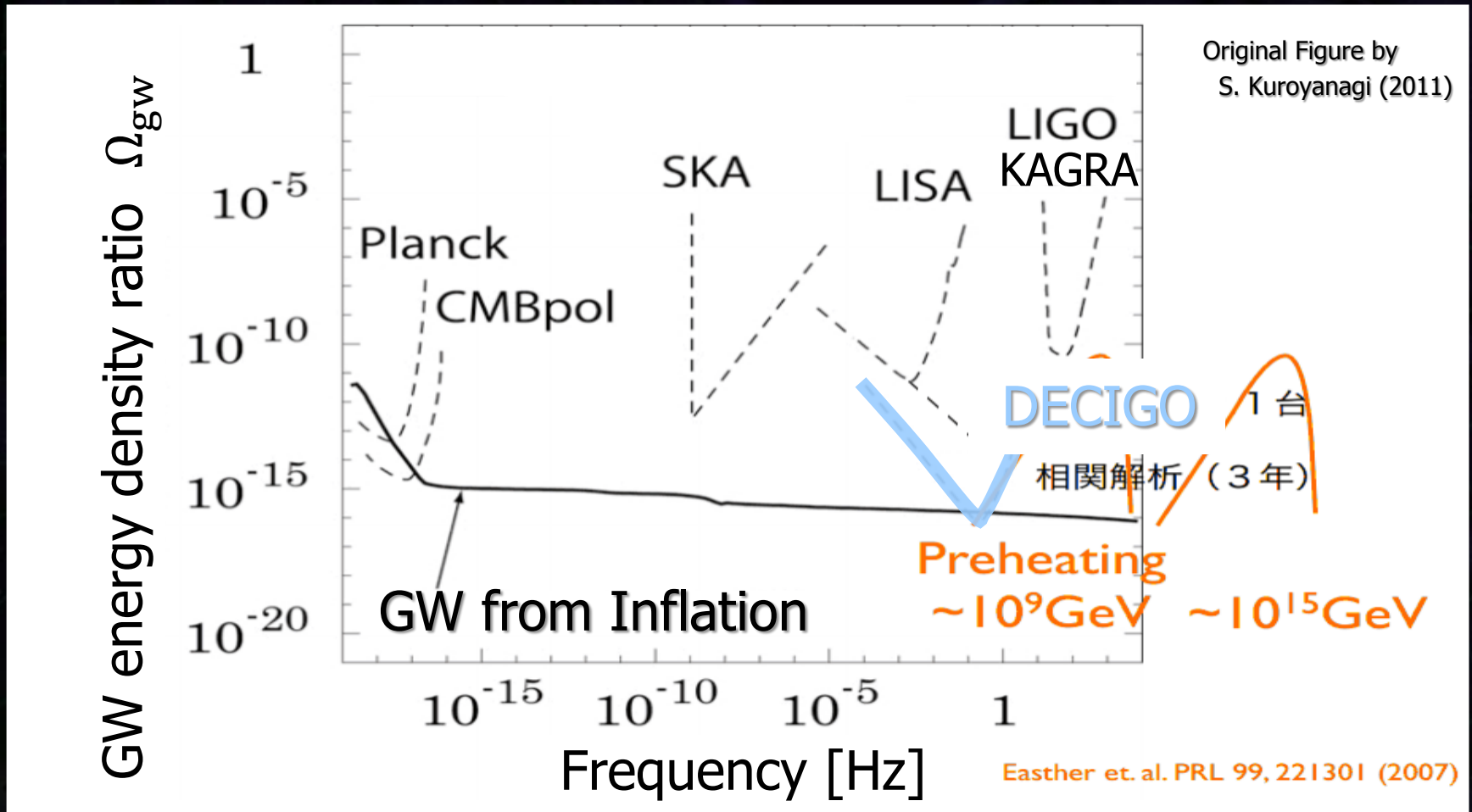


Galaxy formation (Massive BH)
Cosmology (Inflation, Dark energy)
Fundamental physics



Primordial GW

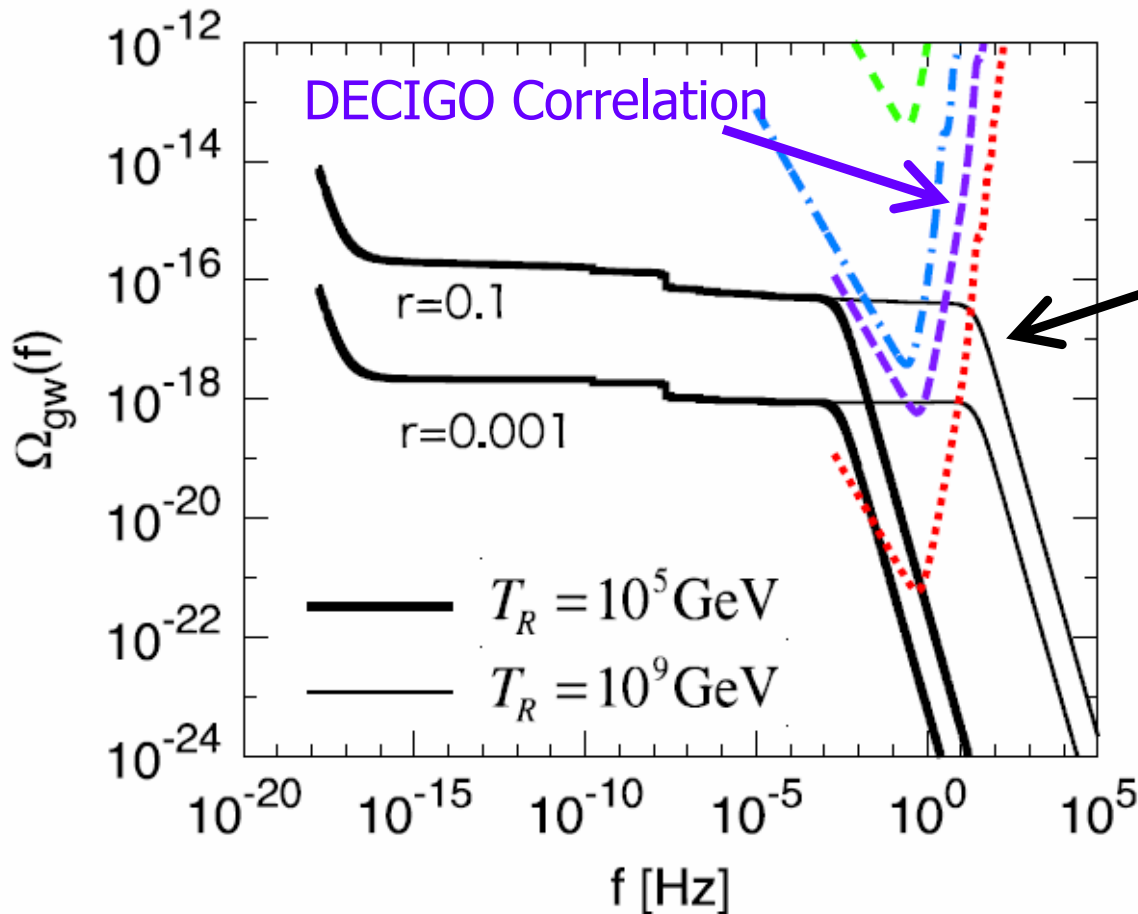
Earlier universe \rightarrow Smaller horizon scale \rightarrow High GW freq.



GW from Inflation

Energy density \propto Tensor-Scalar Ratio (r).

Power spectrum : Evolution history of the Universe.



- Spectrum Power.
→ Energy scale of inflation
- Cut-off freq.
→ Energy scale of Reheating

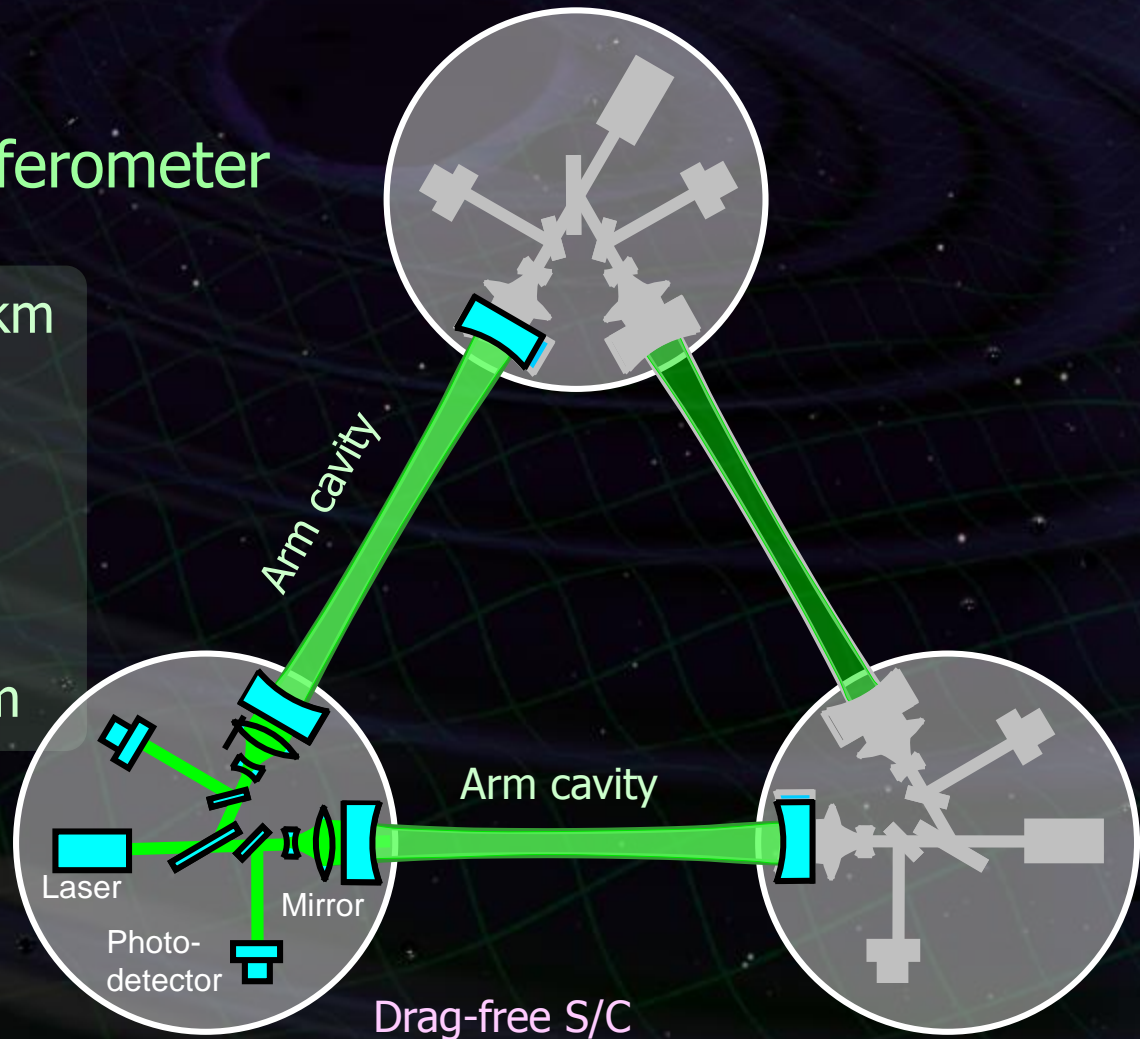
Nakayama+,
Journal of Cosmology
and Astroparticle Physics
06 (2008) 020.

Interferometer Unit:

Differential FP interferometer

Arm length:	1000 km
Finesse:	10
Mirror diameter:	1 m
Mirror mass:	100 kg
Laser power:	10 W
Laser wavelength:	532 nm

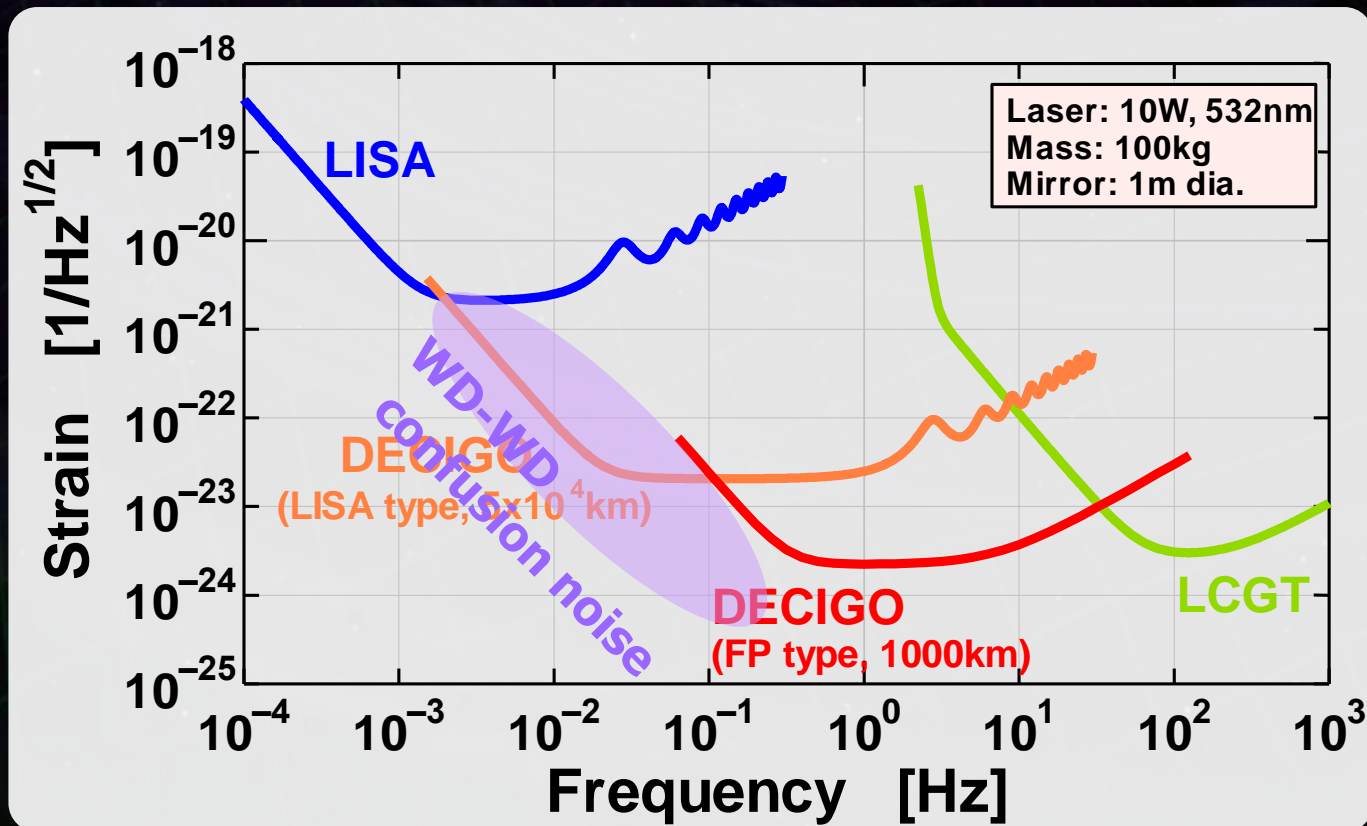
S/C: drag free
3 interferometers



Transponder type vs Direct-reflection type

Compare : Sensitivity curves and Expected Sciences

⇒ Decisive factor: Binary confusion noise



Arm length

Cavity arm length : Limited by diffraction loss

Effective reflectivity ($TEM_{00} \rightarrow TEM_{00}$)

Laser wavelength : 532nm

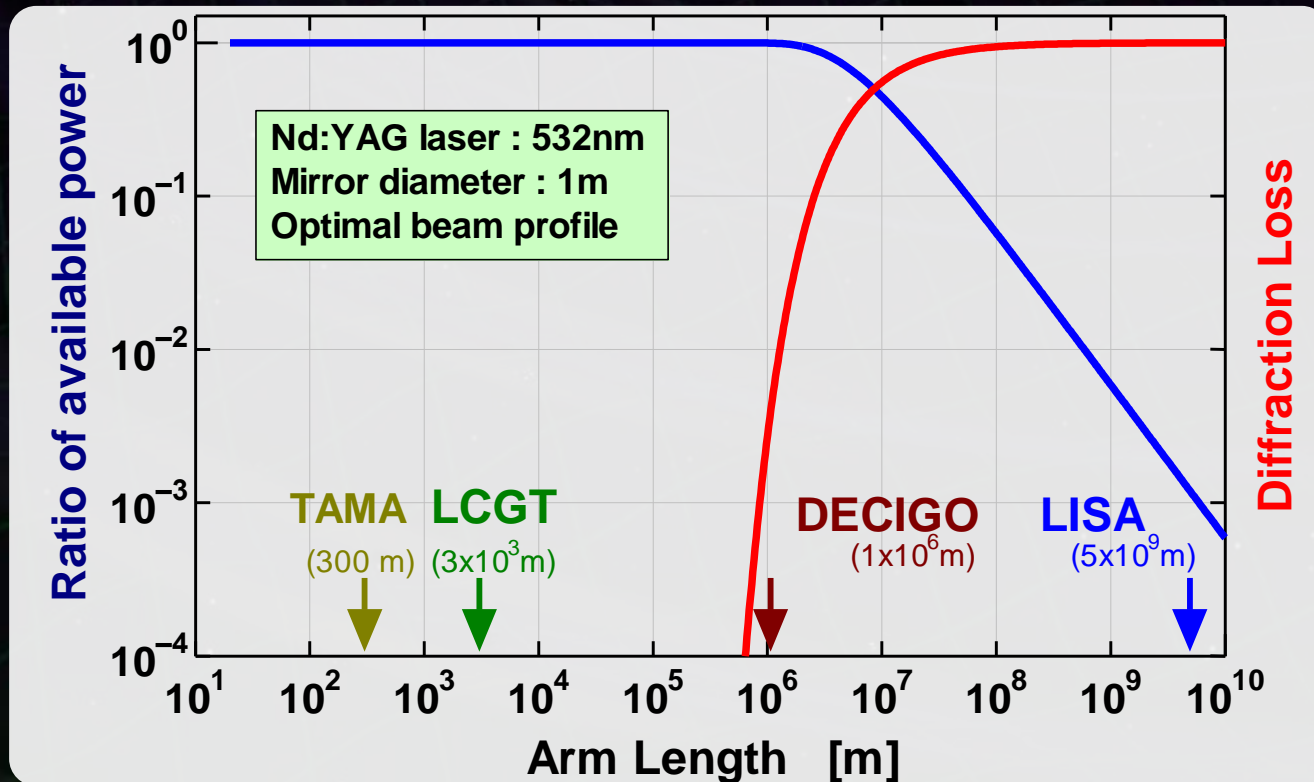
Mirror diameter: 1m

Optimal beam size



1000 km

is almost max.



Cavity and S/C control

Cavity length change

PDH error signal \rightarrow Mirror position (+Laser freq.)

Relative motion between mirror and S/C

Local sensor \rightarrow S/C thruster

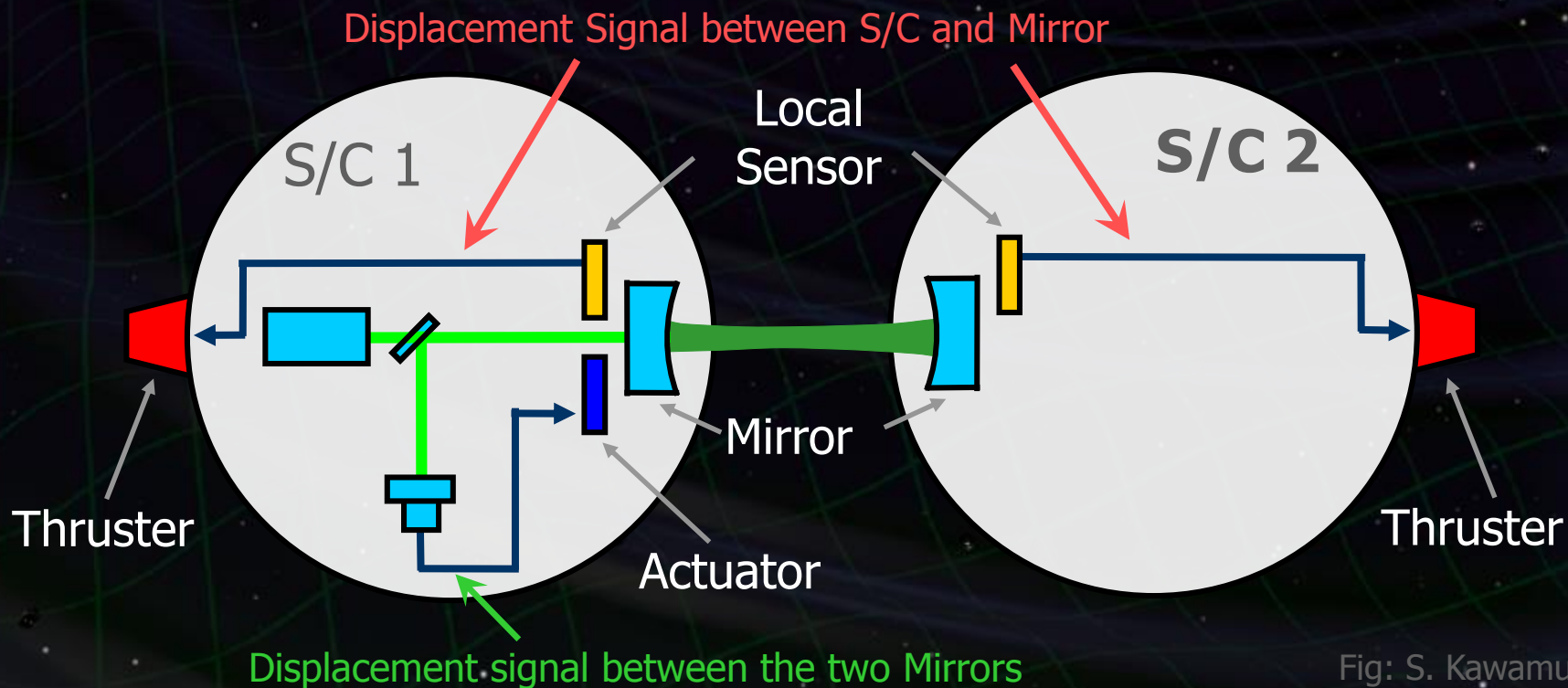


Fig: S. Kawamura

Displacement Noise

Shot noise $3 \times 10^{-18} \text{ m/Hz}^{1/2}$ (0.1 Hz)

⇒ x 10 of KAGRA in phase noise

Other noises should be well below the shot noise

Laser freq. noise: $1 \text{ Hz/Hz}^{1/2}$ (1Hz)

Stab. Gain 10^5 , CMRR 10^5

Acceleration Noise

Force noise $4 \times 10^{-17} \text{ N/Hz}^{1/2}$ (0.1 Hz)

⇒ x 1/50 of LISA

External force sources

Fluctuation of magnetic field, electric field,
gravitational field, temperature, pressure, etc.

Foreground Cleaning

DECIGO obs. band: free from WD binary foreground
→ Open for cosmological observation

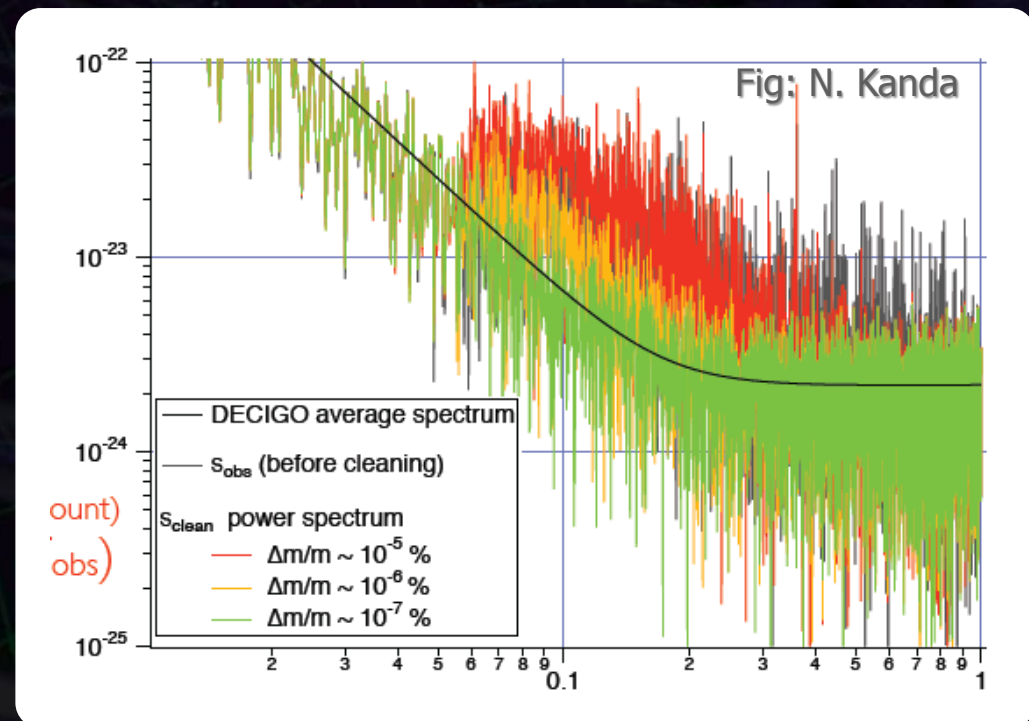
DECIGO will watch
 $\sim 10^5$ NS binaries

⇒ **Foreground for GWB**

In principle, possible
to remove them.

Require waveform

Accuracy $\Delta m/m < \sim 10^{-7}$ %



Considering “Conceptual design”

By T.Akutsu

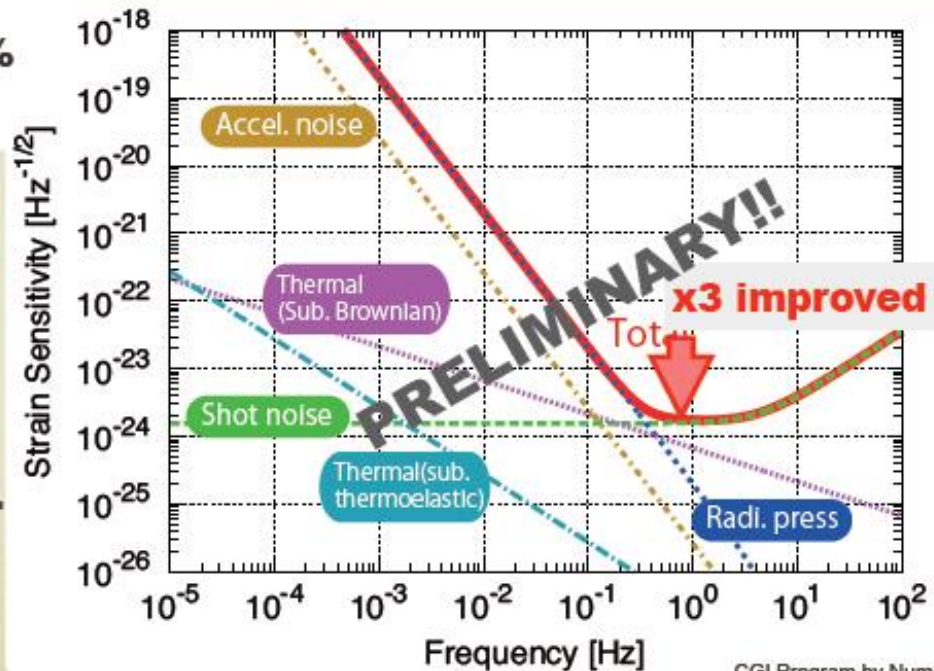
- Arm length: **1,500 km**
- Laser power: **30 W**
- Laser wavelength: **532 nm**
- Mirror diameter: **1.5 m**
- Mirror mass: **100 kg**
- Mirror reflectivity: **77.3%**
- Cavity g-param: **0.1**

Preliminary
← Parameters tuned

This is the first step to considering the **conceptual design**.

Next:

- ➔ Confirm the calculations.
- ➔ Find the realistic way to realize this!



初期宇宙からの重力波観測

BICEP2, (POLARBEAR,...)

マイクロ波望遠鏡を用いた
宇宙背景放射 B-mode偏光
成分の観測.

DECIGO, (KAGRA, aLIGO,...)

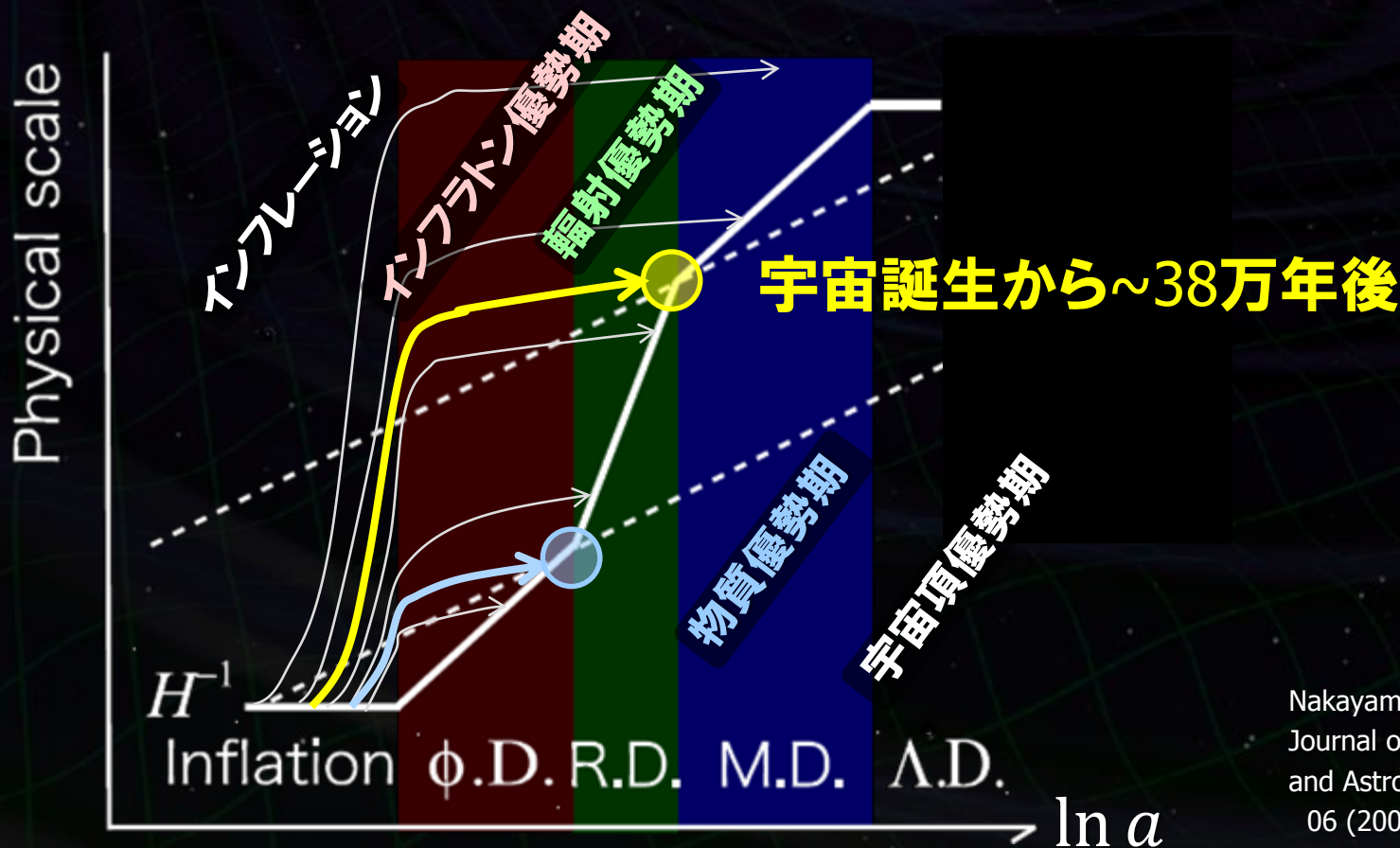
重力波望遠鏡を用いた
宇宙背景重力波の観測.



図: 田島氏談話会資料より(2011 京都大学)

インフレーションからの重力波

計量の量子揺らぎとして生成 → 初期に生成された重力波ほど、長くインフレーションで引き延ばされ、最近に宇宙の地平線内へ。



Nakayama+,
Journal of Cosmology
and Astroparticle Physics
06 (2008) 020.

重力波のエネルギー密度比


重力波のエネルギー密度

$$\Omega_{\text{GW}}(f) = \frac{1}{\rho_c} \frac{d\rho_{\text{GW}}(f)}{d \ln f}$$

宇宙の臨界密度

等価な重力波振幅

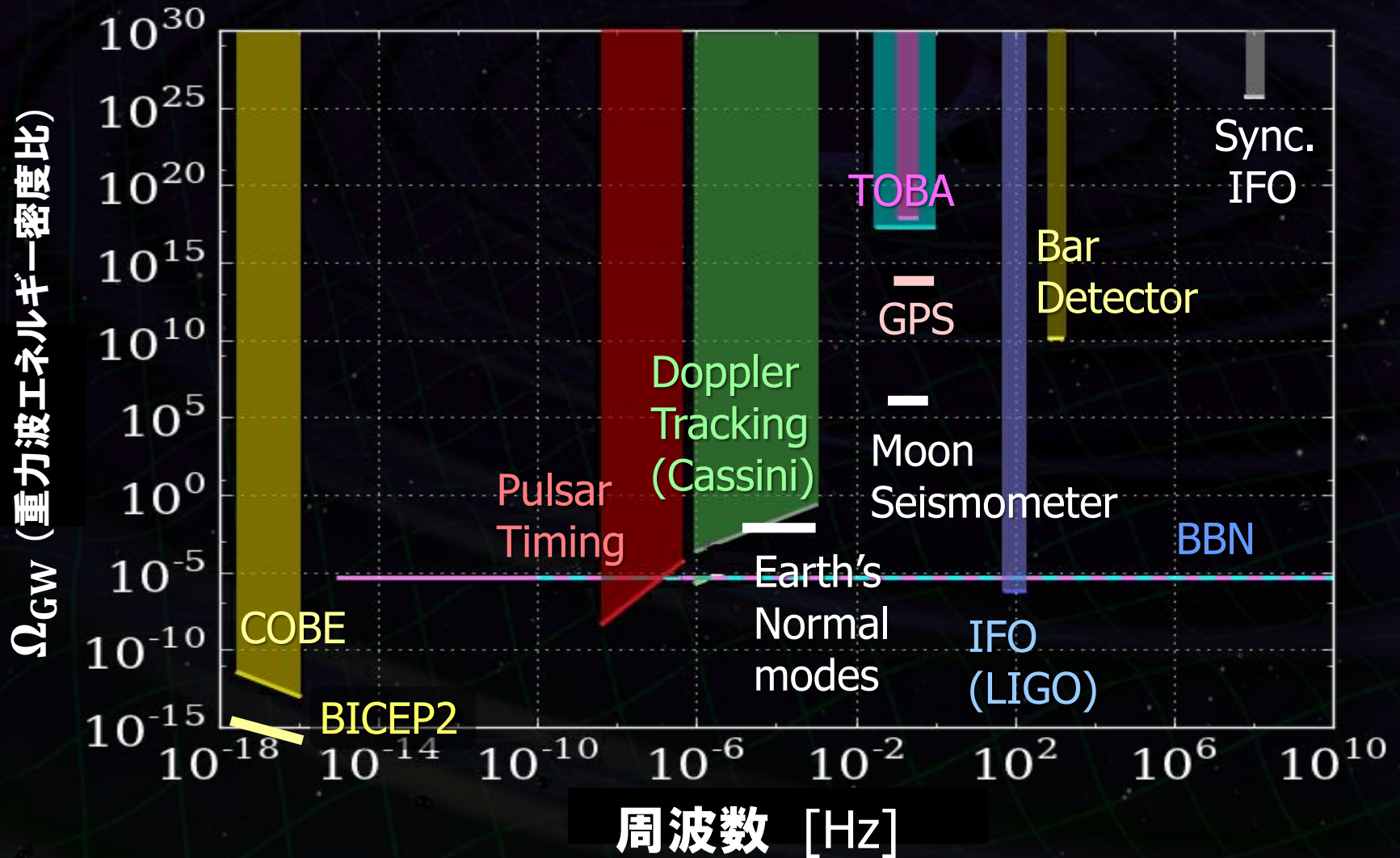
ハッブル定数


$$\tilde{h}_{\text{GW}}^2(f) = \frac{3H_0^2}{10\pi^2 f^3} \Omega_{\text{GW}}(f)$$

地平線内に入った重力波は、宇宙膨張とともに発展。
→ スペクトルの形は、**宇宙進化の情報**を持っている。



背景重力波探査の現状



原図 : Shoda+, PRD (2013)

$$\tilde{h}_{\text{GW}}^2(f) = \frac{3H_0^2}{10\pi^2 f^3} \Omega_{\text{GW}}(f)$$



高周波数では振幅は小さくなる

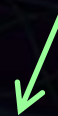
- さまざまな周波数帯で原始重力波観測を観測することで宇宙の進化の情報を得ることが可能.
- インフレーションからの重力波観測には低周波数が有利.
- 0.1Hz以下の周波数帯では, フォアグラウンド重力波が存在.



インフレーションからの重力波観測には,
0.1 -1 Hzの周波数帯が良い.

$$\Omega_{GW} \sim 10^{-16} - 10^{-15}$$
$$\rightarrow \tilde{h}_{GW} \sim 10^{-24} \text{ Hz}^{-1/2} (@ 0.1\text{Hz})$$

- 重力波 – 強い透過力を持ち, 初期宇宙の情報を伝える.
- スペクトルの形 : 初期揺らぎ + 宇宙進化の歴史.



CMB Bモード偏光から
もある程度推定可能.

観測周波数と宇宙の時代が対応.
高周波数 → より初期宇宙の情報.
- Reheating温度(物質の種の形成)
- 宇宙の熱進化史

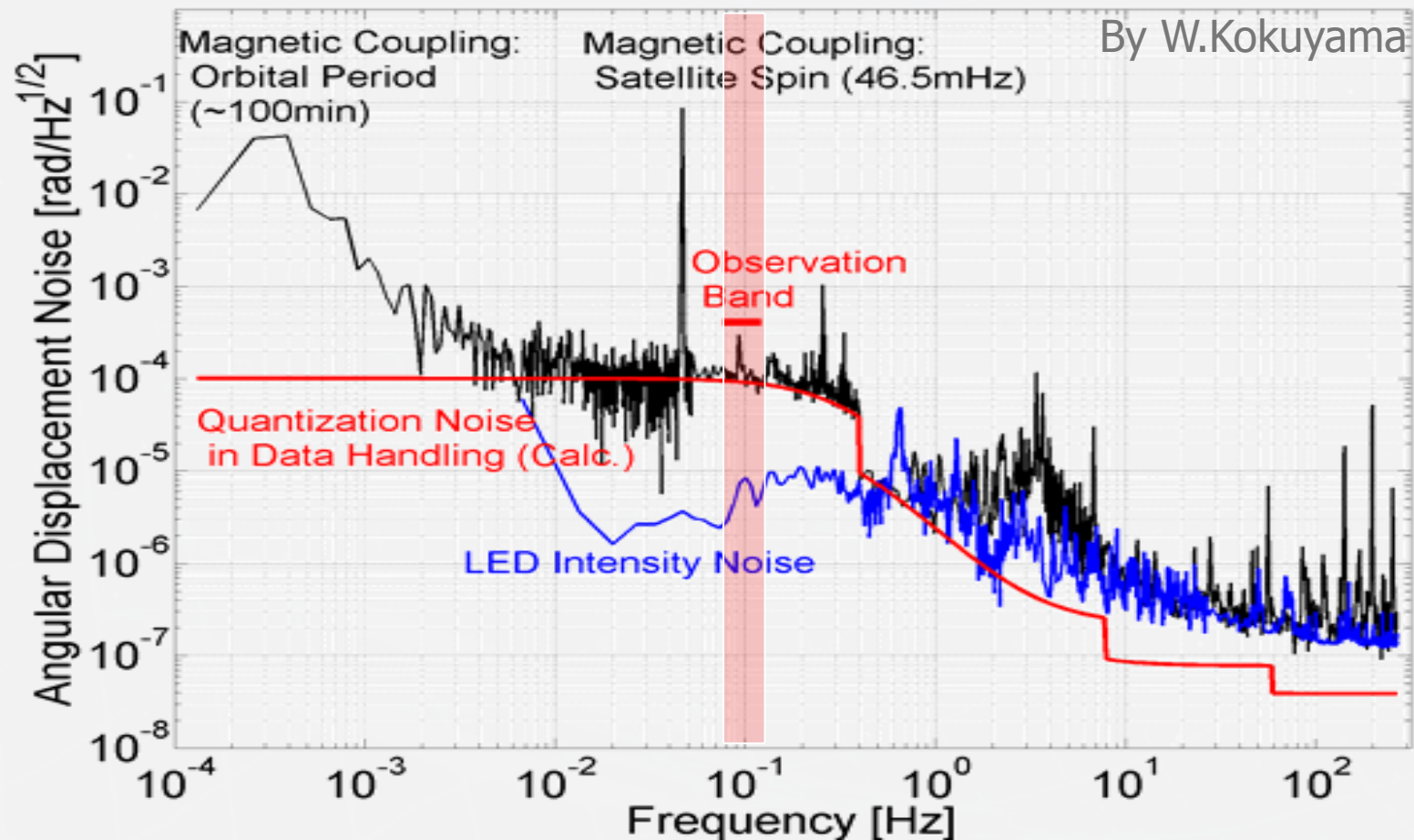
DECIGOが目指す 0.1Hzの周波数帯 :

インフレーション期とBBN期の間の情報

→ CMB-B偏光観測と相補的な観測.

Sensitivity

Though limited by non-fundamental noises,
best as a space-borne GW detector.

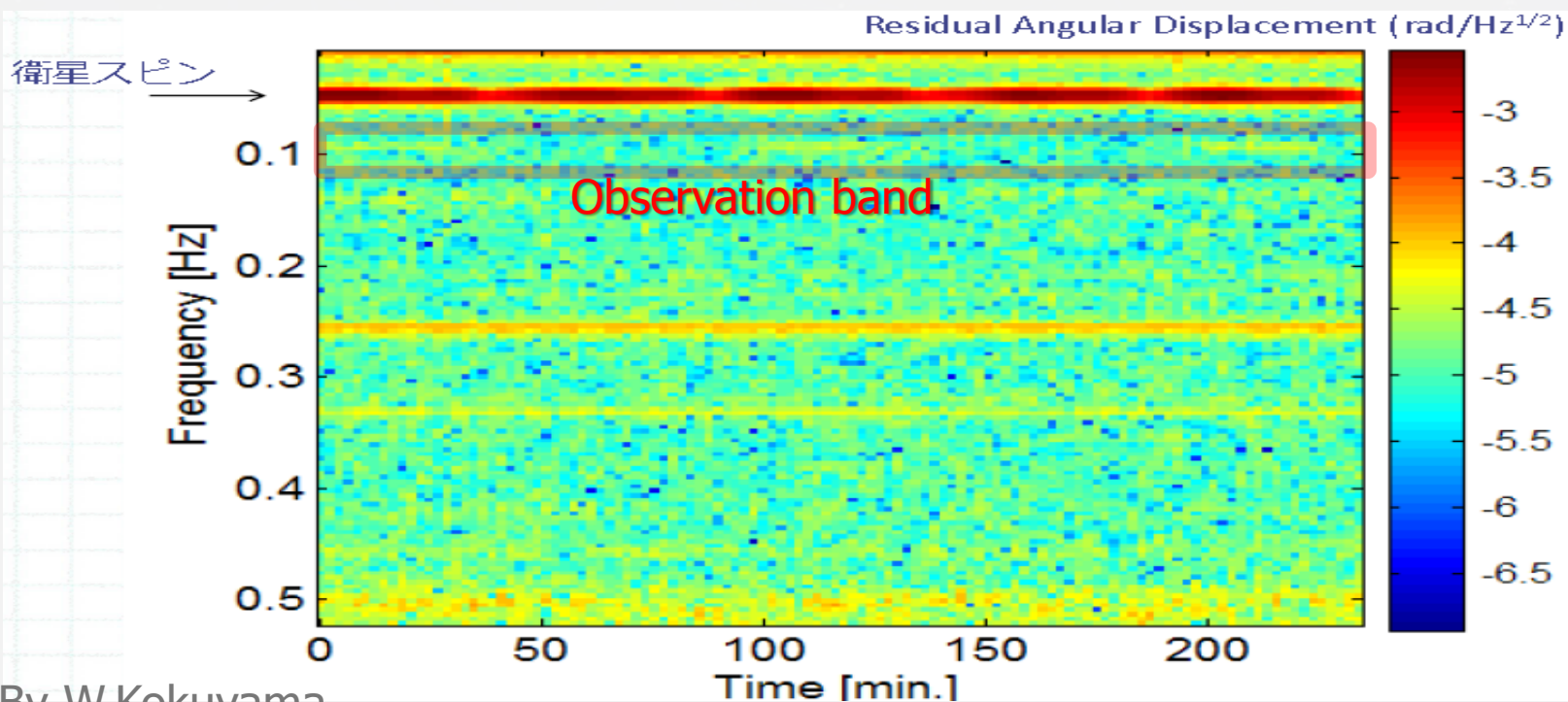
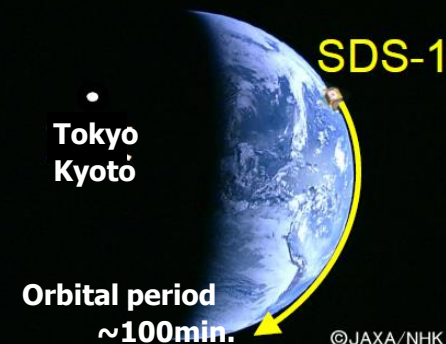


Observation by SWIM

Continuous data taking

Jun 17, 2010 ~120 min.

July 15, 2010 ~240 min.



By W.Kokuyama

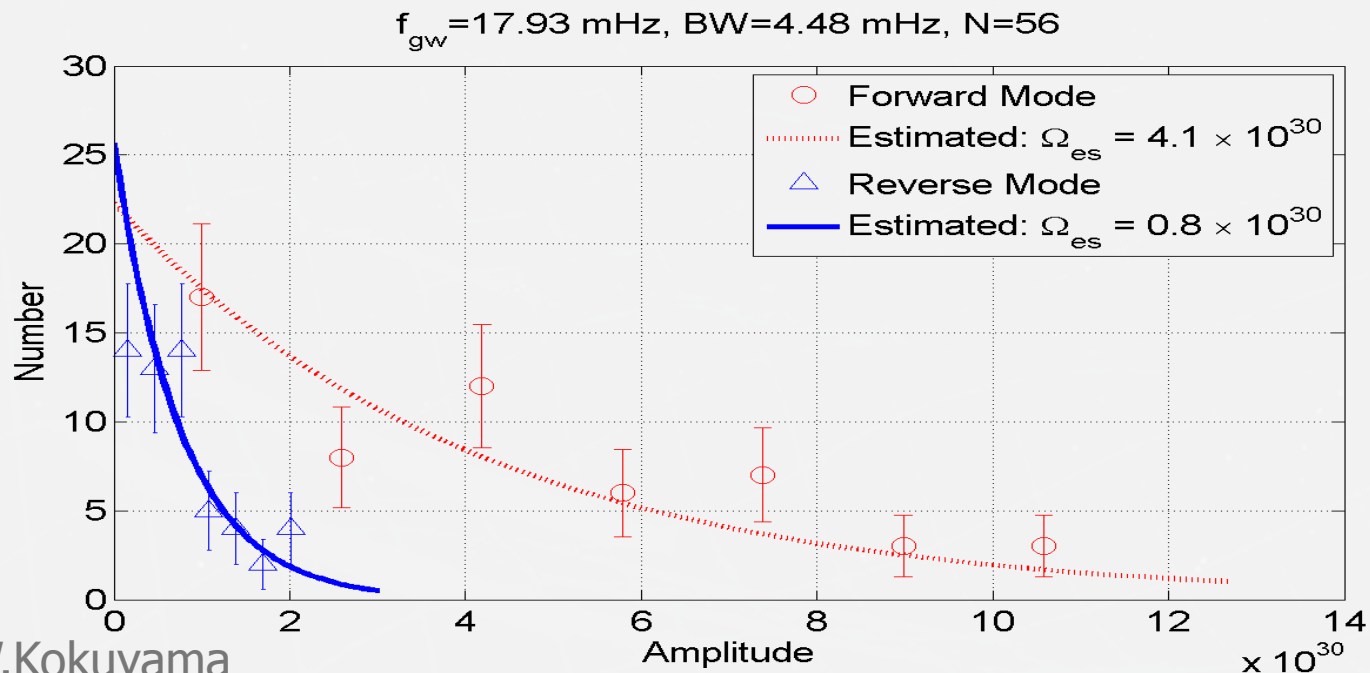
Upper Limit on GWB

Upper Limit at two frequencies (two polarizations)

'Forward' mode $\Omega_{\text{gw}}^{\text{FW}} = 1.7 \times 10^{31}$

'Reverse' mode $\Omega_{\text{gw}}^{\text{RE}} = 3.1 \times 10^{30}$

(C.L. 95%, f_0 18mHz, BW 4mHz)



By W.Kokuyama

背景重力波探査の現状

